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Dynamically updated digital twin for prognostics and health management: Application in permanent magnet synchronous motor

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Dynamically updated digital twin for prognostics and health management: Application in permanent magnet synchronous motor / Guo, Haoyu; Wang, Shaoping; Shi, Jian; Ma, Tengfei; Guglieri, Giorgio; Jia, Rujun; Lizzio, Fausto. - In: CHINESE JOURNAL OF AERONAUTICS. - ISSN 1000-9361. - (In corso di stampa). [10.1016/j.cja.2023.12.031]

Availability: This version is available at: 11583/2985276 since: 2024-01-19T17:27:16Z

Publisher: Elsevier

Published DOI:10.1016/j.cja.2023.12.031

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Chinese Society of Aeronautics and Astronautics & Beihang University

## **Chinese Journal of Aeronautics**

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## 2 FULL LENGTH ARTICLE

# Dynamically updated digital twin for prognostics and health management: Application in permanent magnet synchronous motor

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Received 1 June 2023; revised 16 July 2023; accepted 18 October 2023

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## KEYWORDS

18	Digital Twin (DT);
19	Dynamic Update;

- 20 Independence Principle;21 Multi-field Coupling:
- Multi-field Coupling;
   Permanent Magnet Syn-
- 23 chronous Motor (PMSM);
- 24 Prognostics and Health
  - Management (PHM)



Abstract Current research on Digital Twin (DT) based Prognostics and Health Management (PHM) focuses on establishment of DT through integration of real-time data from various sources to facilitate comprehensive product monitoring and health management. However, there still exist gaps in the seamless integration of DT and PHM, as well as in the development of DT multi-field coupling modeling and its dynamic update mechanism. When the product experiences long-period degradation under load spectrum, it is challenging to describe the dynamic evolution of the health status and degradation progression accurately. In addition, DT update algorithms are difficult to be integrated simultaneously by current methods. This paper proposes an innovative dual loop DT based PHM framework, in which the first loop establishes the basic dynamic DT with multi-filed coupling, and the second loop implements the PHM and the abnormal detection to provide the interaction between the dual loops through updating mechanism. The proposed method pays attention to the internal state changes with degradation and interactive mapping with dynamic parameter updating. Furthermore, the Independence Principle for the abnormal detection is proposed to refine the theory of DT. Events at the first loop focus on accurate modeling of multi-field coupling, while the events at the second loop focus on real-time occurrence of anomalies and the product degradation trend. The interaction and collaboration between different loop models are also discussed. Finally, the Permanent Magnet Synchronous Motor (PMSM) is used to verify the proposed method. The results show that the modeling method proposed can accurately track the lifecycle per-

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Peer review under responsibility of Editorial Committee of CJA.



### https://doi.org/10.1016/j.cja.2023.12.031

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formance changes of the entity and carry out remaining life prediction and health management effectively.

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#### 1. Introduction 33

Due to the significant impacts of product degradation and fail-34 35 ure, such as production disruptions, safety hazards, and higher maintenance costs, the Prognostics and Health Management 36 37 (PHM) technology is widely used in aerospace system. 38 Through real time monitoring the product states, PHM can 39 detect product performance degradation, predict remaining life and diagnose the fault. However, the progress and develop-40 ment of product degradation are difficult to reflect from exter-41 nally detectable sensors, which are the result of internal states 42 changes with the coupling effect of multiple fields. Therefore, it 43 is vital to develop effective health monitoring techniques that 44 can characterize the internal changes under load action. Coin-45 cidentally, the Digital Twin (DT) method has emerged, and 46 47 has received increasing attention through creating a highfidelity digital mirror of physical entity.<sup>1</sup> Michael<sup>2</sup> is credited 48 with proposing DT as a solution for complex product opera-49 tions. The aim of DT is to establish a virtual representation 50 that mirrors the physical entity,<sup>3</sup> enabling the tracking of state 51 variables and parameter changes. On the other aspect, the 52 development of multi-filed coupling modeling provides more 53 detailed possibilities for digital representation of the product 54 55 internal state. DT have the potential to monitor the state, manage the lifecycle, and optimize decision-making for the physical 56 57 entity. The integration of the Internet of Things (IoT) and 58 information systems also brings new energy to the parallel interaction between the physical and cyber domains. Conse-59 60 quently. DT can be seen as highly precise integrated sensors. 61 capturing and representing a wealth of data from physical entities in a virtual environment. Hence, DT-based PHM was first 62 adopted in the aerospace industry,<sup>4</sup> which not only changes 63 with changes in physical entity through real time updating 64 65 DT parameters, but also reflects the changes in the internal state of the product under working conditions. 66

Although DT can synchronously change with changes of 67 physical entity in an ideal state, product degradation is related 68 to multiple factors, and the variation law is very complex in 69 real practice. The synchronous evolution DT is difficult to 70 accurately describe the product lifecycle changes through the 71 entire service life. This paper presents a dual loop DT-PHM 72 73 framework to perform accurate internal state monitoring and 74 product degradation characterization through real time data 75 updating from physical entities to their virtual counterparts. In the first loop, the virtual replica system of physical product 76 is established through gathering the data by sensors, updating 77 the DT and mapping the virtual model. When the process of 78 79 product starts degradation, the DT-based PHM model is switched to the second loop through abnormal detection. 80 The real time data is fed into the DT-based PHM according 81 to the degradation state from time to time, and the entire pro-82 cess DT-based PHM model enables a nearly accurate represen-83 tation during the lifecycle operation through dynamical 84 updating of parameters. The DT-based PHM can foresee 85

probable future issues, and decrease product faults and unnecessary maintenance. In the process of abnormal detection, a 87 theory of Independent Principle is proposed, which supple-88 ments the theoretical system of DT. In this way, DT-based 89 PHM evolves from multi-filed coupling, thus optimizing the 90 updating parameters, increasing the precision of health state 91 recognition and remaining useful life estimation. The Perma-92 nent Magnet Synchronous Motor (PMSM) is used to validate 93 the framework and simulate the wide range of scenarios during 94 the entire operation process. The results show that the PHM 95 DT helps to improve the reliability and maintenance efficiency 96 of complex product. Compared with previous research, the 97 proposed DT-based PHM can accurately describe the working 98 behaviors of the physical entity and the accuracy is improved 99 compared with that of the previous methods.

The contributions of this study are highlighted as follows:

- (1) Considering the difference between normal state DT and degradation DT of product, a dual loop DT based PHM framework is provided, in which the first loop provides the Basic DT (BDT) based on the dynamic updating, and the second loop focuses on the DT-based PHM (PHM-DT) for characterizing the degradation process of product through monitoring the changes in health status.
- (2) The BDT and PHM-DT interaction updates through abnormal detection based on the "Independent Principle", which is a significant contribution to the theory of DT. With regular real-time updates, the comprehensiveness and accuracy of the interaction will steadily increase over time.

At the end of this paper, we validate our approach based on a case study of PMSM. Compared with traditional prediction methods, the method proposed in this paper can better accompany the characteristic change of the entity and the accuracy is improved by about 37%. The rest of the paper is organized as follows: Section 2 introduces the research related to DT and PHM; Section 3 describes the dual closed-loop DT-based PHM framework; Section 4 introduces the DT updates mechanism; Section 5 carries out the case study on PMSM; Section 6 gives conclusions.

## 2. Related work

#### 2.1. Modeling of DT 128

Back in 2011, NASA experts presented the three-dimensional 129 DT that included the physical entity, DT and their bridge, in 130 which the information generated in the physical world was 131 gathered to create and implant DT. This approach emphasizes 132 seamless integration of the IoT and information systems. By 133 harnessing advanced information interaction technologies, 134

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industrial sectors have been pushed towards greater levels of 135 informatization, digitization, and intelligence. This conver-136 gence holds immense potential for driving efficiency, innova-137 tion, and optimization across various industries.<sup>5</sup> Sensors 138 and other devices that collect accurate data on the state of pro-139 cess provide the real time updating strategy to replicate what 140 actually occurs. Building mathematical DT with a sufficient 141 level of complexity are important to forecast real-life behaviors 142 in various contexts. 143

Many manufacturers have already applied the DT concept 144 to actual production. In order to improve the design perfor-145 mance of the product, Marr developed DT for vehicles, which 146 147 helps to realize the optimal design through information interaction between the vehicle and the factory.<sup>6</sup> Boeing CEO said 148 that as we enter the next decade, DT, due to its potential for 149 increased efficiency and improved performance, holds promise 150 for revolutionizing global aircraft manufacturing. In the realm 151 of manufacturing, companies, such as Siemens<sup>7</sup> and the F-35 152 manufacturer Lockheed Martin,<sup>8</sup> are increasingly building vir-153 tual workshops and digital factories to map physical spaces 154 into cyberspace. This innovation holds potential for streamlin-155 ing operations, optimizing resource allocation, and enhancing 156 collaboration. 157

