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# *T*-stress effects on crack deflection: straight vs. curved crack advance

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## **Abstract**

*T*-stress effects on crack deflection in notched brittle materials are investigated by means of an average stress approach. It is shown that the crack advance always reveals to be straight if the *T* contribution is negligible. On the contrary, if *T* is different from zero, the trajectory tends to curve: the higher is  $|T|$ , the larger is the curvature parameter, its value being either positive or negative depending on the sign of *T*. Analogies and differences with respect to straight crack propagation are quantified in terms of the fracture loci and of the critical kinking angle. The investigation on the stability of crack deflection under mode *I* loading conditions concludes the paper, together with a discussion on experimental observations.

*Keywords:* average circumferential stress, *T*-stress, curved crack advance, mode *I* stability

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## 1. Introduction

The classical stress intensity factors (SIFs) govern the singular stress field in the neighborhood of the notch root, whereas higher order terms in William's asymptotic expansion affect the stress solution only at a certain distance from the crack tip. The first non-singular term is known as  $T$ -stress, which represents the constant term parallel to the crack plane. Indeed, its effects on crack deflection in homogenous brittle materials have been widely investigated since the beginning of sixties. Several different models were proposed by imposing either a strain statement, a stress threshold, an energy condition, or by coupling both a stress requirement and the energy balance (Erdogan and Sih, 1963; Cotterell, 1972; Williams and Ewing, 1972; Finnie and Saith, 1973; Carpinteri et al., 1979; He et al., 1991; Becker et al., 2001; Smith et al., 2001, 2006; Leguillon and Murer, 2008; Cornetti et al., 2014; Gupta et al., 2015; Sapora and Mantič, 2016). From the above studies, it emerges clearly that  $T$ -stress affects both the apparent fracture toughness and the critical kinking angle. Furthermore,  $T$ -influence reveals to be more significant for prevailing mode  $II$  conditions and as less brittle materials are considered. A special mention deserve geometries where the SIFs are both null, e.g. large plane slabs under remote uniaxial traction with a central through thickness crack collinear to the force direction (Williams and Ewing, 1972; Carpinteri et al., 1979), where  $T$  governs the whole stress solution.

As concerns the case of pure mode  $I$  cracks under biaxial loading, Cotterell and Rice (1980) proposed in their pioneering work, by a first-order perturbation scheme of complex potentials, that for  $T < 0$  the crack path is always stable, whereas for  $T > 0$  the propagation reveals to be unstable. Indeed, similar results had been obtained even before by Banichuk (1970); Goldstein and Salganik

(1970) through an analogous procedure. The dependence of the crack path stability only upon the sign of  $T$ -stress was object of a heated scientific discussion, rising from experimental tests showing the presence of a positive threshold  $T_{th} > 0$  above which instability occurs (Sumi et al., 1985; Selvarathinam and Goree, 1998; Chao et al., 2001). The problem was investigated by several studies: the point stress method was firstly exploited to evaluate  $T_{th}$  by Smith et al. (2001). On the basis of the studies presented in Broberg (1999), Melin (2002) pointed out that a critical role of the sign of the  $T$ -stress applies only to the situation of a single crack growing in a large plate and cannot be transferred to other situations directly. Leguillon and Murer (2008) evaluated  $T_{th}$  on the basis of the coupled criterion by Leguillon (2002). Focusing the attention on the  $T$ -stress variation after a small extension of the main crack (Li and Xu, 2007), Li et al. (2010) assumed that for positive  $T$  the crack firstly propagates in a stable manner until a critical crack growth distance, and then unstable propagation occurs. The estimate of  $T_{th}$  according to Finite Fracture Mechanics (FFM, Cornetti et al. (2014); Sapora and Mantič (2016)) differs somewhat from that by Leguillon and Murer (2008), although the two approaches provide a similar qualitative behavior. The strain energy density criterion predicts a value of  $T_{th}$  dependent also on the material Poisson's ratio (Ayatollahi et al., 2016). Finally, the role of  $T$ -stress on crack path is still object of investigation in the framework of dynamic propagation and of thermally driven fracture processes in quenched glass plates, although it seems unable to predict stability (Jayadevan et al., 2001; Yang and Ravi-Chandar, 2001).

