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The effect of temperature and strain rate on the grain boundary sliding in a CM247 LC Ni-based superalloy processed with laser based powder bed fusion

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(Article begins on next page)

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The effect of temperature and strain rate on the grain boundary sliding in a CM247 LC Ni-based superalloy processed with Laser Based Powder Bed Fusion
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	Daniele Ugues
Abstract:	The current work explores the meso-scale deformation behaviour of an additively manufactured CM247 LC at room and high temperatures. In particular, the study focuses on assessing grain boundary sliding (GBS), which can play a crucial role in the high-temperature deformation of superalloys. Specific samples were produced using the Laser Based Powder Bed Fusion technique (PBF-LB), heat treated and tested under monotonic compression in a Gleeble® system. Compression tests were carried out in a wide temperature range at two strain rates and the effect of testing parameters on GBS activity was studied. A thorough microstructural characterization of the PBF-LB material using EBSD and TEM revealed a γ/γ microstructure consisting of columnar grains decorated with Hf-rich MC carbides without any segregations of alloying elements. Qualitative and quantitative analysis of GBS was performed using FEG-SEM and AFM, and contribution of GBS into plastic deformation was estimated. It was demonstrated that GBS is activated at 760°C. A direct correlation between the contribution of GBS into plastic deformation and testing temperature was found, while strain rate has the opposite effect. The highest GBS contribution (~32%) was recorded at 1093°C/10-3 s-1. Finally, intergranular microcracking at triple junctions and along grain boundaries was observed when the material was tested at the highest temperatures (871°C and 1093oC). The effect of the testing parameters on the GBS activity in the PBF-LB material is discussed.
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To the Editor-in-Chief of: Intermetallics

Date: 07/11/2023

Dear Editor,

I am a Ph.D. student at the Department of Applied Science and Technology (DISAT) of the Polytechnic of Turin, working on Ni-based superalloys processed via PBF-LB. I am pleased to submit a research article entitled: "The effect of temperature and strain rate on the grain boundary sliding in CM247 LC Ni-based superalloy processed with Laser Based Powder Bed Fusion" by Martelli et al. for consideration for publication in Intermetallics. This work was carried out in collaboration with IMDEA Materials Institute of Madrid.

This is the first article submitted from the research group focusing on the deformation mechanisms of CM247 LC superalloy processed by PBF-LB, whereas author Dr. Sabirov (IMDEA Materials Institute) has proven experience in the assessment of grain boundary sliding (GBS). The present study investigates the triggering of GBS during hot deformation through monotonic compression tests performed in a Gleeble® system. The tests were performed in a wide range of temperatures (from room temperature to 1093°C) and investigating the effect of two strain rates (10⁻³ – 10⁻⁴ s⁻¹). Scanning Electron Microscopy (SEM) and Atomic Force Microscopy (AFM) revealed the activation of GBS mechanism from 760°C, leading to intergranular microcracking at the highest temperatures (871°C and 1093°C). The result is crucial in understanding the creep behaviour of additively manufactured CM247 LC superalloy, because if the GBS is not accommodated it has proven to cause intergranular microcracking, which can lead to in-service drop in ductility and premature failure. Moreover, the influence of the pre-test microstructure is discussed, starting from the characterization made by Transmission Electron Microscopy (TEM) and Electron Backscattered Diffraction (EBSD). Beside the qualitative assessment of the activation of GBS at high temperature, the authors built a quantification scheme for the evaluation and estimation of its contribution to the total plastic deformation and its correlation to the testing parameters.

We believe that the present manuscript is suitable for *Intermetallics* because it investigates the activation of GBS, which is an important deformation mechanism during creep in-service conditions for Ni-based superalloys like the high-fraction γ' CM247 LC. This study is therefore an important milestone in the modelling of PBF-LB CM247 LC superalloy's mechanical behaviour, mostly in high-temperature conditions.

We confirm that neither the manuscript nor any parts of its content are currently under consideration or published in another journal. All authors have approved the manuscript and agree with its submission to *Intermetallics*.

Thank you for your consideration.

Sincerely,

Pietro Antonio Martelli

Ph.D. Student

Department of Applied Science and Technology (DISAT) - Polytechnic of Turin

Dear Reviewer,

First, thank you very much for the time spent in reading and reviewing this manuscript. I looked carefully at your suggestions and comments, and I hereby explain my answers:

1. The manuscript writing is good, but a careful revision must be done to improve clearance and avoid misunderstanding or incomplete ideas; check this sentence of the abstract I.e., "The effect of the testing parameters on the GBS activity in the PBF-LB material is discussed.".