By now, the industrial application of DT technology is 158 mainly concentrated in the fields of production, design, and 159 prediction.<sup>9</sup> DT modeling methods are mainly divided into 160 four types: Finite Element Modeling (FEM), multi-161 disciplinary modeling language, neural network, and mathe-162 matical model. Tao et al.<sup>10</sup> defined DT as the all-factor recon-163 struction and digital mapping of the operational state and 164 operational progress of the physical entity and also informa-165 tion space of the product. Fan et al.<sup>11</sup> proposed a DT visual-166 ization architecture for Flexible Manufacturing Systems 167 (FMS), and explored modeling multi-source heterogeneous 168 information. Dang et al. <sup>12</sup> proposed a DT framework based 169 on cloud computing and deep learning, taking bridge as 170 objects to establish DT. Castellani et al.<sup>13</sup> used the Modelica 171 language to build a DT of a company's power, heating, venti-172 lation and air conditioning systems, and obtained the normal 173 working data of the actual system through the operation of 174 the established DT. Wang et al.<sup>14</sup> established a DT of the 175 image convolutional neural network of the welding process, 176 and obtained the current penetration information by acquiring 177 the weld pool image and current data. Venkatesan et al. 178 developed a neural network DT that conforms to the running 179 state of the motor according to the running time, distance of 180 the vehicle and the health state of the onboard motor. Mogha-181 dam et al.<sup>16</sup> established a torsional dynamic DT for offshore 182 wind turbines, and gave a related parameter estimation algo-183 rithm. Li et al.<sup>17</sup> proposed an adaptive extension-based filter 184 that is robust and accurate in estimating DT parameters for 185 both Li-ion and Lead-acid batteries in the state of charge. 186 Lei et al.<sup>18</sup> took the entire thermal power plant as the object, 187 188 and studied the four-layer architecture about 3D modeling, 189 mathematical modeling, rendering and real time monitoring. Hu et al.<sup>19</sup> constructed a high-precision gas turbine DT, intro-190 duced an error module and kernel density estimation self-191 learning to optimize the update of DT, and carried out fault 192 status diagnosis of the gas turbine based on this method. 193

## 2.2. PHM based on DT

PHM takes center stage in the Industry 4.0 revolution, where 195 the key challenges involve accurately detecting whether equip-196 ment is operating normally and predicting when faults may 197 occur. Effective implementation of PHM holds the potential 198 to minimize the occurrence of catastrophic failures and reduce 190 costs associated with scheduled maintenance. Given the com-200 plexity of understanding internal state changes during degra-201 dation, DT emerges as a promising method to characterize 202 the internal health state across different fields and objects. Lev-203 eraging its technical advantages. DTs enable better monitoring 204 and analysis of the health conditions of products, leading to 205 improved prognostics and decision-making processes. Bai 206 et al.<sup>20</sup> proposed a novel 3D multi-physics DT for proton 207 exchange membrane fuel cell based on the Proper Orthogonal 208 Decomposition (POD) method, and exhibited and analyzed 209 the DT results of voltage, temperature, membrane water con-210 tent and liquid water saturation fields. Ye et al.<sup>21</sup> proposed a 211 reconfigurable Dynamic Bayesian Networks (DBN) method 212 that can capture interactions between damages. Their study 213 shows via a numerical example that the reconfigurable DBN 214 can accurately predict the crack growth acceleration caused 215 by bolt loosening. The method tracks multiple damages and 216 has good physical interpretability. Li et al.<sup>22</sup> proposed a 217 PHM system based on advanced DT technology for the 218 Five-hundred-meter Aperture Spherical radio Telescope 219 (FAST). The PHM system utilizes finite element analysis of 220 the DT to evaluate the safety status and predict the fatigue life 221 of FAST's cable-net structure, enabling effective Condition-222 Based Maintenance (CBM) and ensuring the healthy and safe 223 operation of the structure while improving maintenance effi-224 ciency and reducing costs. Angjeliu et al.<sup>23</sup> established a 225 FEM DT for the Milan Cathedral, which can predict the 226 future damage trend of the building structure through the 227 analysis of the FEM. Liu et al.<sup>24</sup> took High-Speed Permanent 228 Magnet Motor (HSPMM) as an example, and comprehen-229 sively discussed the issues that need to be considered in the 230 construction of a multidisciplinary DT of HSPMM and the 231 fault diagnosis of its electrical drive system. Aivaliotis et al. 232 presented a methodology for calculating machinery equip-233 ment's Remaining Useful Life (RUL) using physics-based sim-234 ulation models and the DT concept, enabling predictive 235 maintenance for manufacturing resources. This method 236 involves modeling resources, gathering data from machine 237 controllers and sensors for tuning digital models, and using 238 simulation results to assess the condition and calculate RUL, 239 and allows for non-invasive monitoring and prediction of 240 machine status. Their methodology is validated through a case 241 study on predicting the RUL of an industrial robot. Oluwase-242 gun et al.<sup>26</sup> proposed a conceptual framework for applying the 243 DT technology to predict the control element drive mechanism 244 and a data-driven method for abnormal detection by using coil 245 current curves to optimize the operation and maintenance pro-246 cess of nuclear power plants. Peng et al. <sup>27</sup> conducted research 247 on the core technology of DT structural rolling bearings, 248 including detection, modeling and PHM technology, and ana-249 lyzed the challenges and future research directions in the devel-250 opment of rolling bearing DT technology. Booyse et al.<sup>28</sup> 251 proposed to use the form of deep DT, which learns from the 252

No. of Pages 18

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253 distribution of health data, and uses its advantage of independence from historical failure data to estimate asset health. 254 Correa-Jullian et al.<sup>29</sup> discussed the core aspects of the design, 255 development, and implementation of data-driven PHM appli-256 cations, and demonstrated that they can improve reliability 257 evaluation in liquid hydrogen storage systems. Based on 258 DT's PHM idea, Candon et al. <sup>30</sup> gave a comprehensive study 259 on DT in the form of machine learning models for aircraft load 260 monitoring, including linear regression models, traditional 261 artificial neural networks, and deep learning strategies. They 262 263 also discussed the need for time-series modeling and explored potential solutions to the issues encountered in traditional or 264 265 modern aircraft data acquisition systems. Their findings hold 266 significant value for researching fatigue problems in mechanical systems. 267

Of all the existing works mentioned above, the DT-PHM 268 methods have the following core limitations: firstly, the estab-269 270 lished DTs are discipline-specific, limiting their ability to com-271 prehensively describe the equipment's state; secondly, the physical interpretation of parameters in DT, built using multi-272 disciplinary language, neural network models, or finite element 273 methods, is unclear; thirdly, the established DTs are open-loop 274 models, lacking a corrective feedback loop for comparison; 275 lastly, most of the DTs are based on fixed-parameter models, 276 277 unable to adapt to changes in actual equipment states. Therefore, the focus of this paper is to establish a multidisciplinary 278 279 coupling DT and its dynamic update mechanism. Moreover, DT is combined with PHM to build a general DT operation 280 and maintenance framework. 281

## 282 3. Dual loop DT based PHM framework

With the long-term operation of actual equipment, the digital 283 model established according to traditional simulation has a 284 certain risk of errors. Due to the factors such as degradation 285 or failure, the parameters or the model will change accord-286 ingly, resulting in deviation of DT performance from the 287 entity. Therefore, a closed-loop comparison link is needed to 288 ensure the consistency of the DT with the actual equipment. 289 The two main aspects of DT research are "establishment" 290 and its "application". Based on this point, this paper presents 291 292 a dual closed-loop DT-based PHM framework, as shown in 293 Fig. 1.

294 (1) First Loop: Basic DT

The first loop is used to establish the virtual model of the physical product under the normal condition that consists of four parts: physical entity, IoT, DT, and DT correction. The physical entity serves as the foundation for DT modeling, and acts as the starting point for the DT system. IoT establishes a data interconnection between the physical space and virtual space. This information connection provides the necessary "data nutrients" required for the dynamic updating of the DT system. DT modeling serves as the essence of DT, encapsulating its very soul. Dynamicity and high accuracy form the core essence of DT. We define DT as a dynamic model, wherein the internal parameters are influenced by its self-state and operating conditions, ensuring a continuous adaptation to changing circumstances.

## (2) Second Loop: PHM-DT

The second loop is used to establish the virtual model of product during lifecycle degradation. Here, the PHM is the core of DT. In the second loop, the health status can be detected, faults can be diagnosed, and the remaining useful life can be predicted with PHM-DT. In order to describe the PHM scenarios at different levels, PHM-DT collects the observed information on the physical product, iteratively updates the parameters to the corresponding DT, and integrates comprehensive lifecycle DT. Since the degradation trend of physical entity is closely related to the operational condition and uncertainty factors, the evolution of product status is different under the degradation scenario. In order to capture the degradation state accurately, processing of the integrated data obtained from the physical entity sensors is carried out, and updating of the dynamic parameters is implemented from time to time.

## (3) Interaction: Abnormal detection

As the physical entity operates, the DT remains in a 329 dynamic process of adaptation, and needs the model evolution 330 technology to drive self-updating of the virtual model. Abnor-331 mal detection is the important interaction between the dual 332 loop DT. In this paper, we put forward a novel concept of 333 "Independence Principle" in the abnormal detection process 334 to ensure the significance of the established DT. The specific 335 principle will be discussed in detail in the next section. This 336 paper transforms from basic DT to PHM-DT with abnormal 337 detection. Abnormal detection can establish the seamless con-338

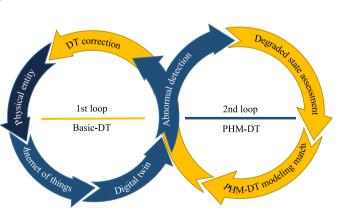


Fig. 1 Dual loop DT based PHM framework.

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nection between basic DT and PHM-DT and automatically 339 respond to the changes in product status of entire lifecycle. 340

341 3.1. Basic DT modeling (1st loop DT)

In general, a DT is an integrated simulation of a physical entity 342 343 that encompasses multiple disciplines, such as mechanical and 344 electrical disciplines, across its operation (see Fig. 2). If the operational physical mechanism is known, we should establish 345 the real-time dynamic operation process model of physical 346 entities with multi-field coupling. The basic DT, as the mirror 347 348 image of physical product, can be established by multi-field 349 coupling modeling.

