

High Albedo Interlocking Concrete Block Pavement for Urban Heat Island Mitigation

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# 1 **Investigation of a GDI injector with an innovative flowmeter for high-pressure transient flows**

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## 6 **Abstract**

7 A new flowmeter for high pressure flows, characterized by intense dynamic events, is presented. The  
8 flowmeter algorithm is based on the measurement of two pressure signals along the investigated pipe, and the  
9 flowrate is obtained through an ordinary differential equation that is obtained from the combination of the  
10 continuity and the momentum partial differential equations. This flowmeter has been applied to a GDI injector  
11 to monitor the flowrate that enters the injector during an injection event. A 1D numerical model of the GDI  
12 injector has been set up and validated, and the numerical outcomes have been used to confirm the consistency  
13 of the experimental results obtained from the new flowmeter, for both single and pilot-main injections. The  
14 internal dynamics of the injector has also been investigated, using both the 1D numerical tool and the  
15 innovative flowmeter. A possible feedback-control strategy has been set up to compensate for any inaccuracy  
16 of the injected mass by applying the flowmeter to the hydraulic high-pressure circuit.

## 17 **Keywords**

18 GDI injector; instantaneous flowrate measurement; high-pressure flow; innovative flowmeter; closed-loop  
19 control.

## 20 **1. Introduction**

21 The prediction of an instantaneous high-pressure flowrate in fluid power hydraulic circuits can only be  
22 obtained by developing numerical models in which all the main hydraulic, mechanical, and electromagnetic  
23 characteristics are taken into account. However, an extensive validation against experimental transients is  
24 required to guarantee an adequate accuracy of these models [1]. Such an approach suffers from the absence of  
25 a flowmeter featuring the following characteristics: reduced dimensions and non-invasiveness, an excellent  
26 dynamic response, and the ability to work under high-pressure levels.

27 Various technologies and operating principles have been developed for flowrate measurements. Turbine  
28 flowmeters are used to measure the frequency generated by the rotation of a rotor, which is directly  
29 proportional to the flowrate [2]. These steadily calibrated devices can be applied to measure unsteady flowrates  
30 [3], but only pulsating [4], [5] or intermittent [6] ones with a lower frequency than 1 kHz can be considered.

31 Orifice flowmeters are widely diffused in industry, due to their simplicity in terms of construction and  
32 installation [7], [8]; however, they are generally employed for steady-state flows [9] and particular correction  
33 curves have to be employed to modify the steady equation for pulsating flows [10].

34 Electromagnetic flowmeters are suitable for flows characterized by a pronounced frequency content, but the  
35 constraint on the required electrical conductivity of the measured fluid, which should be lower than 0.1 uS/cm,  
36 excludes the possibility of applying this category of flowmeters to diesel or gasoline engines, since both are  
37 characterized by a much lower conductivity than that of the aforementioned threshold [11].

38 The currently available Coriolis flowmeters can be applied for the measurement of a flowrate characterized by  
39 a pressure of up to around 400 bar [12], although modern diesel injection systems can reach pressure levels of  
40 up to 3000 bar [13], while the 500 bar level is the target for next-generation GDI systems [14]. However, the  
41 invasiveness of this family of devices may represent a limit to their application [15].

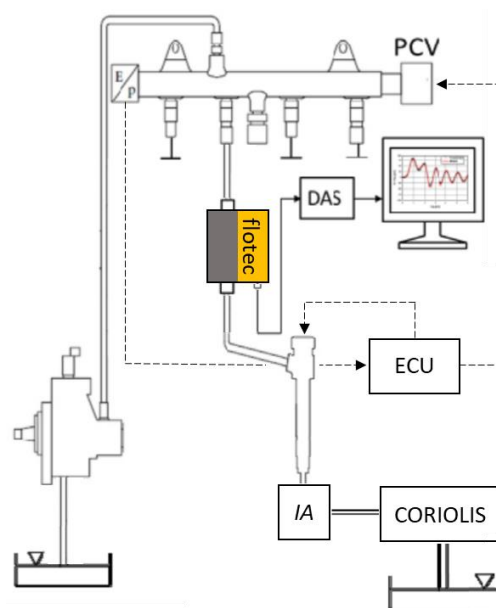
42 In this context, a flowrate estimation, based on the measurement of instantaneous pressure traces, represents a  
43 very attractive opportunity, due to the characteristics of the pressure sensors, in terms of miniaturization and  
44 dynamic response [16]. If the layout of the hydraulic system ensures that the pressure wave propagation occurs  
45 along a single direction, a mathematical relation between the measured pressure time history and that of the  
46 flowrate can be established for each pipe section [17]. However, if the pressure waves travel back and forth  
47 along the pipe, as in the rail-to-injector pipe of a Common Rail system [18], one pressure signal is not enough  
48 for the evaluation of the time distribution a consistent flowrate.

49 In the present investigation, an instantaneous high-pressure flowrate has been determined by measuring two  
50 pressure traces through the use of piezoresistive transducers. A first-order nonlinear ordinary differential  
51 equation has been obtained by combining Euler's mass conservation and momentum balance equations, where  
52 both the steady-state and the unsteady friction terms have been considered. The presented flowmeter, which  
53 has already been satisfactorily applied to Common Rail systems [19] (with pressures of up to 1800 bar) and to

54 volumetric pumps [20, 21] (with pressures in the 4-140 bar range), has here been applied to a GDI system,  
55 with the support of the 1D diagnostic tool, to investigate its internal fluid dynamics . Furthermore, a possible  
56 closed-loop control strategy of the injected mass, which is based on the application of the presented flowmeter  
57 to the fuel injection hydraulic circuit, is presented.

## 58 2. Experimental facilities

59 Tests were performed in the research laboratory at the Rabotti headquarter. The hydraulic test bench is a  
60 Rabotti TEC201 Evolution. It is composed of a conventional fuel injection system test stand featuring an  
61 electric motor, the speed of which can be controlled over the 0-4000 rpm range, a thermal regulation system  
62 of the tank and a low-pressure feeding circuit, which allows the high-pressure pump to be filled with  
63 pressurized oil over the 0-40 bar range. A typical Common Rail module is installed on this bench. This module  
64 contains the high-pressure pump, the rail, and some measurement devices (pressure and temperature sensors,  
65 Coriolis flowmeter, Injection Analyzer). The driving signals to the injector and to the rail/pump pressure  
66 control devices, i.e. the pressure control valve (PCV) and the fuel metering valve (FMV), are generated by an  
67 in-house developed control unit that offers the possibility of freely managing the injection pressure, the number  
68 of injections, the energizing time of each shot (ET), the dwell time between each shot (DT), the injection  
69 frequency and the configuration of the injector current profile (voltage, current levels and duration of each  
70 phase of the driving command).