Most of the approaches cited above refer to kinked or slightly curved cracks, which however were treated as perturbed straight cracks. Nevertheless, the presence of the curvature of the fracture plane was often detected experimentally,

even under mode  $I$  loading conditions (Radon et al., 1977; Ayatollahi et al., 2016): Radon et al. (1977) observed that ignoring the crack curvature for kinking angles up to  $30^\circ$  introduces errors of about 7% on the related failure load. The deviation increases up to 20% for angles close to  $50^\circ$ . Among the few (at least theoretical) attempts that were made to take the curvature into account, let us mention that by Karihaloo et al. (1981), who extended the first order perturbation scheme mentioned above to the second order perturbation approach, coming yet to the same conclusions by Cotterell and Rice (1980) on crack growth stability, and the subsequent works by Karihaloo (1982); Sumi et al. (1985); Amestoy and Leblond (1992); Broberg (1999); Salvadori (2008).

In the present paper, all these concepts will be revisited through the average stress criterion proposed by Seweryn (1998). The approach lies in the framework of the Theory of Critical Distances (Taylor, 2007), stating that failure takes place when a mechanical quantity (in this case the average circumferential stress) at a finite distance from the crack tip reaches a critical value. The finite distance (also known as crack extension or crack advance) results to be a material function, depending on the tensile strength  $\sigma_u$  and the fracture toughness  $K_{Ic}$ . For the sake of simplicity, the crack advance will be assumed of type  $y_1 \sim x_1^{3/2}$ ,  $(x_1, y_1)$  being the frame of reference obtained by rotating the cartesian system  $(x, y)$  by an angle  $\gamma$  (Fig. 1). The above function represents a simplified form of the one considered by Amestoy and Leblond (1992) (see also (Karihaloo, 1982; Sumi et al., 1985)), where it was stated that this shape must be necessarily assumed if the extensions are to be obtainable by crack propagation and not by arbitrary machining of the body. Before proceeding, it is important to underline that the present analysis will focus on the first crack advance stemming from the notch tip and it will not

describe the complete crack path.

## 2. Average stress criterion

According to the average stress criterion, failure takes place when the average circumferential stress  $\sigma_{\theta\theta}$  over a segment of length  $\Delta$  equals the tensile strength  $\sigma_u$ . Differently from FFM (Carpinteri et al., 2011), where the crack advance depends also on the geometrical features thus resulting to be a structural parameter, here  $\Delta$  depends merely on the material properties and it can be expressed as:

$$\Delta = \frac{2}{\pi} \left( \frac{K_{Ic}}{\sigma_u} \right)^2 = \frac{2}{\pi} l_{ch}, \quad (1)$$

$l_{ch}$  being the well-known Irwin's length. The value of  $\Delta$  expressed by Eq.(1) is not arbitrary, but it must be necessarily assumed in order to provide the Linear Elastic Fracture Mechanics (or Griffith) relationship  $K_I = K_{Ic}$  for a mode  $I$  specimen containing a sufficiently large crack. On the other hand, when comparing theoretical predictions with experimental results, the value of  $\sigma_u$  (and thus of  $\Delta$ ) is often fitted for some materials, such as polymers and metals. As a matter of fact, the experimental estimation on  $\sigma_u$  when testing plain specimens is strongly affected by the presence of micro-cracks/defects, and by crazing or plasticity phenomena (see, for instance, Taylor (2007)).

Although the stress criterion dates back to Neuber (1958) and Novozhilov (1969), it was formalized to V-notched elements under mode  $I$  loading conditions by Seweryn (1994), who later extended the approach to take  $T$ -stress effects and mixed-mode loading conditions into account (Seweryn, 1998; Seweryn and Lukaszewicz, 2002). Afterwards, several researchers applied this criterion over a wide range of geometries and materials (Taylor et al., 2004; Priel et al., 2007;

Ayatollahi and Torabi, 2010; Carpinteri et al., 2012; Sapora et al., 2013, 2014; Ayatollahi et al., 2015).

Let us recall the expression for the circumferential stress field in polar coordinates:

$$\sigma_{\theta\theta}(r, \theta) = \frac{K_I}{\sqrt{2\pi r}} f_{\theta\theta}^I(\theta) + \frac{K_{II}}{\sqrt{2\pi r}} f_{\theta\theta}^{II}(\theta) + T \sin^2 \theta, \quad (2)$$

where  $K_I$  and  $K_{II}$  are the SIFs related to the parent (or mother) crack,  $f_{\theta\theta}^I$  and  $f_{\theta\theta}^{II}$  are the corresponding angular functions

$$f_{\theta\theta}^I(\theta) = \cos^3(\theta/2) \quad f_{\theta\theta}^{II}(\theta) = -3\sin(\theta/2)\cos^2(\theta/2), \quad (3)$$

whereas the last term represents the  $T$ -stress contribution.