As shown in Fig. 2, the physical entity and DT share the 350 same input U. The actual output from the physical product 351 is represented as O. Based on the first-principle structure, the 352 output of basic DT can be shown as 353 354

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$$\hat{O} = f(x_1, x_2, \cdots, x_m, x_{m+1}, \cdots, x_n, U) + D$$
 (1)

357 where  $\hat{O}$  is the performance output of DT. U is the input of 358 DT, and  $\boldsymbol{D} = [d_1, d_2, \cdots, d_r]$  is the disturbance and uncertainty. The of DT parameters is expressed as 359 360  $X = [x_1(t), x_2(t), \cdots, x_m(t), x_{m+1}, x_{m+2}, \cdots, x_n],$ in which 361  $[x_1(t), x_2(t), \dots, x_m(t)]$  varies with the operational condition, and  $[x_{m+1}(t), x_{m+2}(t), \dots, x_n(t)]$  keeps constant like the geomet-362 ric size. The impact on DT parameters of operational condi-363 tion can be expressed as 364 365

$$x_i(t) = x_0 + q(c_1, c_2, \cdots, c_l)$$
(2)

368 where  $x_0$  is the initial value of the parameter, q represents the operating conditions on the parameters, and  $C = [c_1, c_2, \cdots, c_l]$ 369 370 is the set of condition parameters.

371 In certain scenarios, model-based approaches may not 372 effectively capture the complexities or uncertainties present in real systems. In such cases, data-driven approaches can be 373 employed to complement the model and enhance its perfor-374 375 mance. Data-driven methods leverage the analysis and mining of extensive system data to extract valuable patterns, relation-376 ships, and laws. By incorporating actual data, these 377 approaches can provide additional insights to improve the 378 379 accuracy and effectiveness of the model.

By leveraging their respective advantages, they complement 380 381 one another and aid in joint characterization of the target 382 object. This integration can be mathematically expressed as 383 Eq. (3)

$$\hat{\boldsymbol{O}} = f(\vartheta(\cdot), \eta(\cdot), \boldsymbol{U}) + \boldsymbol{D}$$
(3)

where  $\vartheta(\cdot)$  represents the model-driven model,  $\eta(\cdot)$  represents the data-driven model and the parameters X included in the overall model.

Since the physical mechatronic product operates in multiple disciplines such as mechanical, electrical, control et al., its dynamics, electricity and another part directly affect product performance. When the basic DT is applied to the physical product, multi-field coupling DT is established based on the dynamics equation and Newton's law. In order to promote the application of virtual spaces in the DT establishment, information interaction is responsible for updating the DT parameters collected from entity sensors. It is important to note that when creating a DT for different objects, it is essential to incorporate the relevant disciplines specific to the entities. After establishing the basic DT based on multi-field coupling, its parameter updating mechanism can be shown in Fig. 3.

## 3.2. PHM based on DT (2nd loop DT)

Although basic DT can accurately map the physical product with the changes under the operating condition, it is difficult to express the physical product corresponding to the full life cycle process. PHM DTs are different in differed health states in 2nd loop DT.

This paper defines the health state index of product as

This paper defines the health state index of product as  

$$H = 1 - \frac{\sigma_0 - \sigma_t}{\sigma^{\tau}}$$
(4)
(4)

where  $\sigma_0$  represents the initial value of the key parameter,  $\sigma_t$ represents the value of the key parameter at time t, and  $\sigma^{\tau}$  represents the degradation threshold value of the key parameter. The health state index can reflect the degradation degree of product within the range of 0 to 1. Let  $H = \{H_1, H_2, \dots, H_b\}$ represents the product state from complete wellness, minor fault, moderate fault to total failure. By assigning appropriate degradation level to the actual situation, we can effectively determine the operational status of an entity and build the appropriate DT. Fig. 4 shows the PHM based on DT in the 2nd loop DT.

In Fig. 4, PHM-DT modeling results not only form the final 425 insights through model matching, but also realize the co-426 evolution through model switching updating. For example, if 427

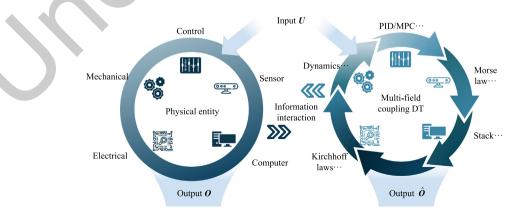


Fig. 2 Basic DT modeling with multiple-field coupling.

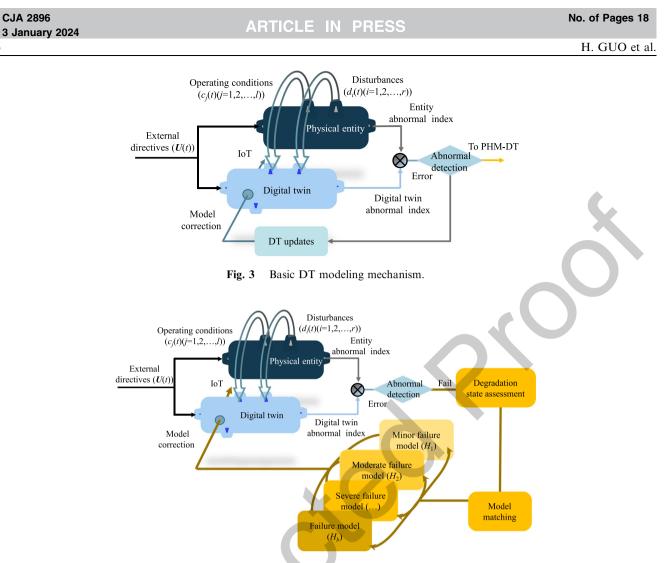


Fig. 4 PHM-DT modeling mechanism.

we divide the health state into five parts, the product health
state index can be written as

$$H = \begin{cases} 1, \text{Normal} \\ H_1, \text{Minor fault} \\ H_2, \text{Moderate fault} \\ H_3, \text{Severe fault} \\ 0, \text{Failure} \end{cases}$$
(5)

- The meaning of the health state of product is as follows:
- 434 (1) H = 1 represents the normal state, in which the product 435 has no or a little change without affecting product 436 performance.
  - (2)  $H = H_1$  indicates the minor fault state, in which minor degradation of the product occurs during the operation.
- 439 (3)  $H = H_2$  expresses the moderate fault state, in which 440 moderate degradation of the product arises after run-441 ning a considerable period of time.
- 442 (4)  $H = H_3$  is the severe fault state, in which serious degra-443 dation of the product occurs in the late stages of product 444 life.
- 445 (5) H = 0 represents the failure state, which means the pro-446 duct reached its lifespan.

Different health state corresponds to different parameters 448 or models within the DT. Therefore, it is necessary to 449 update the DT based on the current health state of equip-450 ment by detecting its status changes. For example, the val-451 ues of key parameters within the DT may vary through 452 updating the algorithm when the equipment deteriorates 453 from minor fault to moderate fault state. Hence, by employ-454 ing appropriate algorithms, the DT is updated by consider-455 ing the current health state of the equipment to ensure its 456 alignment with the physical entity. In the case study of this 457 paper, we utilize the recursive least squares method to 458 update the flux  $\psi_f$ , resistances R, and inductance L in the 459 PMSM DT based on the current state of health, which 460 ensures that the DT remains consistent with the entity 461 across different health stages. 462

In summary, adjusting the parameters or structure of the DT based on the equipment's health state is a crucial step in ensuring its consistency with the actual equipment. By updating the DT using suitable algorithms when the equipment's state changes, the reliability and adaptability of the DT can be enhanced. Such update strategies help monitor and maintain the entity more effectively and provide accurate predictions and decision support.

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DT appropriately. Because the indices are independent of pro-478 479 duct output and internal parameters, we define the idea as "Independence Principle". According to this principle, anoma-480 lies in DT should be identified based on "indices", and the DT 481 482 is subsequently updated based on the characteristic informa-483 tion of the real equipment. In other words, it is not meaningful to directly compare the output or the main parameters 484 485 between DT and the physical entity, because we can already obtain information about the target by detecting the signals 486 from the entity. The purpose of building DT is not simply to 487 create it, but rather to improve its accuracy by incorporating 488 relevant and independent indices. 489 As shown in Fig. 3, the differences between DT and entity 490

4. Dynamic update mechanism of dual loop DT based PHM

It is obvious that abnormal detection is the most important

interaction of dual loop DT based PHM framework. Accord-

ing to the health state of physical entity, abnormal detection

can be triggered between BDT and PHM-DT. This paper pre-

sents "indices" of abnormal detection as a trigger to update the

are initially identified through abnormal indices. If the differ-491 ence detection is passed, the DT is considered reliable and 492 doesn't require updates, and the output result  $\hat{O}$  of the DT 493 can be directly utilized. Once the DT's consistency is con-494 firmed, the corresponding output result can be provided to 495 external sources. If the performance of a physical entity 496 degrades and abnormal detection is triggered, the parameters 497 of the DT are needed to be updated using parameter estima-498 tion techniques. Furthermore, if the DT still fails to pass 499 abnormal detection after several parameter updates, it suggests 500 the occurrence of a change in the mechanism or structure of 501 502 the entity.

Consequently, the DT needs to be remodeled through system identification and other appropriate methods. It is essential for all updated DTs to successfully pass the consistency measure before they can be deemed consistent with the technical state of the physical entity.

## 508 4.1. Definition of the trigger based on abnormal indices

Components in actual equipment will degrade or fail as the
operating time increases, which will cause the entity to deviate
from the state of the DT. To ensure consistency between DT
and the entity, we utilize the indices as the triggering criteria.
The abnormal trigger is defined as

516 trig = 
$$g(G_{\text{entity}}^0, G_{\text{entity}}^t, G_{\text{DT}})$$
 (6)

where  $G_{\text{entity}}^0$  is the abnormal index at the previous moment of 517 physical entity,  $G'_{entity}$  is the abnormal index at time t of entity, 518 and  $G_{\rm DT}$  is the abnormal index at the current moment of DT. 519 When the threshold is not satisfied, the abnormal trigger is ini-520 tiated, which indicates that the DT needs to be updated. The 521 abnormal trigger is a function that depends on three abnormal 522 indices  $(G_{\text{entity}}^0, G_{\text{entity}}^t, G_{\text{DT}})$  selected from the DT system. Its 523 functional relationship can be described by the function  $g(\cdot)$ . 524 The specific form of  $g(\cdot)$  is not fixed, and determining the 525 appropriate pattern for obtaining abnormal triggers based on 526 the abnormal indices requires consideration of different 527 objects and actual situations. 528

In the following are the guidelines of Independent Principle:

- (1) The abnormal factor  $G_*$  is a parameter that can effectively characterize the abnormal features between DT and physical entity.
- (2) Try to avoid selecting output or key internal parameters as abnormal indices. Since the relevant parameters can already be obtained by traditional means, the existence of parameters derived from DT becomes meaningless.