The manuscript has been revised and a few changes have been made to the text to improve its clearance.

2. The introduction can be improved with a brief contextualization of the crack problem of aeronautical components and concerns about the use of LPBF in the industry and regulation before introducing CM247 LC.

A paragraph was included to address this topic. The added part is reported here as well.

In the realm of aeronautics, stringent regulations govern the permissibility of material defects, and these guidelines are rigorously enforced on a daily basis in accordance with certification protocols outlined by manufacturers or governmental entities like the FAA (Federal Aviation Administration). While these regulations seamlessly apply to conventionally sourced materials, they are currently in a developmental phase for materials produced through Additive Manufacturing (AM). When it comes to AM components, the norm involves either aligning with existing certification protocols or navigating through novel protocols specifically tailored for AM structural parts. The intricacy and seriousness of these protocols hinge on the criticality and failure mode of a component. These evolving protocols are designed to devise practical certification strategies that leverage the advantages of emerging AM technologies without compromising the safety of aircraft or space vehicle operations. They encompass a blend of indepth insights into material properties, manufacturing processes, and design considerations related to critical failure modes, potential defects, and part classification. Typically, these protocols offer guidance on optimal post-processing standards. For instance, recommendations may include the application of processes such as Hot Isostatic Pressing (HIP) to mitigate the criticality of internal defects impacting component functionality. Additionally, these protocols aim to enhance material microstructure and address anisotropy concerns. Remarkably, these protocols adhere to safe-life design methodologies that factor in the presence of internal crack-like defects. Through meticulous calculations of minimum critical flaw sizes, an allowable service life is defined, ensuring a comprehensive approach to structural integrity and safety in the realm of aeronautical engineering.

3. Include a brief review of directional properties obtained through variations in manufacturing strategies with LPBF or anisotropy.

A paragraph has been added in the introduction chapter: "Alloys processed by Additive Manufacturing typically show finer grains elongated along the building direction and crystallographic texture, which often depend on the process parameters and the scanning strategy [31–33]. These microstructures can significantly impact the final properties of the material, leading to mechanical anisotropy [34-35]."

4. If available, include the particle distribution of the powder used.

The D10, D50 and D90 values obtained through laser granulometry have been added in Section "Materials and methods".

5. Is there some procedure or automation to guarantee that the K-type thermocouple welded on the midsection of each tested sample did not affect the mechanical properties or had the same effect on all samples?

A paragraph has been added at the end of section 2.1: "The welding was performed according to standard procedures applied in Gleeble testing. It should be noted that the heat-affected zone in the subsurface area is negligibly small compared to the sample size. It has been well established that it does not affect the mechanical properties of samples tested in Gleeble, and this welding procedure has been employed in labs worldwide [37,38]."

6. There is data of the assay at 1093°C with a low strain rate of 10-4 s-1?

As explained in the text, the test at 1093°C and 10-4 s-1 was not performed based on the authors' experience. Indeed, the abovementioned test parameters may cause creep of the anvils: an example is reported in the image below. Moreover, the surface oxidation would have been so intense due to long test that no reliable quantitative results would be obtained from surface analysis.



I hope that the improvements and explanations may satisfy your suggestions and comments, and I hereby take the occasion to thank you again for your work.

With kind regards,

Dear Reviewer,

First, thank you very much for the time spent reading and reviewing this manuscript. I looked carefully at your suggestions and comments, and I hereby explain my answers:

1. In the introduction the following sentence needs to be reworded as its awkwardly worded. "However, Ni-based superalloys processed via additive manufacturing demonstrate very different microstructures compared to those observed in their counterparts processed using conventional techniques (such as, for example, investment casting) [30]."

The sentence "However, Ni-based superalloys processed via additive manufacturing demonstrate very different microstructures compared to those observed in their counterparts processed using conventional techniques (such as, for example, investment casting)" has been replaced with "However, the microstructures inherited from the Additive Manufacturing are somewhat different than those obtained with traditional technologies, such as investment casting".

2. In Figure 1, point to what the offer refers to as Flat Surface.

Fig.1 has been updated and the new image is attached to the re-submission as Fig1_updated.