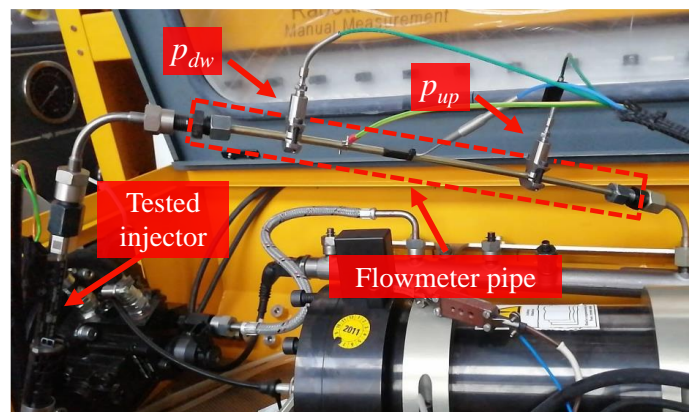
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72

Fig. 1 – Layout of the experimental apparatus.

73 Furthermore, it can be used in either manual mode, by introducing the test operative conditions and recording  
74 the test results manually, or automatically, using a pre-recorded test plan.

75 A scheme of the experimental layout is shown in Fig. 1. A high-pressure rotary pump feeds the rail to which  
76 the injector is connected, through an 840 mm long pipe, with an internal diameter,  $d$ , of 2 mm. This pipe is  
77 equipped with the new high-pressure flowmeter, which has been called “Flotec”. The flowmeter, represented  
78 as a colored box in Fig. 1 and in the photos in Fig. 2, is constituted by a piece of pipe onto which two  
79 piezoresistive pressure sensors are mounted at a distance  $l=200$  mm from each other. The upstream pressure  
80 sensor,  $p_{up}$ , is 320 mm far from the rail, while the pressure sensors  $p_{dw}$  is 200 mm far from  $p_{up}$  (and closer to  
81 the injector). This portion of pipe is installed between the rail and the tested injector, as shown in Fig. 2a,  
82 where the two pressure sensors constituting the flowmeter can be seen, while, in Fig. 2b, the flowmeter is  
83 presented with its external shell.



84  
85 (a) The flowmeter pipe equipped with the two pressure sensors.

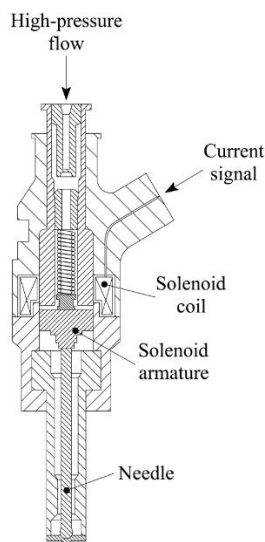


86  
87 (b) The flowmeter with its external shell.

88 Fig. 2 - Pictures of the Flotec flowmeter installed in the GDI injection system.

90 The employed piezoresistive transducers present an accuracy of  $\pm 1.25$  bar. Because of the difficulty of  
91 measuring the flow temperature (in order to evaluate the flow properties, such as density and dynamic  
92 viscosity), a thermocouple (accuracy  $\pm 2$  °C), which is directly linked to the fluid temperature, is used to  
93 measure the temperature of the external surface of the pipe between the two pressure sensors, and the flow  
94 temperature is thus obtained with a simple sub-model. The portion of the pipe in which the thermocouple is  
95 placed has been insulated in order to quickly reach a steady-state measurement, as the operative condition of  
96 the system is modified. The injected flowrate is measured by the Injection Analyzer, that is, a Bosch method-  
97 based flowmeter (labelled “IA” in Fig. 1, with an accuracy of  $< \pm 0.1$  mg/shot), in which the pressure signal is  
98 measured downstream of the injector tip and the injected fuel travels along a 11 m long pipe to prevent pressure  
99 wave reflections that could disturb the ongoing injection event; a Coriolis flowmeter, placed downstream of  
100 this pipe, measures the average flowrate ( $\leq \pm 0.1\%$  accuracy of the measured flowrate).

101 The GDI injector, a scheme of which is shown in Fig. 3, is normally closed, because the spring preload and  
102 the fuel pressure create a closure force. When the current signal acts on the injector, a magnetic force, generated  
103 at the solenoid, pushes the needle up until it reaches its stroke-end (the injector is not ballistic), thus making  
104 the injection start. As soon as the current signal is switched off, the magnetic force becomes null, the needle  
105 reaches the end of its downstroke and the nozzle closes, and this determines the end of the injection. Because  
106 of the absence of a pilot stage, the GDI injector can be assimilated to a fast two-way electro-valve with an inlet  
107 and one outlet port.



108

109

Fig. 3 – Scheme of the GDI Injector.

### 110 3. Flowmeter algorithm

111 The one-dimensional form of the partial differential equations of the continuity and momentum balance of a  
112 slender pipe can be written as [22]:

$$113 \begin{cases} \frac{d\rho}{dt} + \rho \frac{\partial u}{\partial x} = 0 \\ \frac{\partial u}{\partial t} + u \frac{\partial u}{\partial x} + \frac{1}{\rho} \frac{\partial p}{\partial x} = -\frac{4\tau_w}{\rho d} \end{cases} \quad (1)$$

114 where  $u$ ,  $p$ , and  $\rho$  indicate the fluid velocity, fluid pressure, and fluid density averaged over the pipe cross-  
115 section area, respectively;  $x$  stands for the spatial coordinate oriented along the axis of the pipe,  $t$  is the time,  
116 and  $\tau_w$  is the wall shear stress. The one-dimensional assumption is justified by the fact that the aspect ratio of  
117 the considered pipe portion is equal to  $l/d= 100$ , where  $l$  stands for the length of the pipe-portion between the  
118 Flotec pressure sensor locations. Furthermore, the accuracy of the unidimensional approach has been proved  
119 in previous research where 3D model results have been compared for some Flotec applications [20, 21].

120 It is assumed, through the hypothesis of a local incompressible fluid for the piece of pipeline between the two  
121 pressure transducers, that  $\frac{d\rho}{dt} \approx 0$ . Hence, the liquid density variations due to the change in the mean pressure  
122 level are considered, while the density fluctuations due to the pressure oscillations around the time-averaged  
123 pressure level are neglected. The consistency of this hypothesis can be verified when the Mach number, defined  
124 as the ratio of the flow velocity to the speed of sound velocity, is lower than 0.1 [23]. Since the speed of sound  
125 under the isothermal assumption is higher than 1000 m/s for gasoline, diesel oil, and mineral oils, the  
126 abovementioned condition is satisfied for a large number of fluid power applications, if cavitation is not  
127 experienced. Therefore, the continuity equation reduces to  $\frac{\partial u}{\partial x} = 0$ , and the momentum balance equation  
128 becomes

$$129 \frac{\partial u}{\partial t} + \frac{1}{\rho} \frac{\partial p}{\partial x} = -\frac{4\tau_w}{\rho d} \quad (2)$$