However, the specific process of selecting these factors should be based on the unique characteristics of the actual equipment. It is essential to consider the factors such as design, functionality, and operational environment of the equipment. By thoroughly understanding these characteristics, we can determine the specific abnormal factors that are most relevant and significant for detecting deviations in the equipment's performance.

4.2. DT updates

 $\boldsymbol{K}_{t+1} = \boldsymbol{P}_t$ 

For different form DT, parameter updates need to be combined in different ways to achieve the best results. According to the description of the model correction link in the previous chapter, the update of DT can be divided into parameter update and model update. For linear models, the Recursive Least Square (RLS), Gradient Descent (GD), and Conjugate Gradient (CG) can be used to update parameters. For nolinear models, model parameters can be updated using Particle Swarm Optimization (PSO), Maximum Likelihood estimation (ML), Artificial Neural Networks (ANN), and other methods.

Taking the RLS algorithm as an example, consider the linear model. We can assume that there exists a relationship between the variable parameters and their observed values, which can be represented as

$$\boldsymbol{Z}_{t} = \boldsymbol{Q}_{t}\boldsymbol{\theta} + \boldsymbol{V}_{t}t = 1, 2, \cdots, l \tag{7}$$

where  $Z_t$  represents an observation output vector consisting of Z(t), *t* represents the observation at the *t*<sup>th</sup> step,  $Q_t$  represents an observation vector consisting of *t* state values,  $\theta = [x_1, x_2, ..., x_m]^T$  represents the parameter that requires updating, *l* is the number of observation, and  $V_t$  refers to the uncertainty term or perturbation.

The estimation algorithm of RLS parameters is

$$\boldsymbol{P}_{t+1}\boldsymbol{q}^{\mathrm{T}}(t+1) \times [\lambda + \boldsymbol{q}(t+1)\boldsymbol{P}_{t}\boldsymbol{q}^{\mathrm{T}}(t+1)]^{-1}$$
$$\boldsymbol{P}_{t+1} = \frac{1}{2}[\boldsymbol{P}_{t} - \boldsymbol{K}_{t+1}\boldsymbol{q}(t+1)\boldsymbol{P}_{t}] \tag{8}$$

$$\hat{\boldsymbol{\theta}}_{t+1} = \hat{\boldsymbol{\theta}}_t + \boldsymbol{K}_{t+1} \Big[ \boldsymbol{Z}(t+1) - \boldsymbol{q}(t+1) \hat{\boldsymbol{\theta}}_t \Big]$$
(6)

where  $K_{t+1}$  is the correction gain; q(t) is the  $t^{\text{th}}$  observation, and is an element of  $Q_t$ ;  $\hat{\theta}_t$  is the estimated value of the parameter at the  $t^{\text{th}}$  step of  $\theta$ ; Z(t) is the measured value at the  $t^{\text{th}}$ step, and is an element of  $Z_t$ ;  $P_{t+1}$  is the prediction of the  $(t+1)^{\text{th}}$  step based on the measurement at the previous step, and  $P_0$  needs to be preset with a suitable initial value;  $\lambda$  is forgetting factor. The parameters can be updated through iterative updates using this method.

In the linear model, parameter  $\hat{\theta}_i$  can be updated iteratively using Eq. (8) to gradually approach the true value of the parameter. This iterative updating process allows for refining

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the parameter result and reducing the discrepancy between the estimated and actual values.

As non-linear models, taking the PSO algorithm as an example, we define the objective function as

$$\chi(\boldsymbol{\theta}) = \pi(\boldsymbol{\theta}) \cdot \operatorname{sign}(\pi(\boldsymbol{\theta})) \tag{9}$$

where  $\pi(\cdot)$  represents the deviation function between the entity 592 observation and the DT result observation, and  $sign(\cdot)$  is a 593 594 symbolic function. In the context mentioned, the deviation function can be defined as  $\pi(\theta) = \mathbf{Z} - \hat{\mathbf{O}}$ . The task of PSO is 595 to find the minimum value  $\gamma(\theta)$ , and determine the zero solu-596 tion for  $\theta$ . The resulting solution represents the optimal 597 parameter that satisfies the entity observation conditions. 598 599 Therefore, the objective function is denoted as  $\chi(\boldsymbol{\theta}) = (\boldsymbol{Z} - \hat{\boldsymbol{O}}) \cdot \operatorname{sign}(\boldsymbol{Z} - \hat{\boldsymbol{O}})$ . The PSO algorithm can be used 600 to obtain the minimum value of  $\chi(\theta)$  and the resulting 601 solution. 602

The parameters that need to be updated in the non-linear models are  $\boldsymbol{\theta} = [x_1, x_2, \dots, x_m]^{\mathrm{T}}$ . The iterative update of the optimal solution is shown in Eq. (10).

$$\begin{cases} v_i^{k+1} = \omega v_i^k + c_1 r_{i1}^k (\kappa_i^k - x_i^k) + c_2 r_{i2}^k (\kappa_g^k - x_i^k) \\ x_i^{k+1} = x_i^k + v_i^{k+1} \end{cases}$$
(10)

where  $v_i^k$  represents the speed of the *i*<sup>th</sup> particle at the *k*<sup>th</sup> iter-609 ation, with the initialization speed being 0;  $\omega$  denotes the iner-610 tia weight, which controls the impact of the particle's previous 611 velocity on the current velocity;  $c_1$  represents the individual 612 learning factors, typically assigned a value of 2; and it determi-613 614 nes how much a particle relies on its own best solution;  $c_2$  is the 615 social learning factor, also typically set to 2. and it decides how much a particle considers the global best solution found by all 616 particles;  $r_1$  and  $r_2$  are random numbers with the values rang-617 ing from 0 to 1;  $\kappa_i^k$  refers to the optimal target solution 618 obtained by the *i*<sup>th</sup> particle as of the  $k^{th}$  iteration;  $\kappa_a^k$  represents 619 the optimal target solution found by all particles up to the  $k^{th}$ 620 iteration;  $x_i^k$  denotes the solution corresponding to all particles 621 at the  $k^{\text{th}}$  iteration. 622

By assigning appropriate values to these variables and 623 applying them within the PSO algorithm, it becomes possible 624 to optimize the search for the optimal solutions in a multi-625 dimensional space. It is important to note that PSO may pro-626 duce multiple sets of solutions that meet the objective function. 627 628 Therefore, it is necessary to filter the obtained solutions based 629 on the specific situation and select the parameter set that best 630 fits the actual situation as the updated value. Hence, selecting an appropriate updating method for DT requires considering 631 the complexity of the model, the nature of the data, and the 632 desired optimization objectives. By taking these factors into 633 account, we can achieve the best results in updating DT 634 parameters and updating accuracy and performance. 635

## 636 4.3. Consistent measurement

It is crucial to verify and compare the output characteristics of
the DT with those of the actual equipment to ensure its effectiveness and accuracy. This validation process is necessary to
confirm that the updated DT appropriately reflects the state
of the entity. By assessing the similarity between the updated

DT and the entity, we can ensure the reliability of the degradation assessment.

The Mahalanobis Distance (MD) is a commonly employed algorithm in machine learning for measuring the dissimilarity between two samples. It calculates the difference by considering the covariance between variables. However, a disadvantage of MD is its tendency to amplify the influence of variables with small changes. This feature can be considered as an advantage in monitoring the disparity between the updated DT and the entity.

We define state vectors and a set of vectors as follows:

where  $\theta_w$  represents the state matrix composed of  $\omega$  measurements of the entity;  $\theta_{DT}$  is the running state vector value of DT; *X* consists of DT and *v* entity state vectors under the same working condition. In calculating the MD, the number of the sample size is required to be greater than the number of dimensions of the sample, that is v + 1 > m. The mean of these *v* entity state vectors is defined as

$$\boldsymbol{\mu}_{\alpha} = \frac{1}{v} (\boldsymbol{\theta}_1 + \boldsymbol{\theta}_2 + \ldots + \boldsymbol{\theta}_v) \tag{12}$$

Let matrix D be the inverse of the transposed covariance of X, as shown as follows:

$$\boldsymbol{D} = \operatorname{Cov}(\boldsymbol{X}^{\mathrm{T}})^{-1} \tag{13}$$

Therefore, the MD between  $\mu_{\alpha}$  and  $\theta_{DT}$  is

$$MD = \sqrt{\left(\boldsymbol{\mu}_{\alpha} - \boldsymbol{\theta}_{DT}\right)^{T} \boldsymbol{D}(\boldsymbol{\mu}_{\alpha} - \boldsymbol{\theta}_{DT})}$$
(14) 674

We need to set a threshold  $\varepsilon$  based on actual experience. When MD  $< \varepsilon$ , it is considered that the updated DT has a high state consistency with the entity; if MD  $> \varepsilon$ , it is considered that the DT still needs to be updated. It is important to note that validation of the model is required after update of the parameters and structure.