3. For table 4, Ic is the mean distance between GBS, or what could be classified as nearest neighbor distance, not frequency.

The term "mean distance" has been updated in the Table caption as well as in the text lines.

4. Make it clear when stating that strain rate doesn't affect intensity that the author means that intensity is strain rate independent at only 649 C. It currently reads that the author is stating its strain rate independent for all temperatures.

The reference to the specific temperature has been added to the text.

5. Add to the discussion/conclusion at the cause of transition from no GBS at 649 C and significant GBS at 760 C.

The sentence "The triggering of the GBS at a certain temperature is given by the thermal nature of the atomic diffusion and the dislocation mobility" has been added in the conclusions.

I hope that the improvements and explanations may satisfy your suggestions and comments, and I hereby take the occasion to thank you again for your work.

With kind regards,

Highlights

- CM247 LC was processed by PBF-LB and heat treated
- Grain Boundary Sliding was studied at different temperatures and strain rates in a Gleeble 3800
- Grain Boundary Sliding activates at 760°C after hot compression at 10⁻³ and 10⁻⁴ s⁻¹
- Grain Boundary Sliding contribution increases with increasing temperature and decreasing strain rate
- The assessment of Grain Boundary Sliding allows forecasting the creep properties of PBF-LB CM247
 LC superalloy

The effect of temperature and strain rate on the grain boundary sliding in a CM247 LC Ni-based superalloy processed with Laser Based Powder Bed Fusion

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Abstract

The current work explores the meso-scale deformation behaviour of an additively manufactured CM247 LC at room and high temperatures. In particular, the study focuses on assessing grain boundary sliding (GBS), which can play a crucial role in the high-temperature deformation of superalloys. Specific samples were produced using the Laser Based Powder Bed Fusion technique (PBF-LB), heat treated and tested under monotonic compression in a Gleeble® system. Compression tests were carried out in a wide temperature range at two strain rates and the effect of testing parameters on GBS activity was studied. A thorough microstructural characterization of the PBF-LB material using EBSD and TEM revealed a γ/γ' microstructure consisting of columnar grains decorated with Hf-rich MC carbides without any segregations of alloying elements. Qualitative and quantitative analysis of GBS was performed using FEG-SEM and AFM, and contribution of GBS into plastic deformation was estimated. It was demonstrated that GBS is activated at 760°C. A direct correlation between the contribution of GBS into plastic deformation and testing temperature was found, while strain rate has the opposite effect. The highest GBS contribution (~32%) was recorded at 1093°C/10⁻³ s⁻¹. Finally, intergranular microcracking at triple junctions and along grain boundaries was observed when the material was tested at the highest temperatures (871°C and 1093°C). The effect of the temperature and the strain rate on the GBS activity in the PBF-LB material is discussed.

Keywords: Nickel-based superalloys; powder metallurgy; heat treatment; grain boundary sliding; grain boundary; plastic deformation mechanisms

1. Introduction

Ni-based superalloys are widely used in the aeronautical and energy production fields due to their

ability to preserve high mechanical strength at high temperatures and resistance to oxidation [1]. The exceptional mechanical properties are provided by both solid-solution [2] and precipitation strengthening mechanisms [3]. The achievement of these high-end performance is tightly bonded to the material microstructure and chemistry, in particular the quantity of Al and Ti are extremely relevant. At the same time, by increasing the amount of these elements, makes the material stronger and more stable at higher temperature, but weakens it formability and weldability, and introduce the risk of defects formation. Specifically, in the realm of aeronautics, stringent regulations govern the permissibility of material defects, and these guidelines are rigorously enforced on a daily basis in accordance with certification protocols outlined by manufacturers or governmental entities like the FAA (Federal Aviation Administration). While these regulations seamlessly apply to conventionally sourced materials, they are currently in a developmental phase for materials produced through Additive Manufacturing (AM). When it comes to AM components, the norm involves either aligning with existing certification protocols or searching through novel protocols specifically tailored for AM structural parts. The intricacy and seriousness of these protocols hinge on the criticality and failure mode of a component. These evolving protocols are designed to devise practical certification strategies that leverage the advantages of emerging AM technologies without compromising the safety of aircraft or space vehicle operations. They encompass a blend of in-depth insights into material properties, manufacturing processes, and design considerations related to critical failure modes, potential defects, and part classification. Typically, these protocols offer guidance on optimal postprocessing standards. For instance, recommendations may include the application of processes such as Hot Isostatic Pressing (HIP) to mitigate the criticality of internal defects impacting component functionality. Additionally, these protocols aim to enhance material microstructure and address anisotropy concerns. Remarkably, these protocols adhere to safe-life design methodologies that factor in the presence of internal crack-like defects. Through meticulous calculations of minimum critical flaw sizes, an allowable service life is defined, ensuring a comprehensive approach to structural integrity and safety in the realm of aeronautical engineering.