130 If Eq. 2 is multiplied by  $\rho A/l$ , where  $A$  represent the pipe cross-section area, by integrating over the distance  $l$ ,  
131 one obtains:

132 
$$\frac{d\bar{G}}{dt} = \frac{A}{l} \Delta p - \pi d \bar{\tau}_w \quad (3)$$

133 where  $G$  represents the mass flowrate and  $\Delta p = p_{up} - p_{dw}$ . The overlined quantities refer to the space-  
 134 averaged quantities along length  $l$ . By performing an integration of Eq. (3) with respect to time, one obtains  
 135 the space averaged flow-rate time history:

136 
$$\bar{G}(t) = \bar{G}_0 + \int_0^t \frac{A}{l} \Delta p dt - \pi d \int_0^t \bar{\tau}_w dt \quad (4)$$

137 where  $\bar{G}_0$  represents the initial value of  $\bar{G}(t)$ , which is usually unknown. By conducting further analytical  
 138 steps, detailed in [19], it is possible to rewrite Eq. 4 as

139 
$$\bar{G}(t) = \langle G \rangle + \frac{A}{l} \int_0^t [\Delta p - \langle \Delta p \rangle] dt - \pi d \Delta \bar{\Gamma}_{faf} \quad (5)$$

140 where the angular brackets indicate time-averaged quantities and  $\Delta \bar{\Gamma}_{faf}$  is a function that depends on the  
 141 unsteady friction [19]. As can be inferred from Eq. (5), the instantaneous mass flowrate can be determined by  
 142 measuring the time-average flowrate  $\langle G \rangle$  (e.g., by means of a Coriolis flowmeter) and the pressure difference,  
 143  $\Delta p$ , along the considered pipe.

#### 144 **4. Injector numerical model**

145 A 1D numerical model of the injector has been developed together with its feeding pipe and validated using  
 146 numerous experimental data for both single and pilot-main injections, over a range of nominal rail pressure  
 147 levels ( $p_{nom}$ ) from 80 bar to 150 bar. This numerical diagnostic tool considers the main characteristics of the  
 148 high-pressure hydraulic circuit, the electromagnetic driving circuit and the mechanical components. Figure 4  
 149 shows a schematic of the injector numerical model. Generally, the circuit is constituted by OD chambers linked  
 150 together by means of 1D pipes [24]. The Flotec pressure signal closest to the rail ( $p_{up}$ ), as well as the current  
 151 ( $I$ ) and the voltage ( $V$ ) signals to the solenoid were selected as boundary conditions. By selecting this set of  
 152 boundary conditions, it was possible to avoid any inaccuracy in the modelling of the rail, the pump, and the  
 153 ECU [25]. The fluid is assumed as isothermal, and the energy equation therefore reduces to a state equation,  
 154 and the pipe model implements the generalized Euler partial differential equations, which are solved by means  
 155 of a Lax-Wendroff numerical method. The moving elements are governed by the ordinary differential form of

156 Newton's law, according to a mass-spring-damper approach. The forces involved in the numerical model can  
 157 be ascribed to hydraulic, mechanical, and electromagnetic phenomena. In particular, the electromagnetic force,  
 158  $F_E$ , acting on the needle was evaluated as [1]:

$$159 \quad F_E = -\frac{1}{2} \Phi^2 \frac{d\mathfrak{R}}{dx} \quad (6)$$

160 where  $\frac{d\mathfrak{R}}{dx}$  represents the variation in the circuit reluctance, namely  $\mathfrak{R}$ , with respect to the needle lift, and  $\Phi$  is  
 161 the magnetic flux, which can be determined as:

$$162 \quad \Phi = \frac{1}{S} \int_0^t [V(t) - RI(t)] dt \quad (7)$$

163 Quantities  $V$  and  $I$  stand for the voltage and current that are applied to the injector solenoid, while  $R$  and  $S$   
 164 represent the measured resistance and the number of solenoid windings, respectively. During the needle  
 165 movement, a viscous force is generated between the armature and the injector body, due to the presence of a  
 166 thin boundary layer. The flow in this small annular passage is laminar and follows Poiseuille's law. The viscous  
 167 friction force on the armature can therefore be evaluated as:

$$168 \quad F_{vis} = \frac{\mu \cdot L_c \cdot v \cdot \pi \cdot d_a}{\delta_r} \quad (8)$$

169 where  $d_a$  is the armature diameter,  $L_c$  is the length of the annular passage,  $\delta_r$  is the radial distance between the  
 170 armature and the injector holder,  $\mu$  is the flow dynamic viscosity, and  $v$  is the velocity of the needle.

171 The main geometrical features of the GDI injector, implemented in the numerical model, are reported in Table  
 172 1.

173 *Table 1 - Main geometrical features of the tested injector.*

| Property                                     | Value |
|--|-------|
| Number of holes [-]                          | 6     |
| Nozzle hole diameter [mm]                    | 0.22  |
| Needle length [mm]                           | 30.9  |
| Needle ball diameter [mm]                    | 2.7   |
| Armature diameter ( $d_a$ ) [mm]             | 10    |
| Annular passage length ( $L_c$ ) [mm]        | 4.45  |
| Armature radial distance ( $\delta_r$ ) [mm] | 0.12  |

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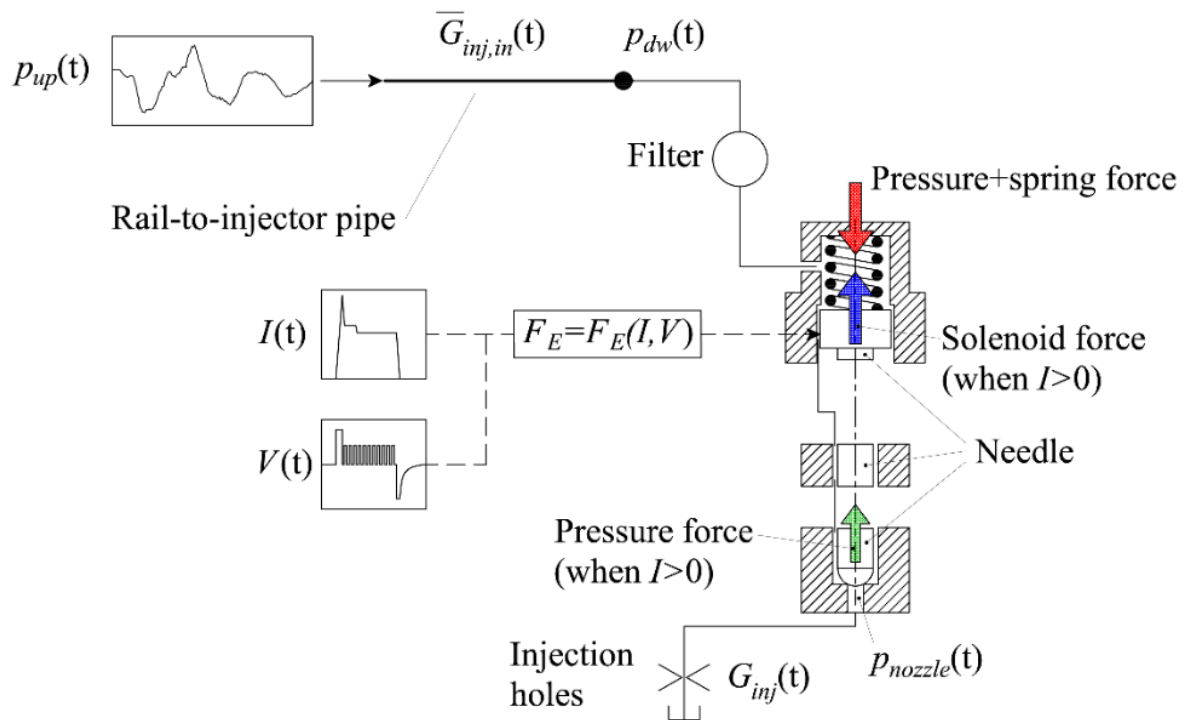