## 5. Case study: PMSM

The PMSM is a highly integrated and multi-filed electrome-682 chanical device that serves a crucial role in energy transforma-683 tion through electric energy. The accurate modeling and 684 coupling of its multiple disciplines are essential for establishing 685 the DT based model. This paper focuses on developing a com-686 prehensive DT for a surface-mounted PMSM that takes into 687 account various disciplines, including electricity, control, 688 dynamics, power loss, and thermal aspects. The DT is estab-689 lished by combining both model-driven and data-driven fusion 690 techniques. 691

## 5.1. Multi-filed modeling of PMSM 692

## 5.1.1. Electrical & dynamics models of PMSM

The classical PMSM's d-q voltage mathematical model694(model-driven) with uncertainty are695696696

$$\int u_d = Ri_d + L_d \frac{di_d}{dt} - \omega_e L_q i_q + \delta_d \tag{15}$$

$$\begin{cases}
 u_q = Ri_q + L_q \frac{di_q}{dt} + \omega_e (L_d i_d + \psi_f) + \delta_q
\end{cases}$$
(15)
  
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where  $u_d$ ,  $u_q$  are the *d*-*q* components of the stator voltage;  $i_d$ 699 and  $i_q$  are the *d*-*q* axis components of the stator current; *R* is the resistance of the stator;  $L_d$  and  $L_q$  are the *d*-*q* components of the inductance components;  $\omega_e$  is the electrical angular velocity;  $\psi_f$  is the permanent magnet flux;  $\delta_d$  and  $\delta_q$  are the uncertainty of the voltage equation, which are used to ensure 704 the accuracy of PMSM DT and are the basis for parameter 705 706 identification.

The equation for the electromagnetic torque is shown as follows:

$$T_{e} = \frac{3}{2} P_{n} i_{q} \psi_{f}$$
(16)

712 where  $P_n$  is the number of pole pairs of the PMSM. The dynamics equation of the PMSM is 713

$$T_e = T_m + B\omega_r + J\frac{\mathrm{d}\omega_r}{\mathrm{d}t}$$
(17)

where  $T_m$  is the load torque; *B* is the friction coefficient;  $\omega_r$  is 717 the mechanical angular velocity; J is the total moment of iner-718 tia of the rotor and the external load. 719

#### 5.1.2. Control model of PMSM 720

The PMSM entity used in this paper is controlled by the dual 721 closed-loop Field-Oriented Control (FOC) method with  $i_d = 0$ 722 A. The external rotation speed and torque measured from 723 724 actual PMSM are used as the input to the control system to drive the PMSM DT. The PMSM DT electrical status can 725 be controlled by means of a physical rotation speed signal 726 and torque condition (see Fig. 5). 727

Now there are a variety of control methods for PMSM, 728 such as Model Predictive Control (MPC), ML-assisted meth-729 ods, etc. In this paper, only PI control in the actual PMSM 730 object is used as a part in the DT. The study of PMSM control 731 732 methods is not included in this paper.

#### 5.1.3. Power loss model of PMSM 733

734 Copper, iron and magnetic losses are the main causes of 735 736 PMSM heating. The calculation model of copper loss is

$$P_{\rm Cu} = \frac{3}{2} i_q^2 R \tag{18}$$

where  $i_q$  is the result of the constant amplitude transformation, 739 so its RMS value needs to be divided by  $\sqrt{2}$ . 740

Commonly used the calculation methods for analyzing iron 741 and magnetic losses in PMSM include empirical formulas and 742 Finite Element Method (FEM). However, empirical formulas 743

often involve complex non-linear links and numerous parameters that are inconvenient to measure directly, such as eddy current density and conductivity.<sup>31</sup> Moreover, discrepancies in parameter values can lead to deviations between calculated and actual results. To overcome these challenges and achieve higher accuracy, the model-driven method is selected for analyzing the loss in PMSM.

This paper uses the experiment and FEM analysis tool Maxwell to conduct 121 sets (5-15 N·m; 1000-2800 r/min) simulation near the rated working conditions (10 N·m, 2500 r/min) of the PMSM (see Fig. 6).

The three graphs presented above depict the iron loss, magnetic loss, and copper loss of the PMSM with various rotation speeds and torque conditions. Observing the graphs, it is evident that the iron and magnetic losses of the PMSM demonstrate a robust correlation with the rotation speed, while the copper loss exhibits a strong correlation with the torque applied. Moving on to the three graphs below, they showcase the input power, output power, and the error between them (input power - output power).

Based on the relationship observed between iron and magnetic losses, which is related to the rotation speed and torque, a quadratic linear formula is fitted to establish a data-driven model for the iron and magnetic losses of the PMSM. The resulting data-driven model can be represented as follows:

$$P_{\text{Iro}} = L_1(i_q, \omega_r)$$

$$P_{\text{Mag}} = L_2(i_q, \omega_r)$$
(19)

where  $P_{\rm Iro}$  and  $P_{\rm Mag}$  represent the iron loss and magnetic loss, respectively;  $\omega_r$  represents the rotation speed;  $L_1$  and  $L_2$  represent the function of iron loss and magnetic loss.

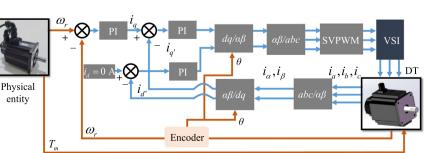
Additionally, the normal distribution of the power error is verified using a Q-Q plot. The analysis reveals that the error between the input and output power follows a normal distribution with a mean of 0. This confirms the acceptability of the power loss model derived from the FEM calculations of iron, magnetic and copper losses (see Fig. 7).

## 5.1.4. Thermal model of PMSM

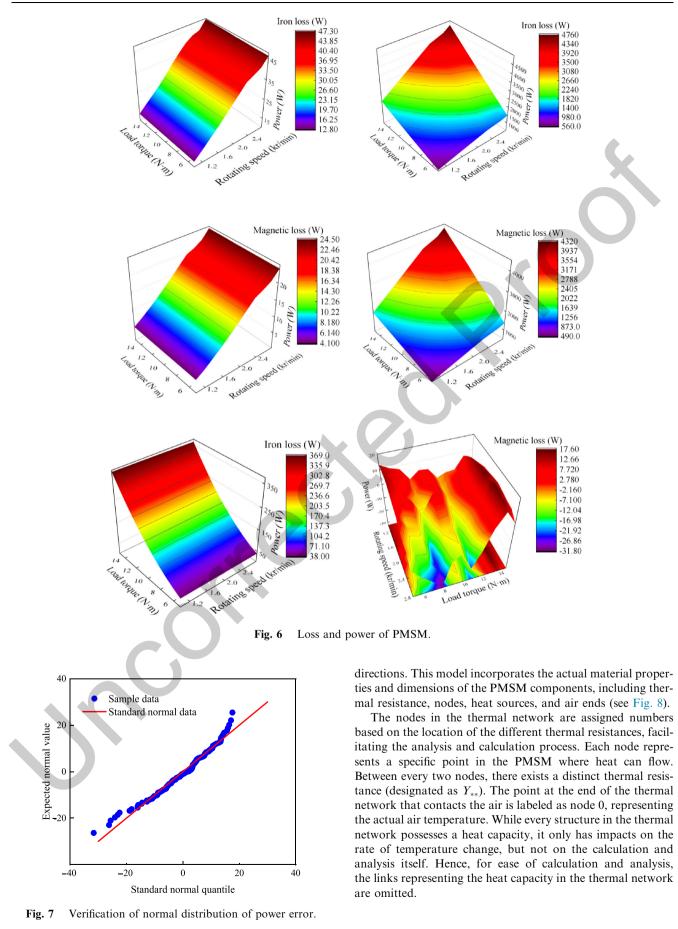
Similar to electrical resistance, heat transfer in PMSM is 782 impeded by various structures, which can be referred to as 783 thermal resistance. The value of thermal resistance is typically 784 determined by the material and dimensions of the correspond-785 ing structure. To accurately model the thermal behavior of the 786 PMSM, an online thermal model has been developed based on 787 the Mellor thermal network, considering the radial and axial 788

Electrical, control and dynamics model of PMSM

Fig. 5 Control method of DT interaction with entity PMSM.



H. GUO et al.



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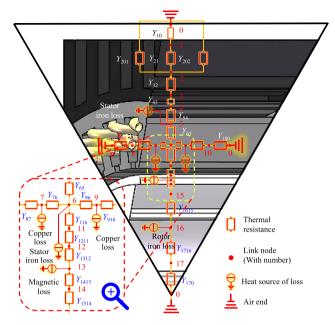


Fig. 8 Schematic diagram of PMSM thermal network structure.

The primary sources of heat within the PMSM are the iron, copper, and magnetic losses. When considering steady heat generation, these losses can be likened to current sources in an electrical circuit. Based on the specific locations of these heat sources, we assign nodes 5, 6, 7, 9, 13, and 16 to represent their respective heat source locations within the thermal network.

To enable real-time temperature calculations for each node 812 in the PMSM, we draw a comparison between the PMSM's 813 thermal network and a power grid, and employ the admittance 814 matrix algorithm, commonly used in power system analysis, to 815 achieve real-time monitoring of the temperature for each node 816 within the thermal network. This approach allows for accurate 817 818 and timely temperature estimation in the PMSM during operation (see Fig. 9). 819

According to the PMSM structure, we establish a framework consisting of 17 nodes to represent the temperature at various locations within the PMSM. In Fig. 9, nodes 5, 6, 7, 9, 13, and 16 within the heat source vector on the left side denote the input points for heating power related to iron, magnetic, and copper losses. The remaining nodes are set to zero, indicating no heat source at those locations. The unit of the heat source vector is measured in Watts.

The matrix shown in the middle of Fig. 9 represents the admittance matrix associated with the PMSM. Each element in the matrix corresponds to the admittance of the respective thermal resistance in the thermal network, which is essentially the reciprocal of the thermal resistance. This matrix is a large  $17 \times 17$  sparse diagonal matrix, where all elements except for the identified ones are set to 0. The element highlighted in yellow within the matrix represents the admittance at the corresponding heat source. On the right side, the node temperature vector represents the temperatures of the 17 nodes relative to the air end point, which is designated as 0. The unit of temperature for all nodes is Kelvin. The unit of the elements in the admittance matrix is Watts per Kelvin.