Among these materials, CM247 LC is characterised by a γ -FCC matrix reinforced by ordered L1₂ Ni₃(Al,Ti) γ ' precipitates [4,5]. Many studies have been conducted to assess the processability of this alloy using the Laser Based Powder Bed Fusion (PBF-LB) and its microstructural evolution during heat treatments due to the capability of additive manufacturing techniques to produce nearnet shape (NNS) complex parts [4–7]. Ni-based superalloys are typically used in high-temperature environment, and it is important to assess their deformation behaviour in such conditions. Particularly, loading these materials at high homologous temperatures ($\geq 0.5 T_m$) may trigger grain boundary sliding (GBS), when two grains slide with respect to each other along their boundary [8,9]. The grains then become incompatible with each other leading to failures with low ductility,

unless the incompatibilities created by sliding are accommodated. Two main accommodation mechanisms are considered in the current literature. First, grain boundary diffusion, where vacancies flow along the grain boundaries, and the individual grains become elongated. This process is known as Lifshitz GBS. It typically occurs in fine grained materials deformed at very high temperatures [8]. Second, accommodation by dislocation glide and climb, where grains retain their original shape. The latter is referred to as Rachinger sliding, and it occurs in materials with larger grains [8]. In general, crystalline slip accommodates most of the plastic deformation in metals. Nevertheless, there are numerous reports where GBS accommodates a significant proportion of the total strain depending on the testing parameters [8]. High temperatures promote GBS due to enhanced diffusion coefficient and accelerated dislocation glide and climb. As it is well known, the self-diffusivity of an atom is correlated to the temperature by the Arrhenius-like relation given in (1):

$$D = D_0 e^{-\frac{Q}{RT}} \tag{1}$$

where D_0 is a pre-exponential term, Q is the activation energy for self-diffusion, R is the gas constant and T is the temperature [10]. Similarly, dislocation glide and climb processes are thermal-dependent [11], and their velocity (v) is correlated to T as given in (2) [10]:

$$v \propto e^{-\frac{Q}{RT}} \tag{2}$$

At the same time, it is recognised that low strain rates also promote GBS activity [12]. Atom diffusion and dislocation movements are necessary to relax the high stress concentration levels arising at the grain boundaries during the sliding due to strain incompatibilities [13]. If the local stresses are too high to be accommodated and the alloy faces a drop in ductility, inter-granular cracking can take place leading to failure [14]. For this reason, it is very important to assess the activity of GBS and its contribution to the plastic deformation of polycrystalline alloys which can be subjected to creep condition [15–17]. Simple monotonic uniaxial tests in a wide temperature and strain rate ranges have been widely used in literature for this purpose. Special attention was paid to this phenomenon occurring in ultrafine-grained Al alloys during their plastic deformation at relatively low temperatures [18]. Also, its key role in superplastic deformation of fine-grained Mg alloys was demonstrated [19]. There is a significant body of experimental research focusing on GBS and its quantification in light alloys [20–25]. On the contrary, very few experimental studies have been done to investigate the occurrence of GBS in Ni-based superalloys and its dependence on the microstructure and testing parameters. Torres et al. [26] focused on the GBS in-situ measurements on a Ni-based superalloy tested at high temperature. Takizawa et al. [27] analytically assessed the GBS as the controlling deformation mechanism in superplastic highpressure sliding formability of Inconel 718 at high temperature. Others focus on molecular dynamics simulations rather than experimental studies [28,29].

Additive manufacturing is becoming increasingly important for producing complex shape parts for aeroengines [30].

However, the microstructures inherited from the Additive Manufacturing are somewhat different than those obtained with traditional technologies, such as investment casting [30]. Alloys processed by Additive Manufacturing typically show finer grains elongated along the building direction and crystallographic texture, which often depend on the process parameters and the scanning strategy [31–33]. These microstructures can significantly impact the final properties of the material, leading to mechanical anisotropy [34,35].