Fig. 4 - Scheme of the numerical model of the GDI injector.

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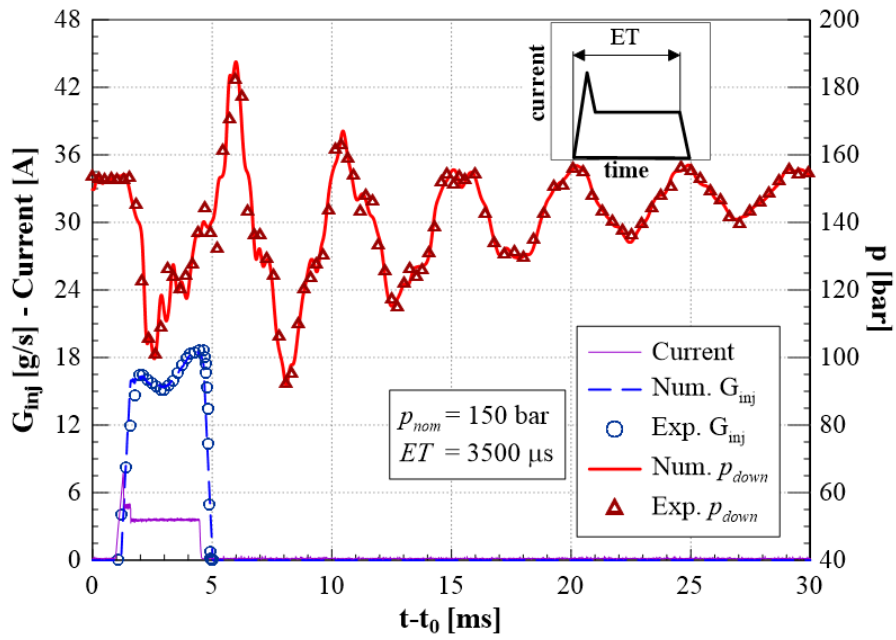
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180 The model validation involved a comparison of the numerical injected flowrate ( $G_{inj}$ ) and of the numerical  
 181 pressure time history ( $p_{dw}$ ) with the corresponding experimental traces, for both single and pilot-main  
 182 injections.

183 Furthermore, the numerical injector characteristic curves for different rail pressure levels over the previously  
 184 mentioned range were determined and compared with the experimental ones.

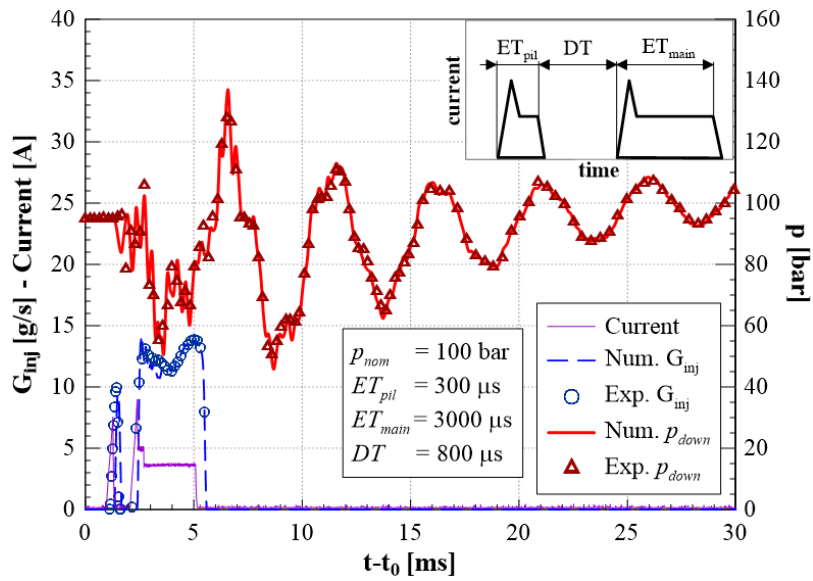
185 Figures 5 and 6 show the comparisons of the  $p_{dw}$  and  $G_{inj}$  time histories for a single and a pilot-main injection,  
 186 respectively: the symbols identify the experimental traces, while the lines refer to the numerical outcomes. As  
 187 can be inferred, the numerical results match the experimental ones perfectly, for both cases. This satisfactory  
 188 performance of the numerical model was verified over several  $p_{nom} - ET$  working points for single injections,  
 189 and for several  $p_{nom} - DT - ET_{main} - ET_{pil}$  sets for pilot-main injections.



190

191

Fig. 5 - Comparison of the numerical and the experimental results for a single injection.



192

193

Fig. 6 - Comparison of the numerical and the experimental results for a pilot-main injection.

194

195 The comparison between the numerical and experimental injector characteristics is reported in Fig. 7 for  
 196 different  $p_{nom}$  levels (80 bar, 100 bar, 120 bar and 150 bar). The agreement between the numerical and the  
 197 experimental results is satisfactory, since the maximum error between the experimental injected mass and the  
 198 numerical one (such a difference includes the experimental inaccuracy) is less than 1.7 mg, a result that further  
 199 validates the model.