It is important also to remind that the relationships linking the cartesian frame of reference  $(x, y)$  to the polar frame of reference  $(r, \theta)$  can be expressed as

$$\begin{cases} \theta = \arctan\left(\frac{y}{x}\right) \\ r = \sqrt{x^2 + y^2} \end{cases} \quad (4)$$

The rotated frame of reference  $(x_1, y_1)$  is related to the cartesian frame of reference  $(x, y)$  by means of

$$\begin{cases} x = x_1 \cos(\gamma) - y_1 \sin(\gamma) \\ y = x_1 \sin(\gamma) + y_1 \cos(\gamma) \end{cases} \quad (5)$$

### 2.1. Straight crack propagation

Let us firstly assume that the kinked crack propagates (at least at the first step) straightly. By referring to Fig. 1 and imposing  $a = 0$ , the following conditions are

satisfied:  $\theta \equiv \gamma$ ,  $r \equiv x_1$ ,  $y_1 = 0$ . The average stress criterion can be expressed as:

$$\frac{1}{\Delta} \int_0^{\Delta} \sigma_{\theta\theta}(r, \theta) dr = \sigma_u \quad (6)$$

Substituting Eqs. (1) and (2) into (6) and supposing that failure takes place when  $K_I$  reaches its critical value  $K_{If}$ , yields:

$$\frac{K_{If}}{K_{Ic}} = \frac{1}{f'_{\theta\theta} + f''_{\theta\theta} \tan \psi + \tau \sin^2 \theta \sqrt{1 + \tan^2 \psi}}, \quad (7)$$

where the mode mixity  $\psi$  is defined as  $\tan \psi = K_{II}/K_I$  and  $\tau$  represents the dimensionless  $T$ -stress  $\tau = T \sqrt{l_{ch}} / \sqrt{K_I^2 + K_{II}^2}$  (also known as biaxiality ratio, (Leavers and Radon, 1982; Smith et al., 2001)). Once the loading conditions and the structural properties (i.e., the geometry and the material) are given, the  $\theta$ -value which maximizes the denominator at the right-hand side of Eq. (7) represents the critical kinking angle  $\theta_c$ . The corresponding value provides the critical mode  $I$  SIF  $K_{If}$ . Results are presented in Figs. 2 and 3 as concerns the critical kinking angle and the fracture loci, respectively. As it is expected, negative  $T$ -values decrease the circumferential stress  $\sigma_{\theta\theta}$ : both the failure load and the negative kinking angle increase. An opposite trend is recovered for positive  $T$ -stresses. Observe that for sufficiently high  $T > 0$ , the crack ceases to propagate collinearly under mode  $I$  loading conditions ( $\psi = 0^\circ$ ) and the corresponding critical mode  $I$  SIF deviates from the fracture toughness  $K_{Ic}$ . This point will be discussed in details in Section 3.

Similar curves to those depicted in Fig. 2 and 3 were obtained by implementing different fracture criteria, from the point stress model (Smith et al., 2001), which in this case provides identical predictions to those by the present average approach, to FFM (Cornetti et al., 2014), where a different definition of  $\tau$  was

adopted.

## 2.2. Curved crack propagation

Let us now suppose that fracture propagates (at least at the first step) by a curved crack advance described by the function  $y_1 = ax_1^{\frac{3}{2}}$  (Fig. 1),  $a$  being a curvature parameter. The crack path can be described in parametric form as  $\Gamma(t) = (t, at^{\frac{3}{2}})$ , and the average stress criterion (6) can be recast as:

$$\frac{1}{\Delta} \int_{\Gamma} \sigma_{\theta\theta} ds = \frac{1}{\Delta} \int_0^{t_1} \sigma_{\theta\theta}(r(t), \theta(t)) \sqrt{1 + \frac{9}{4}a^2 t} dt = \sigma_u \quad (8)$$

with

$$t_1 = \frac{4}{9a^2} \left[ \left( \frac{27}{8}a^2\Delta + 1 \right)^{\frac{2}{3}} - 1 \right] \quad (9)$$

Upon substitution of Eqs. (4) and (5) into the stress field (2), and of the corresponding expression into Eq. (8) it is possible to get a solution for the problem under investigation. The procedure consists in looking for the couple of critical parameters  $(a_c, \gamma_c)$  which minimizes the fracture load (i.e.,  $K_{If}$ ). Equation (8) together with (9) were solved numerically and results were then interpolated by means of spline functions.