Additionally, the Temperature Coefficient of Resistance (TCR) is defined as

$$\Gamma CR = \frac{R_2 - R_1}{R_1(T_2 - T_1)}$$
(20)

where  $R_1$  and  $R_2$  is the thermal resistance at  $T_1$  and  $T_2$ , respectively; TCR is around 0.004 at 25 °C (common metal). The material and size inside the PMSM will not change. When the PMSM works stably, there will not be such a large temperature change during stable operation. In other words, the influence of thermal resistance on  $Y_{**}$  can be ignored.

According to the first-principle structure of dynamic DT, the input is  $U = [\omega_r, T_m]$ ; *D* is disturbance  $[\delta_d, \delta_q]$ ;  $c_j(t)$  is the working condition (temperature, etc.); internal time-varying parameters of the model are  $x_m = \{\psi_f, R, L_d, L_q\}$ ; the static parameters are  $x_n = \{P_n, B, J, Y_{**}\}$  and the inner parameters in Eq. (19); DT output are  $\hat{O} = [i_d, i_q, u_d, u_q]$ . Therefore, the multi-field coupling model of PMSM DT can be described as

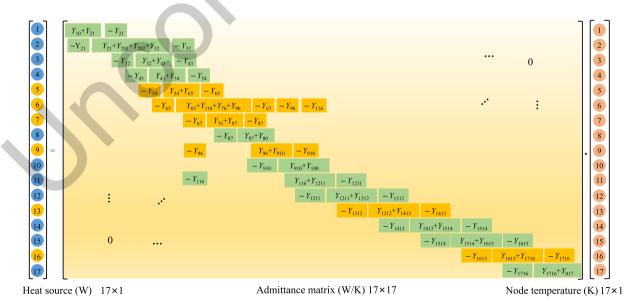


Fig. 9 Admittance matrix of entire thermal network of PMSM.

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$$\boldsymbol{T} = [Y_{**}] \cdot \begin{bmatrix} P_{\text{Cu}}([\tau(\psi_f, R, L_d, L_q, P_n, B, J, \omega_r, T_m)] + (\delta_d, \delta_q)) \\ P_{\text{Iro}}([\tau(\psi_f, R, L_d, L_q, P_n, B, J, \omega_r, T_m)] + (\delta_d, \delta_q)) \\ P_{\text{Mag}}([\tau(\psi_f, R, L_d, L_q, P_n, B, J, \omega_r, T_m)] + (\delta_d, \delta_q)) \end{bmatrix}$$
(21)

862 where  $\tau(\cdot)$  represents the model of the PMSM, which includes electrical, control, and dynamic aspects (see Eq. (15));  $P_{Cu}$ ,  $P_{Iro}$ 863 and  $P_{Mag}$  represent the power loss at the given input and cor-864 865 responding operating conditions;  $[Y_{**}]$  represents the admittance inverse matrix of the PMSM, which is then used to 866 determine the temperature T of each node in the PMSM. In 867 Eq. (21), the dimensions of the matrices on the right side of 868 the equation need to be adjusted according to the actual struc-869 ture and specific conditions of the system. The heat source vec-870 tors provided are intended for illustrative purposes only. They 871 serve as examples to demonstrate the concept of heat sources 872 in the system. In practice, the actual heat source vectors would 873 be specific to the particular system be analyzed or 874 875 implemented.

In this paper, we establish a PMSM DT based on a combi-876 877 nation of model-driven and data-driven methods. In terms of 878 the electrical, control, power and temperature of the PMSM, 879 we use the model-driven way; as the power loss of the PMSM's mathematical model is nonlinear and complex, the results 880 under multiple working conditions are used to build data-881 driven power loss model. Through the combination of the 882 models of the two forms, DT can be further enriched to more 883 comprehensively reflect the multidisciplinary coupling opera-884 885 tion status of PMSM under different working conditions. and provide a more reliable basis for subsequent analysis 886 887 and decision making.

#### 5.2. Dynamic update mechanism of PMSM DT 888

#### 5.2.1. Abnormal detection 889

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890 The electronic components within actual equipment deterio-891 rate or eventually fail as the operating time increases. To deter-892 mine potential abnormalities, this paper select the input power as a trigger index according to the Independence Principle. 893 Because the input power is neither the model's output nor an 894 internal critical parameter in the PMSM system, the trigger 895 is defined as the ratio of the power error between the initial 896 moment and the present moment, as shown in Eq. (22). 897 898

$$\operatorname{trig} = \frac{P_{\operatorname{entity}}^{0} - P_{\mathrm{DT}}}{P_{\operatorname{entity}}^{t} - P_{\mathrm{DT}}}$$
(22)

where  $P_{\text{entity}}^0$ ,  $P_{\text{entity}}^0$  and  $P_{\text{DT}}$  are the input power of the entity at 901 the initial time, time t and DT, separately; trig is abnormal 902 trigger. The input power of an entity PMSM is usually large 903 than the input power calculated by DT due to additional 904 power losses (e.g., air friction loss). 905

The increase in power resulting from demagnetization is 906 significantly higher compared to that caused by resistance 907 changes. When the set threshold is not met, the abnormal trig-908 ger will be initiated, indicating the need to update the DT. 909

#### 910 5.2.2. Dynamic update of DT

Online measurement is hard when used to update the flux link-911 912 age value, and the magnetic field sensor is expensive. Moreover, the compact structure of PMSM is not convenient to measurement.

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(21)

Therefore, we perform online parameter estimation for the d-q voltage model of PMSM by the RLS algorithm with forgetting factor. This paper adopts the RLS, and realizes the function of online dynamic integration update parameters. In surface-mount PMSM,  $L_d = L_a = L$ . When the PMSM works stably, the *d-q* voltage models of the PMSM are be written as

$$\begin{pmatrix} u_d = -\omega_e l_q L + \delta_d \\ u_q = \begin{bmatrix} i_q & -\omega_e \end{bmatrix} \begin{bmatrix} R \\ \psi_f \end{bmatrix} + \delta_q$$

$$(23)$$

According to the updated RLS Eq. (8), in the d-axis formula,  $-\omega_e i_a$  corresponds to the  $Q_t$  term,  $u_d$  corresponds to the  $Z_t$ , term, and L is the  $\theta$  to be identified; in the q-axis formula,  $i_a$  and  $\omega_e$  correspond to the  $Q_t$  term,  $u_a$  corresponds to the  $Z_t$  term, and  $R, \psi_f$  are the  $\theta$  to be identified.  $V_t$  is the random indeterminate item ( $\delta_d$ ,  $\delta_a$ ).

Taking  $u_d$  (in Eq. (23)) as an example, the estimation algorithm of RLS parameters is

$$\begin{cases} \boldsymbol{K}_{t+1} = \boldsymbol{P}_{t+1} \left( -\omega_e i_q \right)^{\mathrm{T}} (t+1) \times \left[ \lambda + \left( -\omega_e i_q \right) \boldsymbol{P}_i \left( -\omega_e i_q \right)^{\mathrm{T}} \right]^{-1} \\ \boldsymbol{P}_{t+1} = \frac{1}{\lambda} \left[ \boldsymbol{P}_t - \boldsymbol{K}_{t+1} \left( -\omega_e i_q \right) \boldsymbol{P}_t \right] \\ \hat{\boldsymbol{L}}_{t+1} = \hat{\boldsymbol{L}}_t + \boldsymbol{K}_{t+1} \left[ U_d(t+1) - \left( -\omega_e i_q \right) \hat{\boldsymbol{L}}_t - \delta_q \right] \end{cases}$$

$$(24)$$

By setting the corresponding initial value of K, P and  $\lambda$ , an estimate of L in the current state can be obtained after iteration.

## 5.2.3. Consistent measurement

We define PMSM state vectors and a set of vectors as follows:

$$s_w = \begin{bmatrix} u_{qw}, u_{dw}, i_{qw} \end{bmatrix}^1$$

$$X = \begin{bmatrix} s_1, s_2, s_3, s_{\text{DT}} \end{bmatrix}$$
(25)
942

where  $s_w$  represents the measured state matrix of w measurements of  $u_a$ ,  $u_d$  and  $i_a$  of the PMSM entity;  $s_{DT}$  is the running state vector value of DT; X consists of DT and three entity state vectors under the same working condition. The mean of these three entity state vectors is defined as

$$\boldsymbol{\mu}_{\alpha} = \frac{1}{3} \left( \boldsymbol{s}_1 + \boldsymbol{s}_2 + \boldsymbol{s}_3 \right) \tag{26}$$

Therefore, the MD between PMSM and DT is

$$MD = \sqrt{(\boldsymbol{\mu}_{\alpha} - \boldsymbol{s}_{DT})^{T} \boldsymbol{D}(\boldsymbol{\mu}_{\alpha} - \boldsymbol{s}_{DT})}$$
(27) 954

## 5.2.4. Overall algorithm for PMSM DT dynamic update

There is not a standard for the selection of MD and trig values, which need to be determined according to the specific object and the different tolerance for DT deviation. In this paper, the upper limit value of MD is set to 2.5, and the lower limit value of trig is set to 0.97. The whole DT update algorithm is as shown as Algorithm 1.