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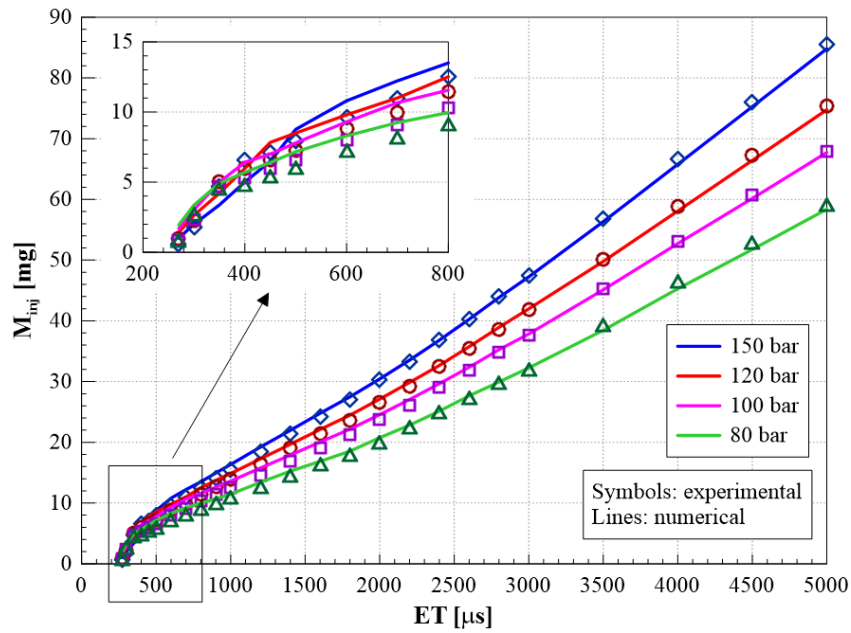


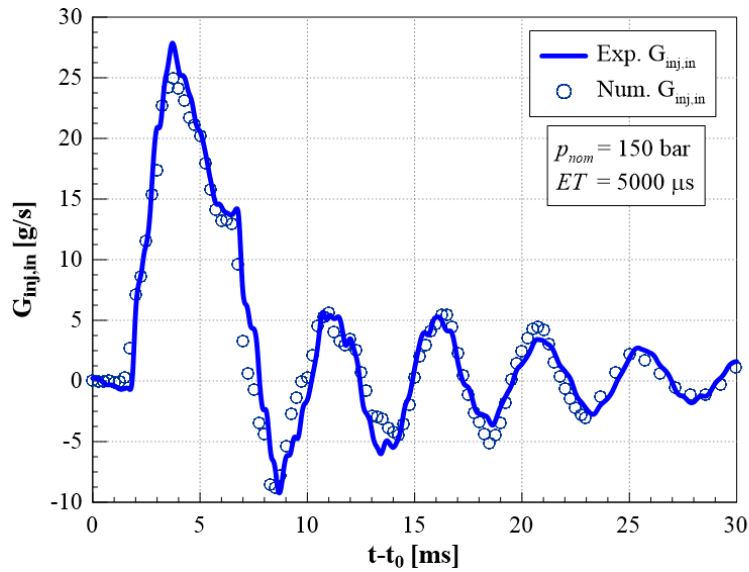
Fig. 7 – Numerical and experimental injector characteristics.

## 5. Experimental and numerical results

The numerical flowrate through the injector feeding pipe was determined from the 1D diagnostic tool (cf. Fig. 4). This numerical flowrate was compared with the experimental one measured with the flowmeter, and the results of this comparison are reported in Figs. 8 (single injection) and 9 (pilot-main injection): symbols are used to represent the numerical results and solid lines are utilized for the flowmeter data. As can be inferred from Figs. 8 and 9, the measured flowrates agree with those of the numerical diagnostic tool, and this further validates the new flowmeter. Since the compressibility of the flow is taken into account in the 1D numerical model, the good agreement shown in Figs. 8 and 9 demonstrates the consistency of the incompressible flow assumptions considered for the determination of the flowmeter.

From Fig. 7, it can be seen that the slope of the injector characteristic changes with respect to  $ET$ . The injected mass sharply increases with  $ET$  over the  $270 \mu s \div 550 \mu s$   $ET$  range. Figure 10 reports the normalized needle lifts ( $N/N_{max}$ ) obtained using the numerical tool, the experimental injected flowrates, and the numerical pressure trace inside the nozzle ( $p_{nozzle}$ , cf. Fig. 4), for different  $ET$  values at  $p_{nom} = 150$  bar. As can be seen in Fig. 10, when  $ET = 350 \mu s$  or  $ET = 450 \mu s$ , the injected flowrate is controlled by the needle seat area (the needle does not reach the end of its upper stroke over the considered  $ET$  range). When the needle reaches the end of its upper stroke ( $ET \approx 550 \mu s$ ), the injected flowrate presents a local maximum and is controlled by the pressure drop

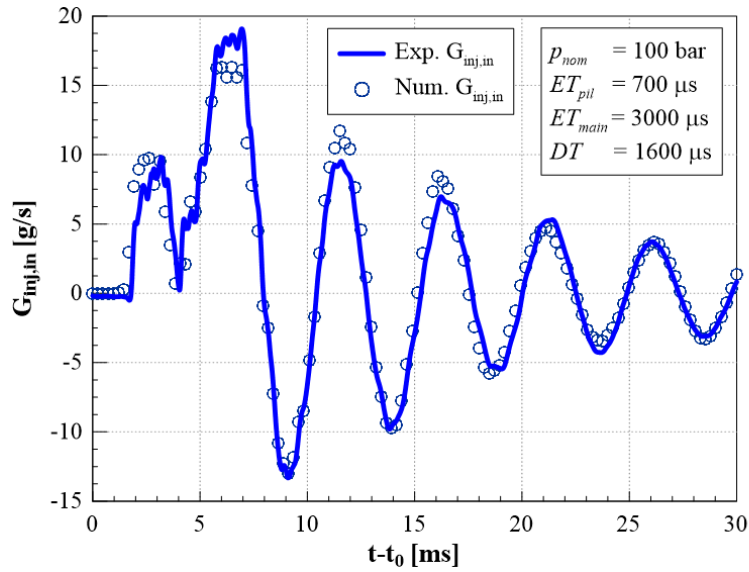
219 through the nozzle holes (cf.  $ET= 600 \mu\text{s}$  and  $ET= 1200 \mu\text{s}$  in Fig. 10). This explains the sharp augment in the  
 220 injected mass over the  $270 \mu\text{s} \div 550 \mu\text{s}$   $ET$  range.



221

222

Fig. 8 - Comparison of the numerical and the measured injector-inlet flowrates for a single injection.



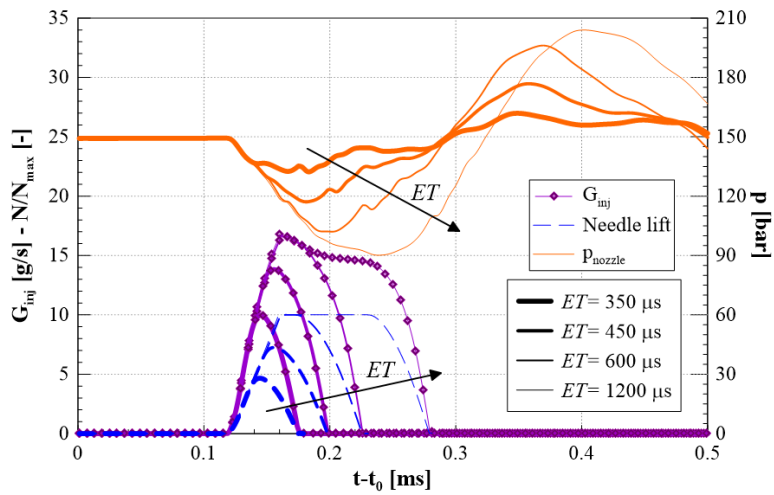
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Fig. 9 - Comparison of the numerical and the measured injector-inlet flowrates for a pilot-main injection.