From Fig.3 it can be seen that if  $T$  is different from zero, the curved crack always corresponds to a lower failure load with respect to the straight case. The difference is more appreciable for higher  $|\tau|$ . As concerns the curvature parameter  $a$ , it is a monotonic function in  $|\tau|$  and the dimensionless curves in Fig.4 show that: i) for negative  $T$ -stress,  $a$  is always positive. As a function of  $\psi$ , it increases from zero to a maximum (represented by circles in Fig.4), and then it decreases smoothly; ii)  $a$  is always negative for positive  $T$ -values. It attains a minimum in the neighborhood of mode I ( $\psi \leq 10^\circ$ ) and then it increases monotonically. As

mode *II* loading conditions are approached ( $\psi \rightarrow 90^\circ$ ), the difference between  $\tau$ -curves decreases. Thus, positive  $T$ -stresses influence more the curvature under prevailing mode *I* conditions ( $0^\circ \leq \psi \leq 45^\circ$ ), whereas it can be generally said that negative  $T$  affect more the curvature under prevailing mode *II* loading conditions ( $45^\circ \leq \psi \leq 90^\circ$ ).

Finally, the critical kinking angle for various  $\psi$  and  $\tau$  is represented in Fig. 5. The difference observed for both positive and negative  $T$ -values is not so significant when compared to the null  $T$ -stress. Indeed, the following explanation can be provided: the angle is governed by the parent mode mixity, as the  $T$  stress is negligible in comparison with singular stresses of the parent crack. On the other hand, as commented above, the kink curvature is strongly controlled by  $T$  stress which has influence on the kink-part more distant from the parent crack tip. Note also that the crack advance tends to curve (but not to kink,  $\gamma_c = 0^\circ$ ) for sufficiently high  $\tau$  even under mode *I* loading conditions ( $\psi = 0^\circ$ ).

In Table 1 the dimensionless  $d/l_{ch}$  distance between the end points of the crack advance related to straight and curved trajectories is reported for different  $\tau$  and  $\psi$ . The values of  $d$  keep always lower than the length of the crack advance  $\Delta$  (Eq.(1)), the condition  $d/\Delta < 0.25$  being always satisfied except for high positive  $\tau$ . Note also that when passing from  $\tau = 0.3$  to  $\tau = 0.4$  for mode *I* loading conditions ( $\psi = 0^\circ$ ),  $d$  diminishes due to the fact that for  $\tau = 0.3$  the extension tends to curve (Fig. 4) but not to kink (according to both straight and curved extensions), whereas for  $\tau = 0.4$  the straight extension assumption predicts  $\theta_c \neq 0^\circ$  (Fig.2).

### 3. Mode I loading conditions

Let us now focus our attention on pure mode I loading conditions ( $K_{II} = 0$ ). For a kinked crack, if  $T$  is sufficiently large,  $T \geq T_{th} > 0$ , the crack ceases to propagate collinearly and the critical SIF deviates from the fracture toughness. Predictions according to the average stress criterion are depicted in Figs. 6 and 7. The threshold  $\tau_{th}$  was estimated to be equal to 0.375, coherently with what found in Smith et al. (2001). The value differs from that obtained through FFM  $\tau_{th} = 0.420$  (Cornetti et al., 2014) and that provided by Leguillon's approach  $\tau_{th} = 0.704$  (Leguillon and Murer, 2008).

On the other hand, by assuming a curved crack, the critical SIF starts to deviate from the fracture toughness at a critical threshold  $\tau_{th}^a = 0.229 < \tau_{th}$  (Fig.6). Thus, once again, the failure load related to a curved crack propagation reveals to be lower (or equal) than that corresponding to a straight trajectory. The maximum percent deviation can be quantified in nearly 6% over the whole range of analyzed  $\tau$ . Furthermore, the trajectory starts to curve (with a null kinking angle, Fig. 7) for  $\tau \geq \tau_{th}^a$  (Fig. 8). The curvature parameter can assume either positive or negative values, so that  $\tau_{th}^a$  represents a bifurcation point. Note that  $a$  is a monotonic increasing function in  $\tau$ . Finally, the angle  $\gamma_c$  deviates from  $0^\circ$  at  $\tau_{th}^\gamma = 0.485$ : this threshold influences also the behavior of  $a$  (Fig. 8). Due to symmetry, both positive or negative kinking angles are possible, corresponding to positive and negative curvatures, respectively.