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## 962 Algorithm 1. Online dynamic DT update algorithm of PMSM

Input  $U_{\text{phase}}$ ,  $I_{\text{phase}}$ , Output  $\psi_f$ , R, L, Initialization of  $P_{\rm DT}, P_{\rm entity}^0$  and trig =  $P_{\rm entity}^0$  –  $\frac{P_{\rm DT}}{P_{\rm antitu}^t - P_{\rm DT}}$ 1. Abnormal detection while trig < 0.97 do calculate power factor  $V \stackrel{\text{add}}{\leftarrow} \text{voltage}^2[i], C \stackrel{\text{add}}{\leftarrow}$ current<sup>2</sup>[*i*],  $P \stackrel{\text{add}}{\leftarrow} \text{voltage}[i] \times \text{current}[i], \omega_e = P_n \omega_r$ Calculate power factor using root mean square method  $V_{\rm rms} = \sqrt{V/N}, C_{\rm rms} = \sqrt{C/N}, P = P/N,$  $i_q = \sqrt{2}I_{\text{phase}}, \text{fact} = P/V_{\text{rms}}/C_{\text{rms}}$  $u_q = U_{\text{phase}}\cos(\text{fact}), u_d = U_{\text{phase}}\sqrt{1 - \cos^2(\text{fact})}$ 2. Parameter estimation for each  $u_a, u_d, i_a, \omega_e$  $u_d = -\omega_e i_q L$  $u_q = Ri_q + \omega_e \psi_f$ return  $\psi_f, R, L$ 3. Consistent Measurement  $s_1 = [u_{q1}, u_{d1}, i_{q1}]^{\mathrm{T}}, s_2 = [u_{q2}, u_{d2}, i_{q2}]^{\mathrm{T}}$  $s_3 = [u_{q3}, u_{d3}, i_{q3}]^{\mathrm{T}}, s_{\mathrm{DT}} = [u_{q\mathrm{DT}}, u_{d\mathrm{DT}}, i_{q\mathrm{DT}}]^{\mathrm{T}}$  $\boldsymbol{X} = [\boldsymbol{s}_1, \boldsymbol{s}_2, \boldsymbol{s}_3, \boldsymbol{s}_{\mathrm{DT}}]$  $D = \text{Cov}(X^{\text{T}})^{-1}, \mu_{\alpha} = (s_1 + s_2 + s_3)/3$  $MD = \sqrt{(\boldsymbol{\mu}_{\alpha} - \boldsymbol{s}_{DT})^{T} \boldsymbol{D}(\boldsymbol{\mu}_{\alpha} - \boldsymbol{s}_{DT})}$ if MD < 2.5 then  $P_{\text{Real}}^0 = P_{\text{Real}}^t$ break

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The working mechanism of multi-field coupling PMSM DT with dynamic update capability is shown in Fig. 10.

5.3. Verification of PMSM DT and its demagnetization trend
 prediction experiment

## 968 5.3.1. Establishment of a system of PMSM DT

969 In this paper, the DT system of PMSM is established and validation through the workflow of the Model in the Loop (MIL), Software in the Loop (SIL), and Hardware in the Loop (HIL). DT is encapsulated as a dynamic-link library (DLL) that can be run online. The main parameters of the PMSM this paper used are shown in Tables 1, 2, and 3.

The main equipment of the experiment bench is PMSM, torque sensor, rotation load, speed sensor and the PMSM DT PHM system (see Fig. 11, from left to right). The other devices are the rotation load controller, PMSM controller, current sensor, voltage sensor, capture card, and temperature sensors located at the PMSM shell.

# 5.3.2. Comparison of the electrical state of PMSM entity and DT

The voltage and current characteristics are important criteria to measure the similarity between the PMSM entity and its DT. In this paper, the electrical parameters of the PMSM with a torque of  $4 \text{ N} \cdot \text{m}$ ,  $6 \text{ N} \cdot \text{m}$  and  $8 \text{ N} \cdot \text{m}$  and a speed in the range from 1000 r/min to 1600 r/min are compared with its DT electrical parameters (see Fig. 12).

It can be seen from Fig. 12 that under the three torque conditions, the results of the input line voltage of the PMSM entity and the DT are very close, and the line current of the PMSM entity and the DT have the almost the same error under the same working conditions. Although there is a certain error, we can still consider the DT to be reliable.

## 5.3.3. Comparison of the thermal state of PMSM entity and DT

In order to further verify the correctness of the thermal model, the same working conditions (air temperature 25 °C) was simulated by ANSYS Motor-CAD software (see Fig. 13). Finally, we compared the thermal model, the FEM simulation and the sensor measurement results (see Fig. 14).

Due to the limitation of the space structure of PMSM, the temperature sensor is arranged at the shaft and the shell. From Fig. 14, we noticed that the calculation results of the thermal model are generally consistent with the results of the FEM analysis, which shows the accuracy of the thermal model. By comparing the electrical and temperature parameters of the entity and its DT, the correctness of the multi-field coupling DT of PMSM is proved.

## 5.3.4. Comparison of PMSM DT update algorithm accuracy

Parameter estimation is an important part of DT update. The 1010 lack of model uncertainty ( $\delta_d$ ,  $\delta_q$ ) will introduce large errors 1011

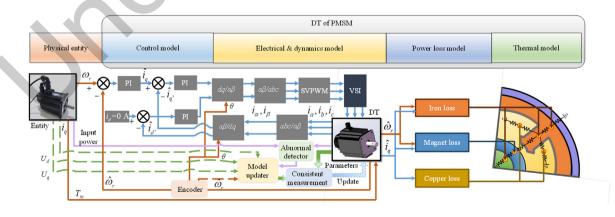


Fig. 10 Dynamically updated multi-field coupling DT of PMSM.

Table 1Size parameters of PMSM.

Parameter	Value
Number of pole pairs	4
Number of stator slots	36
Polar arc coefficient	0.85
Moment of inertia (kg·m <sup>2</sup> )	$1.94 \times 10^{-3}$
Stator outer diameter (m)	$1.22 \times 10^{-1}$
Stator inner diameter (m)	$7.8 \times 10^{-2}$
Rotor outer diameter (m)	$7 \times 10^{-2}$
Shaft diameter (m)	$2.2  imes 10^{-2}$
Magnet thickness (m)	$3 \times 10^{-3}$

<b>Table 2</b> Electrical parameters of PMSM.	
Parameter	Value
Stator resistance $(\Omega)$	$3.65  imes 10^{-1}$
Inductance (H)	$1.225 \times 10^{-3}$
Rotor flux (Wb)	$1.53 \times 10^{-1}$
Rated power (W)	$2.6 \times 10^{3}$
Rated current (A)	10
Rated rotational speed (r/min)	2500
Rated torque (N·m)	10

Table 3Components & Material.	
Component	Material
Stator and rotor	Silicon steel (DW315_50)
Permanent magnets	NdFeB_35
Wire winding	Copper
Shaft	Carbon steel

when using parameters identification method, which will affect the precision of DT. Therefore, the uncertainty of the model needs to be considered when identifying parameters.

By comparing the real value of  $\psi_f$ , *R*, *L* and the results obtained by using the RLS method that introduces model uncertainty, it is proved that this method can make highprecision estimation of entity (see Fig. 15). 1019

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# 5.3.5. Comparison of current change trend due to demagnetization

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According to actual engineering experience, the long-term operating current of PMSM shall not exceed 1.1–1.5 times of its rated current value. Monitoring the trend of current changes due to demagnetization is necessary to analyze the RUL of PMSM. Based on the above criteria, this paper predicts the demagnetization trend of the entity without taking into account mechanical failures. We can monitor the rotor flux condition of the entity at all times using PMSM DT. Under the same working condition (1200 r/min, 6 N  $\cdot$  m), we run the PMSM for a total of 800 h, and simultaneously record the line current of the PMSM (see Fig. 16).

Other recorded data from the experiment are presented in Table 4.

Since only the line voltage  $U_L$  and line current  $I_L$  can be measured from the actual PMSM, according to the constant amplitude relationship, here give the relationship between  $U_L$ ,  $I_L$  and  $U_q$ ,  $I_q$  as:

$$U_L = \frac{\sqrt{6}}{2} U_q$$

$$I_L = \frac{\sqrt{2}}{2} I_q$$
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It is evident that as the running time increases, the input current and power of both the entity and DT exhibit an increase. The voltage change in the entity does not show a distinct trend, whereas the voltage in the DT decreases. Due to the omission of various frictions present in the actual PMSM and the partial voltage of other components in the DT, there may be errors in the results for situations  $P_{\text{entity}}$  and  $P_{\text{DT}}$ . The parameters in Table 4 provide evidence that DT closely approximates the state of the entity.

By incorporating the abnormal factors observed in the experimental data, we can utilize Eq. (22) to analyze the demagnetization situation of the PMSM. Using the first correction as an example, the threshold calculated based on Eq. (28) using the abnormal indexes  $P_{\text{entity}}$  and  $P_{\text{DT}}$  is

$$\frac{P_{\text{entity}}^{l_1} - P_{\text{DT}}}{P_{\text{entity}}^{l_0} - P_{\text{DT}}} = \frac{790.014 - 855.790}{787.979 - 855.790} = 0.9699 < 0.97$$
(28)

At this juncture, Eq. (28) signifies that the DT has surpassed the allowable deviation from the entity, necessitating an update of the DT. The all three correction values and time of PMSM DT flux are recorded, as shown in Table 5.

Autoregressive Integrated Moving Average (ARIMA) is a popular method used for time series forecasting. Its underlying 1063

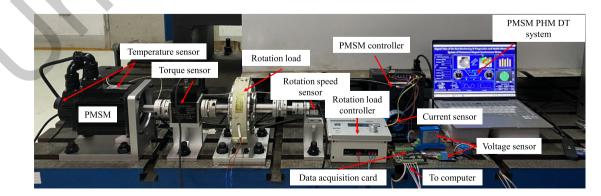


Fig. 11 Experiment bench of PMSM DT system.

Please cite this article in press as: GUO H et al. Dynamically updated digital twin for prognostics and health management: Application in permanent magnet synchronous motor, *Chin J Aeronaut* (2024), https://doi.org/10.1016/j.cja.2023.12.031

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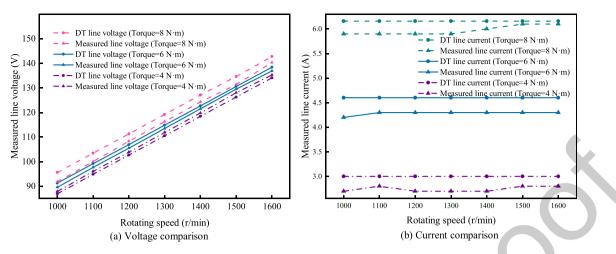


Fig. 12 Comparison of electrical parameters of PMSM entity and DT.