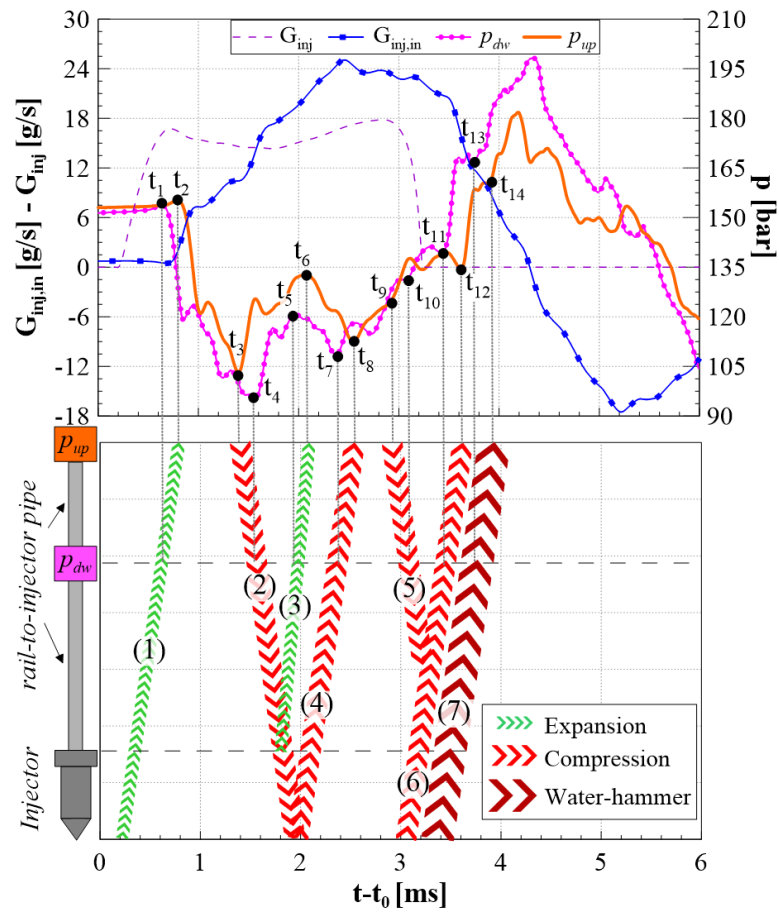
226 Furthermore, the  $ET-M_{inj}$  characteristic slope increases more when  $ET > 2200 \mu\text{s}$  than over the  
 227  $ET= 550 \mu\text{s} \div 2200 \mu\text{s}$  range, for all the considered rail pressure levels (cf. Fig. 7). This phenomenon can be  
 228 explained by considering the measured flowrate entering the injector and by analysing the propagation of the  
 229 pressure waves throughout the high-pressure circuit of the fuel injection system [26].



230

231

Fig. 10 - Needle lift, injected flowrate and nozzle pressure for different ETs.



232

233

Fig. 11 – Propagation of the pressure waves in the hydraulic circuit ( $p_{nom} = 150 \text{ bar}$ ,  $ET = 2800 \mu\text{s}$ ).

234

The upper part of Fig. 11 reports the measured flowrate at the injector inlet, the  $p_{up}$  and  $p_{dhw}$  pressure signals,

235

and the measured injected flowrate for an injection schedule of  $p_{nom} = 150 \text{ bar}$  and  $ET = 2800 \mu\text{s}$ , while a

236

schematic representation of the propagation of the pressure waves through the hydraulic circuit is presented in

237

the lower part, where the circuit is represented in scale on the left of the bottom graph and the vertical axis

238 reports a spatial coordinate. When the needle opens the nozzle and the injected flowrate start to rise, an  
239 expansion wave (1), which is the first wave experienced by the hydraulic system, travels from the injector to  
240 the rail and, consequently, a pressure drop can be observed in  $p_{dw}$  (at  $t=t_1$ ) and, after a certain delay, in  $p_{up}$  (at  
241  $t=t_2$ ). Thus, from these time instants, the first expansion wave travelling from the injector tip to the rail can be  
242 represented, as shown in the bottom graph of Fig. 11, between the spatial locations referring to  $p_{up}$  and  $p_{dw}$ .  
243 When this wave reaches the rail, it is reflected as a compression wave (2); the pressure rise in  $p_{up}$  ( $t=t_3$ ) occurs  
244 in advance, with respect to that in  $p_{dw}$  ( $t=t_4$ ), and this indicates that the pressure wave is travelling from the rail  
245 toward the injector. Such a compression can be extended up to the nozzle. Another reflection occurs in this  
246 hydraulic circuit when a wave reflected by the rail enters the injector, due to the difference in the diameter  
247 between the injector feeding pipe and the injector entrance. The presence of a pressure drop that first occurs  
248 in  $p_{dw}$  ( $t=t_5$ ) and then in  $p_{up}$  ( $t=t_6$ ) justifies the presence of a partial reflection, as an expansion wave (3) of the  
249 compression wave (2) that is entering the injector. Wave (3) can be represented in the graph when the  $t_5$  and  $t_6$   
250 time instants are known. When the compression wave (2) reaches the injector tip, it is reflected as a  
251 compression wave (4), which travels toward the rail, as the occurrence of the pressure rise first in  $p_{dw}$  ( $t=t_7$ )  
252 and then in  $p_{up}$  ( $t=t_8$ ) confirms. In the meanwhile, the expansion wave (3) has reached the rail and has been  
253 reflected as a compression wave (5) that travels toward the injector, thus implying a further pressure rise in  $p_{up}$   
254 ( $t=t_9$ ) and then in  $p_{dw}$  ( $t=t_{10}$ ). Prior to the reflection of this wave (5) at the injector entrance, the injected flowrate  
255 decreases as the needle starts its downward stroke and, when the entering flowrate exceeds the injected one, a  
256 compression wave (6) leaves the needle seat area and propagates upstream, thus leading to a pressure rise that  
257 is detected first in  $p_{dw}$  ( $t=t_{11}$ ) and then in  $p_{up}$  ( $t=t_{12}$ ). In a similar way to the first expansion wave (1), by  
258 identifying the time instants  $t_{11}$  and  $t_{12}$ , which refer to a compression wave moving from  $p_{dw}$  to  $p_{up}$ , and by  
259 extending this wave backward till it reaches the injector tip, it can be confirmed that it starts during the  
260 descending phase of  $G_{inj}$ , i.e., when the needle is moving downward. Finally, as the needle reaches the end of  
261 its downstroke (the injected flowrate is null), a water-hammer (7) is triggered in the nozzle and this drives an  
262 impulsive pressure rise in  $p_{dw}$  ( $t=t_{13}$ ) and then in  $p_{up}$  ( $t=t_{14}$ ). For conciseness reasons, the expansion wave given  
263 by the reflection of the compression wave (4) in the rail has been omitted from the figure (the time instants at  
264 which  $p_{up}$  and, subsequently,  $p_{dw}$  start to decrease can easily be identified between  $t_{10}$  and  $t_{11}$ ). It can be noted,  
265 from the pressure waves propagation motion schematized in Fig. 11, that, at the beginning of the injection, the