While there are several studies reporting the mechanical properties of additively manufactured Ni-based superalloy [36], their mesoscale deformation behaviour has not been investigated up to date. Understanding the mechanisms operating at high temperatures is crucial for microstructural design in these materials to achieve high performance at service temperatures. Particularly, GBS is a very important deformation mechanism controlling their high-temperature creep resistance. No works exploring the GBS activity in additively manufactured Ni-based superalloys can be found in the current literature. The main objective of the present work was to explore the GBS activity at different temperatures and strain rates in CM247 LC Ni-based superalloy processed by PBF-LB. Monotonic compression tests were performed to assess the influence of test temperature and strain rate on the deformation behaviour of the superalloy. The deformed samples were then analysed using Field-Emission Gun Scanning Electron Microscopy (FEG-SEM) and Atomic Force Microscopy (AFM) and a scheme was developed to analyse the contribution of GBS as a function of the testing parameters.

2. Materials and Method

2.1. Material and samples production

The specimens were produced using a PrintSharp 250 PBF-LB system by Prima Additive equipped with a carbon steel building platform heated at 80°C and a rubber recoater. Printing was performed under high purity Ar gas with an oxygen level lower than 0.1%. The parameters' ranges studied during the process optimization are reported in Tab. 1. Pre-alloyed gas atomized powder supplied by Praxair Inc. as NI-1230 was used as raw material. The chemical composition according to the manufacturer datasheet is reported in Tab. 2, while the volumetric powder size distribution measured through laser granulometry resulted in D10, D50 and D90 of 24.5 μ m, 33.7 μ m and 46.2 μ m, respectively. Finally, apparent density and flowability were measured according to ASTM B212 and ASTM B213 and resulted in 4.63 \pm 0.01 g cm⁻³ and 12.9 \pm 0.2 g cm⁻³, respectively.

Tab. 1. Parameters range

Laser power	Scan speed	Hatching distance	Stripes-head
[W]	[mm s ⁻¹]	[mm]	overlapping
170 - 195	1000-1900	0.03 - 0.08	0.1 - 0.2

Tab. 2. Chemical composition (wt.%) of CM247LC alloy according to producer's technical datasheet

Ni	Cr	Co	W	Ta	Mo	Al	Ti	Hf	Zr
[%]	[%]	[%]	[%]	[%]	[%]	[%]	[%]	[%]	[%]
Bal.	8.0	9.3	9.7	3.6	0.5	5.2	0.8	1.7	0.01

The powders were subjected to further chemical analysis to assess the average content of the main interstitial elements: C and S were determined via NIR oxygen combustion using a Leco CS 744. At the same time, O, N and H were evaluated via inert gas fusion in a Leco ONH 836 machine. Results are shown in Tab. 3.

Tab. 3. Residual elements (wt.%) in the raw powder

С	S	0	N	Н
[%]	[ppm]	[%]	[%]	[ppm]
0.083	4.79	0.014	0.0036	7.89
(± 0.001)	(± 0.08)	(± 0.002)	(± 0.0002)	(± 1.38)

Ten specimens for compression tests were produced in a single job as cylinders having a height of 17 mm (aligned along the building direction) and a diameter of 10 mm (Fig. 1). A flat surface of 3.5 mm of width was also added to the model to measure the GBS after compression tests through SEM and AFM. From now on this surface will be referred to as "flat surface". The wires of a K-type thermocouple were welded on the midsection of each tested sample to control the temperature with the accuracy of $\pm 1^{\circ}$ C (Fig. 1). The welding was performed according to standard procedures applied in Gleeble testing. It should be noted that the heat-affected zone in the subsurface area is negligibly small compared to the sample size. It has been well established that it does not affect the mechanical properties of samples tested in Gleeble, and this welding procedure has been employed in laboratories worldwide [37,38].