### 4. Conclusions and discussion of results

The influence of  $T$ -stress on predictions by the average circumferential stress criterion was analyzed by considering a curved crack advance. It was shown that

for  $T \neq 0$  a curved extension is always to be preferred than a straight one, since it corresponds to a lower failure load. Positive  $T$ -stresses originate a negative curvature parameter  $a$ , whereas negative  $T$ -stresses imply positive  $a$ . On the other hand, the variation of the kinking angle for curved cracks is less significant than for straight cracks: whereas  $(K_I, K_{II})$  govern the singular stress field in front of the notch tip and thus the kinking angle as a local parameter, the contribution of  $T$  allows to describe the situation far from the notch root, outlining the effect on the curvature behavior. Besides  $\tau$ , the curvature is also a function of the mode mixity and of the material. For mode  $I$  loading conditions and sufficiently high  $T$  ( $\tau \geq \tau_{th}^a = 0.229$ ), the critical SIF deviates from the fracture toughness and the trajectory starts to curve, but not to kink yet. Only for higher  $\tau$  ( $\tau \geq \tau_{th}^\gamma = 0.485$ ), the critical kinking angle assumes values different from  $0^\circ$ . These results differ from those related to a straight extension, which provides a higher failure load (the mean percent deviation being around 4 – 5%) and a unique bifurcation point ( $\tau_{th} = 0.375$ ).

In order to corroborate the present results, a comparison with experimental results would be suitable. Indeed, the task is not trivial due to many reasons. First of all, our analysis was limited to the propagation at the first step (the crack advance) and it did not cover a complete crack path. Secondly, most of the experimental works presented in the Literature considered the fracture load and the kinking angle, not measuring (although observed) the curvature. As a matter of fact, the distinction between the kinking angle and the curvature may be not a trivial task in the close vicinity of the crack tip. Eventually, the specific function chosen to model the present crack extension can be a limiting assumption.

Despite this, some general observations can be made. The function  $y = ax^b$

represents the typical shape of mode *II* cracks under compression ( $K_I = 0$ ), as described by Isaksson and Ståhle (2002). The investigation carried out in (Isaksson and Ståhle, 2002) on different tests on brittle rock-like materials (Shen et al., 1995; Bobet and Einstein, 1998) revealed an exponent  $b$  belonging to the range (1.43-1.58), and a curvature parameter comprised between 0.052 and 0.147. Since  $\tau$  is negative, this is in fair agreement with what depicted in Fig. 4. Indeed, a complete analysis would require the inclusion of two components of  $T$ -stress (parallel and perpendicular to the crack plane, respectively), the interaction between the crack faces in contact and friction (see (Li et al., 2009; Tang, 2015)). In Smith et al. (2006) specific tests under mode *II* loading conditions were performed to investigate  $T$ -stress effects: PMMA specimens were broken under a positive  $T$  corresponding to  $\tau_c \approx 0.26$  and negative  $T$  related to  $\tau_c \approx -0.4$ . The former specimens presented a slightly curved fracture, whereas in the second samples, the curvature was noteworthy. This is again described qualitatively by Fig. 4.

As concerns mode *I* loading conditions, Radon et al. (1977) performed experiments on biaxially stressed PMMA sheets, observing that the curvature is an increasing function of  $\tau$ , whereas the crack angle deviates from zero only above a threshold. Both positive or negative curvatures were measured, in correspondence to positive and negative kinking angles, respectively. These results agree with the present analysis (Section 3). Moreover, studies carried out in Selvarathinam and Goree (1998) and Chao et al. (2001) on PMMA specimens, showed very low thresholds for  $\tau_{th}$  ( $\tau_{th} \approx 0.07$  in Chao et al. (2001) assuming  $\sigma_u = 70\text{Mpa}$ , which differs significantly from that by the present criterion, and even more from those by the coupled criteria cited in Section 3). The curvature was not measured, but the observed initiation angles did not exceed the value of  $30^\circ$ . More recently, Ay-

atollahi et al. (2016) investigated the fracture trajectory of PMMA samples under pure mode I loading conditions. Five PMMA specimens of different shapes were tested, corresponding to  $\tau=0.06, 0.1, 0.22, 0.24,$  and  $0.25$ . The experimental critical kinking angle was zero for the former two geometries, and then it started to decrease. Note that  $\tau=0.22$  nearly coincides with the threshold  $\tau_{ih}^a$  computed in Section 3. Indeed, the curvature of the crack advance was not analyzed, but that related to the whole crack path was found to increase starting again from  $\tau = 0.22$ .

Finally, note that the present methodology can also be applied to predict onset of small curved cracks at singular points like single material and multi-material corners, an example being such cracks observed in experiments with bimaterial Brazilian disk specimens by Vicentini et al. (2012).