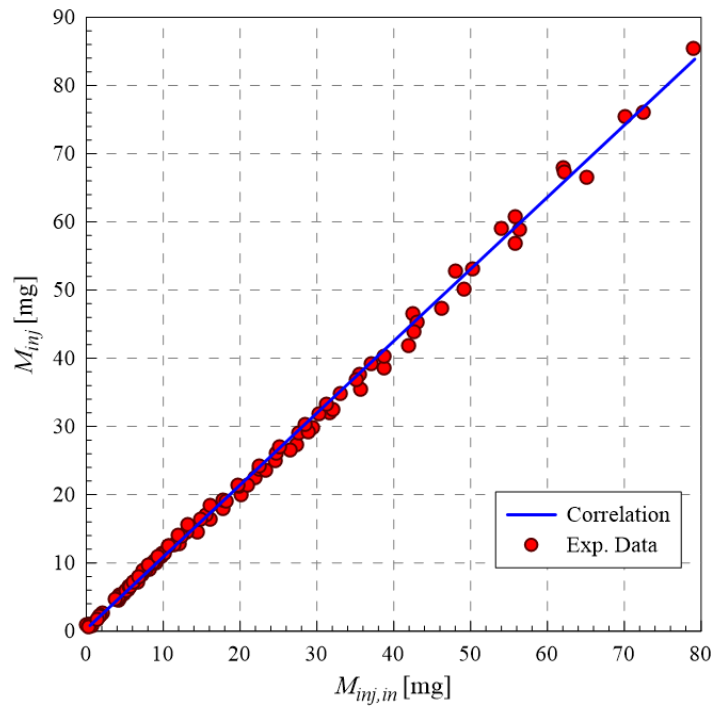
266 fuel in the rail-to-injector pipe is accelerated toward the injector nozzle, due to the presence of the  
267 expansion wave (1); the compression wave (2) then pushes the flow toward the nozzle, and, finally, even the  
268 expansion wave (3) tends to accelerate the injected fuel. Instead, the compression wave (4) opposes the fuel  
269 going toward the injector and, as a result, the measured flowrate features a maximum. For  $ET \geq 2200 \mu\text{s}$  (cf.  
270 Fig. 11), compression waves (2) and (5) help to push a flowrate into the injector, making the nozzle pressure  
271 rise monotonically with respect to the time and, consequently, the increased injected flowrate makes the slope  
272 of the  $ET-M_{inj}$  injector characteristics augment (cf. Fig. 7).  $ET = 2200 \mu\text{s}$  is the first value at which the  
273 compression wave (2) has almost recovered the initial pressure drop, induced by the start of the injection,  
274 within the nozzle.

275 As already presented in the literature [27]–[29], the mass entering the injector can be used to set up a closed-  
276 loop strategy of the injected mass. If the flowrate time history that enters the injector is integrated till it reaches  
277 its absolute minimum (cf. Fig. 8 at  $t = 8.5 \text{ ms}$ ), a robust correlation between the injected fuel quantity ( $M_{inj}$ ) and  
278 the mass that enters the injector during an injection event ( $M_{inj,in}$ ) can be obtained; such a correlation is reported  
279 as a continuous line in Fig. 12, where the experimental points are plotted with symbols. In [27]–[29], the mass  
280 entering the diesel Common Rail injector,  $M_{inj,in}$ , was obtained by ending the integration of the injector inlet  
281 flowrate in correspondence to its absolute maximum value. This difference in the method can be ascribed to  
282 the fact that, since the GDI injector pressure working range is much smaller than that of a diesel Common Rail  
283 injector, the energizing times are generally much longer. Therefore, if one wants to select the absolute  
284 maximum of the flowrate to terminate the integration of large  $ET$ s, the fuel injection would still be underway  
285 at that time, and the obtained entering mass would not be fully representative of the injected one.

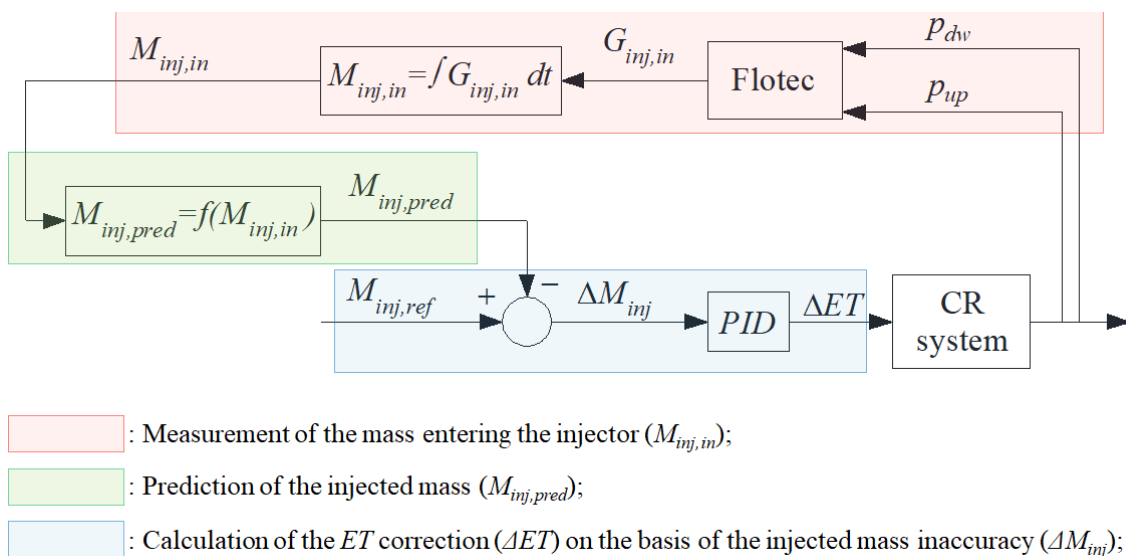
286 The scheme of a possible closed-loop control is represented in Fig. 13. During an injection event,  $p_{up}$  and  $p_{dw}$   
287 are measured by means of the Flotec flowmeter pressure sensors, which provide the flowrate that enters the  
288 injector, namely  $G_{inj,in}$  (cf. the red box in Fig. 13). The mass that enters the injector can be obtained by  
289 integrating  $G_{inj,in}$  with respect to time. This mass is used as input to obtain a prediction of the injected mass (cf.  
290 the green box in Fig. 13), on the basis of the previously-determined correlation  $M_{inj,pred} = f(M_{inj,in})$  represented  
291 in Fig. 12. The  $M_{inj,pred}$  value is then compared with the target value of the injected fuel quantity, i.e.  $M_{inj,ref}$ ,  
292 (this datum is stored in the ECU), and the  $\Delta M_{inj} = M_{inj,ref} - M_{inj,pred}$  difference is generated (cf. the blue box in  
293 Fig. 13). Such an estimation of the error in the injected mass is used as an input datum of a PID controller,

294 which gives an  $ET$  compensation, i.e.  $\Delta ET$ , as output in order to correct the possible inaccuracy of the fuel  
 295 quantity injected during the engine cycle.