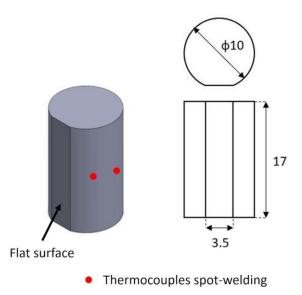


Fig. 1. Raw specimen produced for the compression tests (dimensions in mm)

2.2. Sample post processing and heat treatment

The manufactured samples were cut from the building platform via Wire Electro-Discharge Machining (WEDM), manually sandblasted and stress relieved at 1080° C for 2 hours in an air furnace. Then, the samples underwent Hot Isostatic Pressing (HIP) in a Quintus QIH 15L machine equipped with a Molybdenum furnace to reduce the internal flaws population. This step was performed below the γ ' solvus temperature. Pressure and temperature were increased simultaneously, and the samples were cooled to room temperature at a rate of 15° C min⁻¹ (calculated from HIP temperature to 400° C) after soaking for 3 hours. Subsequent heat treatment was optimized elsewhere by the authors [4]. The recipe is reported briefly:

- Solution annealing (SA): 1245°C for 2 hours + fast cooling at 100°C min⁻¹;
- I aging: 1080°C for 4 hours + fast cooling at 50°C min⁻¹;
- II aging: 870°C for 6 hours to stabilize the microstructure and promote a further precipitation of fine γ'.

The whole cycle was applied in a low-pressure furnace TAV minijet. Even if other literature works use 1260°C as a SA temperature, authors pursued the lower temperature to avoid the formation of thermally induced porosities (TIPs) [4]. Furthermore, performing the SA at a lower temperature could be of industrial interest since it reduces the related costs and increases the available number of certified equipment and subcontractors in the market.

2.3. Machining and polishing

All specimens were machined to a final length of 15 mm to ensure that the two circular faces were parallel. Furthermore, the flat surface was ground with SiC sandpapers up to 1200 grid and polished with 6-3-1 μ m diamond suspensions. Finally, 0.04 μ m colloidal silica suspension was used to obtain the mirror-like surface. Ethanol and ultrasonic baths were used to remove any

debris resulting from the polishing steps.

2.4. Compression test using the Gleeble® system

Compression tests were performed using a Gleeble® 3800 system under constant strain rates of 10^{-3} s⁻¹ and 10^{-4} s⁻¹ at room temperature (RT), 649° C, 760° C and 871° C, while only the strain rate of 10^{-3} s⁻¹ was used during the exposure at 1093° C due to creep of anvils detected during testing at low strain rate of 10^{-4} s⁻¹. Prior to the tests, the height of each sample was measured to calculate the final deformation. Based on the authors' previous experience [18], a total plastic strain of ~8% was set as a target for the GBS to be easily detectable. This value ensured that GBS events could be observed leaving the surface flat enough for the following analysis using AFM. Samples were heated via Joule effect with a constant heating rate of 10° C s⁻¹, and temperature was controlled using a K-type thermocouple as mentioned above (Fig. 1).

2.5. Microstructural assessment

The microstructure of the as-treated alloy was thoroughly studied before compression testing. The Electron Backscatter Diffraction (EBSD) analysis was performed using an Oxford Symmetry detector mounted on a focused ion beam-scanning electron FIB-SEM TESCAN S9000G by Tescan. The specimens' tilting was kept at 70°, with a voltage of 20 kV and rolling direction parallel to the building axis of the sample. The microstructure at grain boundaries was evaluated using a Zeiss EVO15 SEM and a FEG S/TEM (Talos F200X, FEI) operated at an accelerating voltage of 200kV. Elemental mapping was carried out using energy dispersive spectroscopy (EDS) SuperX detector from FEI & Bruker. The foils for TEM observations were cut from a heat-treated sample and prepared with a TenuPol 5 (Struers®) by twin-jet electro-polishing with a solution consisting of 10% perchloric acid (HClO₄), 30% butanol and 60% ethanol at -25°C under an operating voltage of 20 V.

2.6. Grain boundary sliding quantification

After being tested, the specimens were subjected to further inspections to assess the deformation mechanisms and evaluate the activity of GBS. For practical reasons, the authors chose the out-of-plane measurement among the possible quantification methods, which is shown in Fig. 2.