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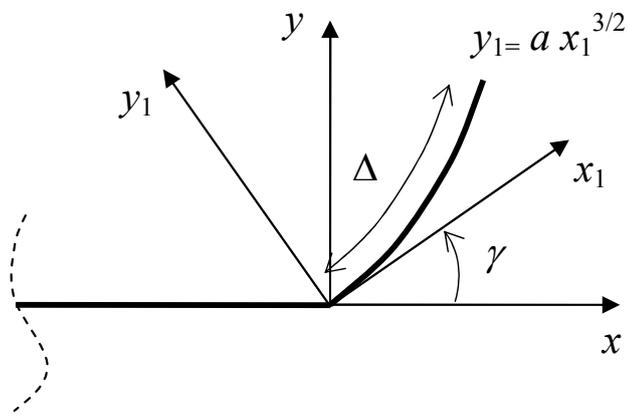


Figure 1: Cracked element with a kinked curvilinear crack of length  $\Delta$ .

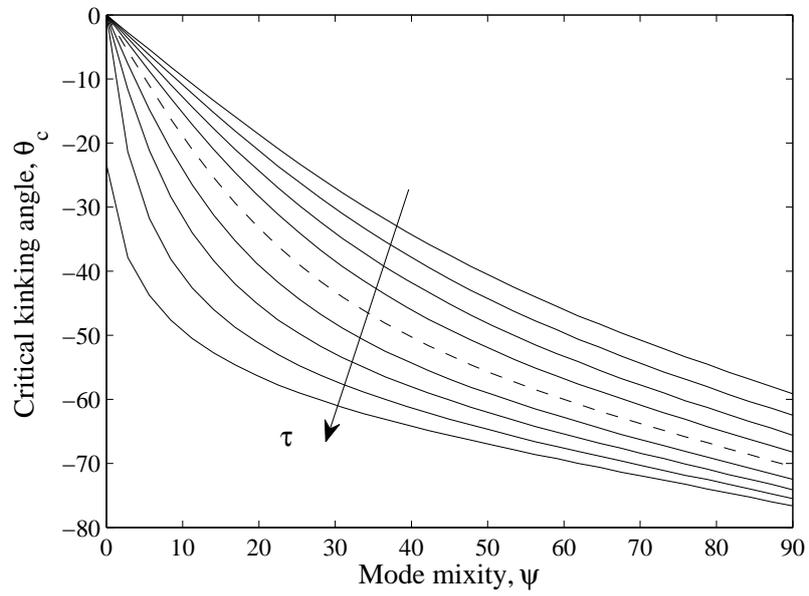


Figure 2: Straight crack propagation: effects of the dimensionless  $T$ -stress, ranging from -0.4 to 0.4 with a step equal to 0.1, on the critical kinking angle  $\theta_c$  (deg). The dashed line refers to the case  $\tau = 0$ .