296 Such a control can be used, for example, to correct the injected mass inaccuracy for a fixed couple of nominal  
 297 rail pressure level and energizing times, when the injector nozzle holes are affected by coking deposits [30],  
 298 [31].



299  
 300 Fig. 12- Correlation between the mass at the injector entrance and the injected one.



301  
 302 Fig. 13 - Scheme of the closed-loop control for the injected mass.

## 303 6. Conclusions

304 The high-pressure, instantaneous, liquid flowrate along a pipeline has been measured by employing an  
305 innovative flowmeter. This flowrate was obtained from two pressure signals,  $p_{up}$  and  $p_{dw}$ , measured at a  
306 distance of 200 mm from each other, along the investigated pipe. The flowmeter algorithm combines the  
307 momentum balance and the mass conservation equations for a 1D locally incompressible flow. The flowmeter  
308 was employed to measure the flowrate that enters a GDI injector during operation. The injector was fed through  
309 an 840 mm long pipe that had been equipped with the innovative flowmeter. A 1D numerical tool of the injector  
310 plus its feeding line was realized to further assess the consistency of the measured flowrate that entered the  
311 injector. This model receives the  $p_{up}$  pressure signal (used for the flowrate measurement) as a boundary  
312 condition and the current and voltage to the injector solenoid as input signals. The model was then validated  
313 successfully by means of comparisons between the numerical pressure time history at the location of the second  
314 pressure transducer ( $p_{dw}$ ) and the corresponding experimental trace, and by comparing the numerical injected  
315 flowrate and the experimental trace, measured by means of a Bosch method-based low-pressure flowmeter.  
316 Both single and pilot-main injections were considered, and the numerical outcomes accurately matched the  
317 experimental traces for different working conditions. The  $ET-M_{inj}$  injector characteristics showed two changes  
318 in slope. It has been shown, using the numerical model, that the needle does not reach the end of its upstroke  
319 over the first  $ET$  range ( $ET < 550 \mu s$ ), and this leads to a sharp increase in  $M_{inj}$  with  $ET$ . When  $ET > 550 \mu s$ , the  
320 flowrate starts to be controlled by the pressure drop across the nozzle holes, and the slope of the curve reduces.  
321 An increment in the characteristic slope can be observed when  $ET \geq 2200 \mu s$ . It has been shown that, at the  
322 beginning of the injection, the flowrate is accelerated toward the nozzle by an expansion wave that arises in  
323 the nozzle, and that it is finally pushed toward the injector by a compression wave that comes from the rail.  
324 When  $ET \geq 2200 \mu s$ , the nozzle pressure rises monotonically, due to a further flowrate that is pushed inside the  
325 injector by two other compression waves. This pressure rise leads to an increased injected flowrate that makes  
326 the slope of the injector characteristics augment. The flowrate that enters the injector can be integrated by  
327 selecting two arbitrary time instants to obtain a mass. In this case, the integration was performed from the start  
328 of the electric command up to the absolute minimum experienced by the measured injector-inlet flowrate. This  
329 entering mass can be correlated to the outlet mass to set up a closed-loop control that can be employed to  
330 modulate the  $ET$  during online injector operation. In fact, by means of the feedback signal of the mass that

331 enters the injector, a prediction of the effective injected fuel quantity can be determined on the basis of the  
332 previously determined correlation. Thus, by comparing the mass prediction with the target value stored in the  
333 ECU, the injected mass inaccuracy due, for example, to the presence of coking deposits on the nozzle holes or  
334 to thermal drifts, can be corrected.

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### 338 **Nomenclature**

|     |                      |  |
|-----|----------------------|--|
| 339 | <b>0D</b>            | zero-dimensional                             |
| 340 | <b>1D</b>            | one-dimensional                              |
| 341 | <b>3D</b>            | three-dimensional                            |
| 342 | <b>A</b>             | pipe cross-section area                      |
| 343 | <b>d</b>             | diameter, pipe diameter                      |
| 344 | <b>DT</b>            | dwel time                                    |
| 345 | <b>ET</b>            | energizing time                              |
| 346 | <b>F</b>             | force  |
| 347 | <b>FMV</b>           | fuel metering valve                          |
| 348 | <b>G</b>             | mass flowrate                                |
| 349 | <b>I</b>             | current signal                               |
| 350 | <b>IA</b>            | injection analyzed                           |
| 351 | <b>L<sub>c</sub></b> | length of the annular passage                |
| 352 | <b>l</b>             | distance between the Flotec pressure sensors |
| 353 | <b>M</b>             | mass   |
| 354 | <b>N</b>             | needle lift                                  |
| 355 | <b>p</b>             | flow pressure                                |
| 356 | <b>PCV</b>           | pressure control valve                       |
| 357 | <b>R</b>             | resistance                                   |
| 358 | <b>ℜ</b>             | reluctance                                   |
| 359 | <b>S</b>             | number of solenoid windings                  |
| 360 | <b>t</b>             | time   |

361  $u$  flow velocity  
362  $V$  voltage signal  
363  $v$  needle velocity  
364  $x$  spatial coordinate  
365  $\Delta ET$  energizing time correction  
366  $\Delta M_{inj}$  difference between the reference injected mass and the predicted one  
367  $\Delta \bar{\Gamma}_{faf}$  unsteady friction function  
368  $\Delta p$  pressure difference along the Flotec pipe  
369  $\delta_r$  radial distance between the armature and the injector holder  
370  $\Phi$  magnetic flux  
371  $\mu$  flow dynamic viscosity  
372  $\rho$  flow density  
373  $\tau_w$  wall shear stress  
374 **Subscripts**  
375  $0$  initial value  
376  $a$  armature  
377  $dw$  referring to the Flotec pressure sensor close to the injector  
378  $E$  electromagnetic  
379  $inj$  injected  
380  $inj, in$  entering the injector  
381  $main$  main injection  
382  $max$  maximum  
383  $nom$  nominal  
384  $nozzle$  referring to injector nozzle  
385  $pil$  pilot injection  
386  $pred$  prediction  
387  $ref$  reference  
388  $up$  referring to the Flotec pressure sensor close to the rail  
389  $vis$  viscous friction

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