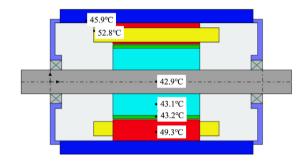
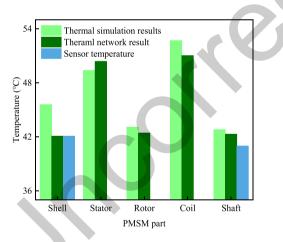


Fig. 13 Thermal simulation results of PMSM.



**Fig. 14** Comparison of temperature results of simulation, DT thermal model and sensors (1200 r/min, 6 N·m).

principle involves regression analysis using historical data and
principle involves regression analysis using historical data and
error data. The ARIMA requires the data to be stationary,
thus necessitating the application of differencing techniques
to the original current data. As the working time progresses
and vibration damage accumulates, the magnetism of the rotor
will undergo irreversible changes. The prediction result of the
ARIMA model based on the current signal is shown in Fig. 17.

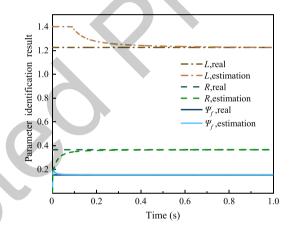


Fig. 15 Results of PMSM parameters identification.

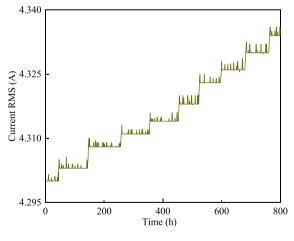


Fig. 16 RMS data of original current.

This model predicts the current from 690 to 810 h according to the ARIMA model. The predicted data are fitted in the form of a quadratic function as follows:

 $i_{\text{pred}}(t) = 6.215 \times 10^{-7} t^2 - 8.5 \times 10^{-4} t + 4.6177$  (29)

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Table 4	Correction p	parameters an	nd time of	PMSM	entity and 1	DT.
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Item	Runtime (h)	$U_L(\mathbf{V})$	$I_L(\mathbf{A})$	$P_{\text{entity}}(\mathbf{W})$	$\hat{U}_q(\mathbf{V})$	$\hat{I}_q(\mathbf{A})$	$P_{\rm DT}(W)$
$t_0$	0	105.800	4.300	787.979	87.290	6.536	855.790
$t_1$	259	105.803	4.311	790.014	87.120	6.550	855.954
<i>t</i> <sub>2</sub>	524	105.800	4.323	792.194	86.940	6.570	856.794
t <sub>3</sub>	764	105.873	4.335	794.943	86.770	6.588	857.460

Table 5Correction value and time of PMSM DT flux.				
Item	Value (Wb)	Runtime (h)		
Initial value	0.153	0		
First correction	0.1526	259		
Second correction	0.1522	524		
Third correction	0.1518	764		

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Curve fitting is also a method for approximating discrete data with analytical expressions. However, when fitting the original data, there may be several difficulties. For example, the signal collected by the sensor will be polluted by noise, and curve fitting is a challenge for data with complex changes. But the data obtained from the DT model is more accurate. According to the data of PMSM DT in Table 5, the demagnetization fitting curve of the PMSM DT is obtained as seen in Fig. 18.

In the experiments, due to the stability and reliability of PMSM, the consistency of DT and entity characteristics can be ensured only through parameter updates. Here, we represent the health state by using flux values as

$$H = 1 - rac{\Delta \psi_f^t}{\psi_f^{ ext{threshold}}} = 1 - rac{\psi_f^0 - \psi_f^t}{\psi_f^{ ext{threshold}}}$$

1093 where  $\Delta \psi_f^t$  is the flux variation value at time t,  $\psi_f^{\text{threshold}}$  is the 1094 threshold value of the flux,  $\psi_f^0$  is the initial flux value, and  $\psi_f^t$ 1095 is the flux value at time t. According to the design requirement, PMSM will malfunction if the demagnetization value exceeds 5%. In the case study, the initial flux value is  $\psi_f^0 = 0.153$  Wb, so the threshold value of the flux is  $\psi_f^{\text{threshold}} = 0.153 \times 0.05$  Wb. The flux value at current time is 0.1518 Wb. Therefore, the health status indicator can be obtained as follows:

cator can be obtained as follows:  

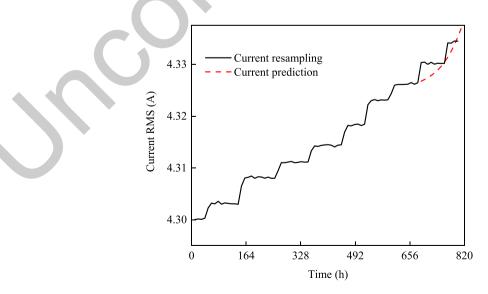
$$H = 1 - \frac{0.153 - 0.1518}{0.153 \times 0.05} \approx 0.843$$
(31)

At this point, through the update algorithm, the magnetic flux value in DT has been updated from the original 0.153 to 0.1518 Wb according to the PMSM current health state by using the RLS method. Therefore, based on assessment of the health factors, we consider the PMSM to be in a minor fault state. If the PMSM continues to operate, the flux will further degrade, and we can define it as in moderate fault or even more severe fault state based on the actual condition of the PMSM.

Here, we focus on the updating of demagnetization  $\psi_{DT}(t)$ from the normal state (H = 1) to the minor fault state (H = 0.843). If PMSM continues to operate, the corresponding DT needs update from time to time according to the health factors from the minor fault to moderate fault states  $(H < H_1)$ . Fig. 18 shows the updated DT form the normal to minor fault states.

After curve fitting, the demagnetization function of PMSM DT is shown as follows:

$$\psi_{\rm DT}(t) = -5.716 \times 10^{-10} t^2 - 1.3048 \times 10^{-6} t + 0.153 \tag{32}$$



(30)

Fig. 17 Prediction result with ARIMA model.

time series prediction error.

6. Conclusions

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**Declaration of competing interest** 

tional time series-based forecasting methods.

DT is obtained from only 4 sets of data. The data resource

consumption is reduced by 200 times. More importantly,

according to the error area calculation in Fig. 19, the error

of DT-based demagnetization trend prediction during this time

period is reduced by about 37% compared with the traditional

fitting forecasting and time series forecasting in both accuracy

and efficiency. Compared with the traditional maintenance

(1) This paper proposes a dual loop DT based PHM frame-

(2) A structure of first-principle dynamic DT is proposed

(3) The "Independence Principle" is proposed to select

(4) The dynamic multi-field coupling DT of PMSM is estab-

appropriate trigger between DT and physical entity dur-

lished. Experimental results show that the DT-based

PHM approach reduces the error in degradation predic-

tion results by approximately 37% compared to tradi-

state DT and degradation DT of the product.

under the normal condition.

ing lifecycle degradation.

work to account for the differences between the normal

method, consumption of computing resources is reduced.

To sum up, the PHM based on DT is superior to traditional

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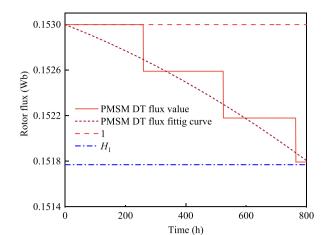
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The authors declare that they have no known competing 1174 financial interests or personal relationships that could have 1175 appeared to influence the work reported in this paper. 1176

This study was co-supported by the National Natural Science 1178 Foundation of China (Nos. U2233212; 51875014), Beijing 1179 Natural Science Foundation, China (No. L221008), and the 1180 China Scholarship Council (No. 202106020001). 1181

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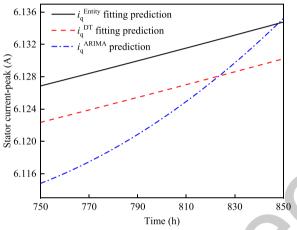
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CJA 2896

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PMSM DT demagnetization and its fitting curve. Fig. 18



Comparison of predicted results of current with three Fig. 19 methods.

We also fit the current original data. The fitting curve of the data collected from the entity is

$$0 i_{\text{Entity}}(t) = 1.659 \times 10^{-8} t^2 + 2.935 \times 10^{-5} t + 4.301 (33)$$

The control model in PMSM uses a constant amplitude 1131 transformation for three-phase alternating current. The entity 1132 acquisition data and the data used for ARIMA prediction are 1133 the RMS values of the current signal. Therefore, the  $i_q$  in DT is 1134  $\sqrt{2}$  times that of the RMS value of the current acquired by the 1135 1136 sensor. According to the relationship between the current and 1137 the rotor flux in Eq. (16), the trend curve of  $i_a$  can be obtained in the three different methods shown as follows: 1138 1139

$$\begin{cases}
\dot{i}_{q}^{\text{ARIMA}}(t) = \sqrt{2}i_{\text{pred}}(t) \\
\dot{i}_{q}^{\text{DT}}(t) = \frac{T_{e}}{1.5P_{n}\psi_{\text{DT}}(t)} \\
\dot{i}_{q}^{\text{Entity}}(t) = \sqrt{2}i_{\text{Entity}}(t)
\end{cases} (34)$$

Finally, comparison of the prediction results of  $i_q$  with the 1142 three different methods in the interval of 750-850 h is per-1143 formed (see Fig. 19). 1144

As shown in Fig. 19,  $i_q^{\text{DT}}$  is closer to the actual curve. More 1145 importantly, the high-precision prediction curve obtained by 1146

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$$i_{\text{Entity}}(t) = 1.659 \times 10^{-8} t^2 + 2.935 \times 10^{-5} t + 4.301$$
 (33)

Acknowledgements

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