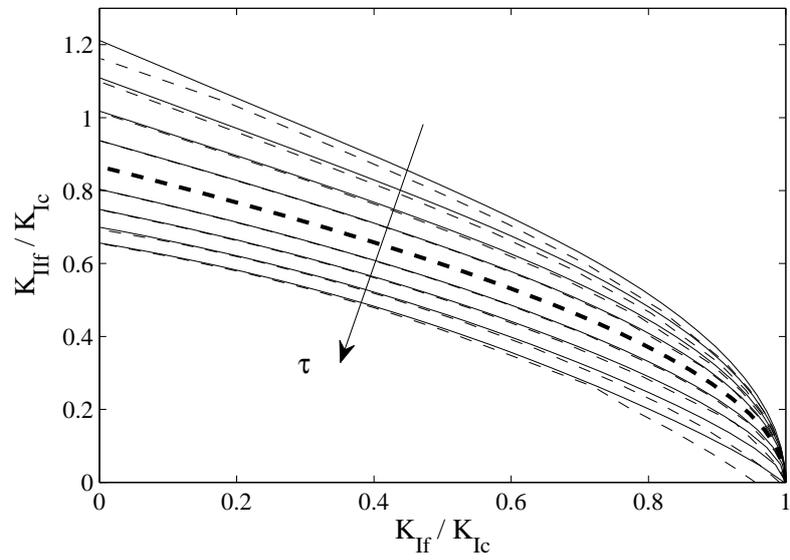


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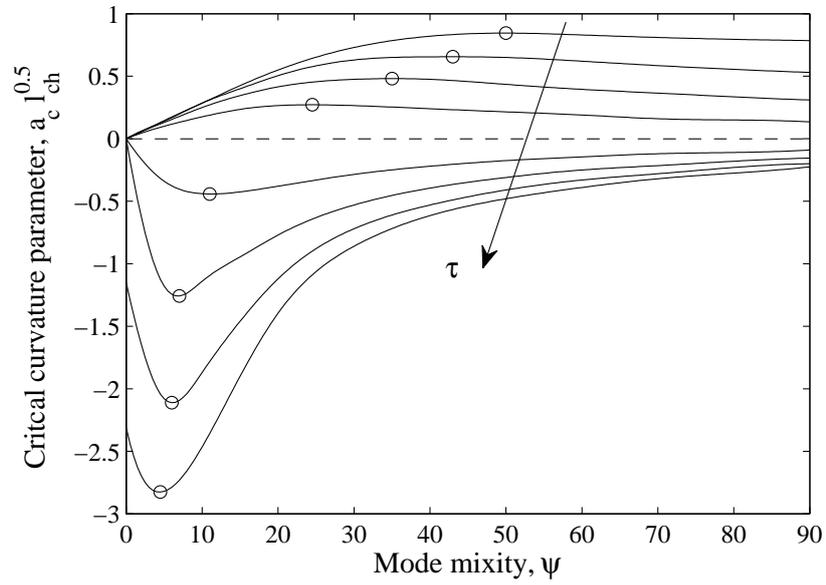


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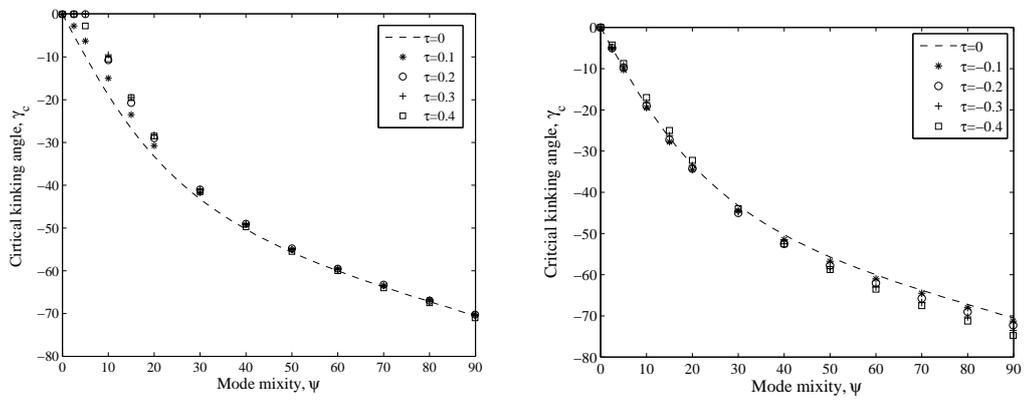


Figure 5: Effects of the dimensionless  $T$ -stress on the critical kinking angle  $\gamma_c$  (deg). The dashed line refers to the case  $\tau = 0$ .

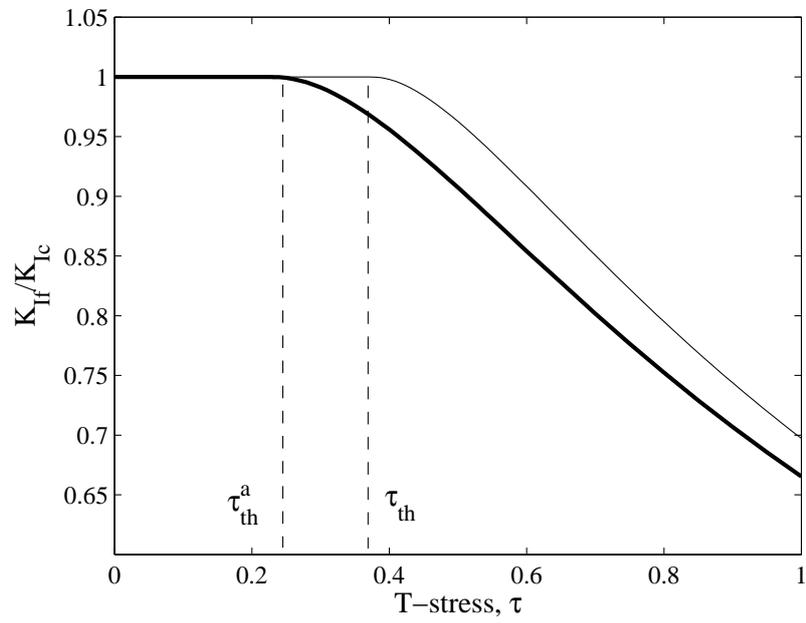


Figure 6: Average stress predictions for mode *I* loading conditions : *T*-stress effects on the critical the failure load according to a straight crack advance (continuous line) and a curved crack advance (thick continuous line).

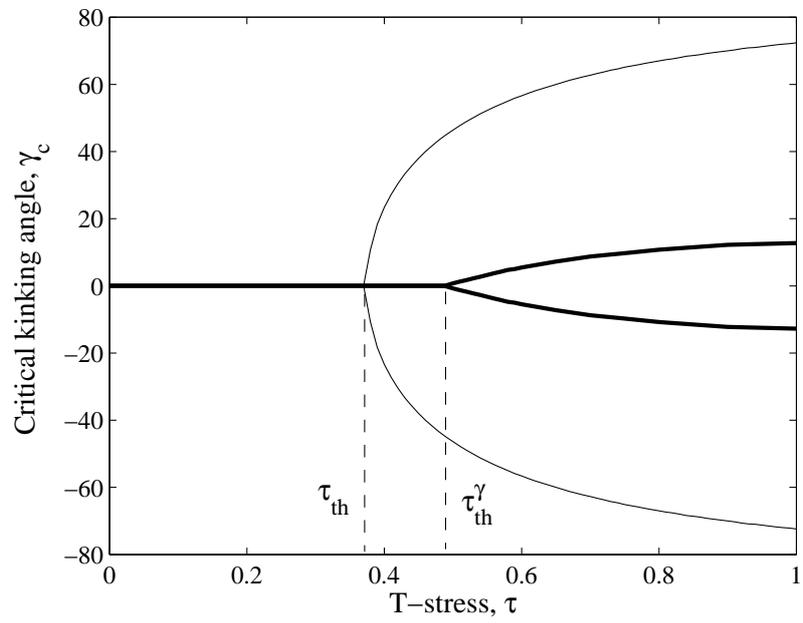


Figure 7: Average stress predictions for mode *I* loading conditions : *T*-stress effects on the critical kinking angle (deg) according to a straight crack advance (continuous line) and a curved crack advance (thick continuous line).

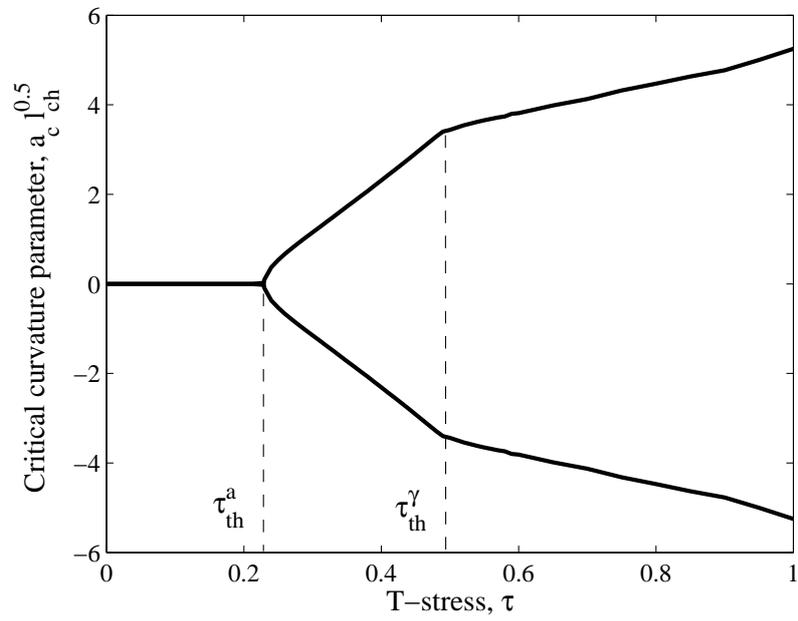


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**List of Tables**

1    Dimensionless distance  $d/l_{ch}$  between the end points of curved  
and straight crack advances. . . . . 30

$\psi \setminus \tau$	-0.4	-0.3	-0.2	-0.1	0.1	0.2	0.3	0.4
0°	0.000	0.000	0.000	0.000	0.000	0.000	0.419	0.337
30°	0.128	0.121	0.101	0.063	0.069	0.112	0.133	0.140
60°	0.152	0.129	0.092	0.048	0.035	0.059	0.075	0.081
90°	0.159	0.122	0.078	0.035	0.021	0.035	0.044	0.049

Table 1: Dimensionless distance  $d/l_{ch}$  between the end points of curved and straight crack advances.