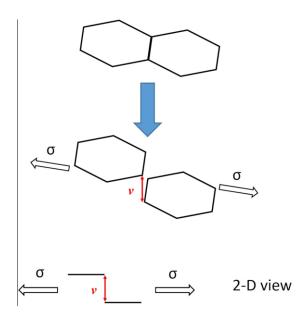


Fig. 2. GBS representation and out-of-plane offset component

The corresponding equation used for the evaluations was reported in the work of Langdon [9] for polished surfaces (3):

$$\varepsilon_{gbs} = 1.1 \cdot \frac{v_r}{l_r} \tag{3}$$

where ε_{gbs} is the plastic deformation related to the grain boundary sliding, \underline{v}_r is the displacement between grains perpendicular to the polished surface and \underline{l}_r the mean distance between sliding events randomly measured. Using equation (4), it is possible to calculate the GBS relative contribution (ξ) to the total plastic deformation ($\varepsilon_{p,tot}$) as a value ranging from 0 to 1 [9]:

$$\xi = \frac{\varepsilon_{gbs}}{\varepsilon_{p,tot}} \tag{4}$$

There are no standard measurement rules for the GBS quantification, so a scheme was defined by the authors. The surface relief of the deformed specimens was analysed using a FEG-SEM APREO 2S LoVac to assess the activity of GBS during plastic deformation. The distance between sliding events was recorded by drawing six lines (three parallels and three perpendiculars to the load direction) on five SEM images randomly taken at a magnification of 500x, as represented by the red lines in Fig. 3. For each line, the number of GBS events was counted and divided by the length of the line. The mean value of the six different counts was used to quantify \underline{l}_T . Once the SEM observation was completed, the specimens were analysed with a Park XE150 Atomic Force Microscope (AFM) to measure nine out-of-plane offsets; their mean value was used to quantify \underline{v}_T . The area subjected to both FEG-SEM and AFM analyses was in the midsection of the specimen's flat surface (Fig. 3).

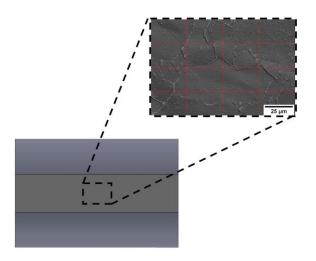


Fig. 3. Investigated area after compression tests with detail on the lines drawn for l_r quantification.

3. Results and discussion

3.1. Microstructural assessment

Fig. 4a shows the microstructure of the as-treated material, while Fig. 4b presents an EBSD map of the same sample. Annealing the samples at 1245° C led only to a partial dissolution of γ ' and grains structure recovery from PBF-LB process. Fig. 4 confirms that annealing the PBF-LB CM247 LC at 1245° C may not be effective in achieving complete recovery of the PBF-LB structure and γ ' dissolution primarily due to uncertainties associated with the industrial furnaces. Fig. 4a shows the coexistence of small and cubic γ ' precipitates (marked by the green arrow) as well as large and irregular ones (indicated by the red arrow), which were retained from the HIP process [4,39]. The larger γ ' precipitates are primarily found at the grain boundaries, Also, SEM images reveal that the grain boundaries appear to be decorated with Hf-rich MC carbides (indicated by the blue arrow) [4,40].

Fig. 4b displays relatively large grains ($\sim 60~\mu m$) that are predominantly elongated along the building direction. The aspect ratio (AR) of the grains, determined as the ratio between the major and the minor axes of the fitting ellipse obtained from EBSD characterisation, is reported as 3.5 ± 2.1 . The grain morphology is noteworthy because some previous publications reported that elongated have a reduced propensity for sliding compared to the equiaxed ones [41].

The majority of grain boundaries observed in Fig. 4 are non-serrated. This can be attributed to the relatively rapid cooling rate (100°C/min) employed after SA heat treatment (see Section 2.2) which results in non-serrated grain boundaries in Ni-based superalloys [39]. Furthermore, the SA treatment performed at 1245°C left a significant fraction of Low-Angle Grain Boundaries (LAGB <10°), estimated to be approximately 32%.

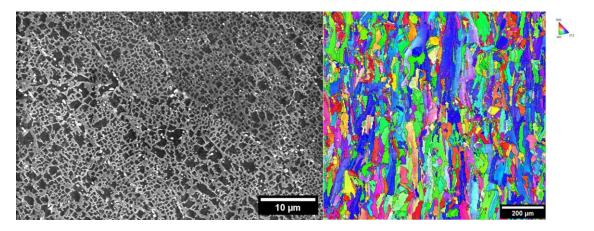


Fig. 4. Microstructure after heat treatment: a) general microstructure observed with SEM and b) EBSD IPF map.

FEG S/TEM analysis was conducted to assess the potential presence of segregations at grain boundaries, which could influence the GBS process in polycrystalline alloys [42,43]. Fig. 5a shows an overall view of a grain boundary (indicated by the white arrow) separating two regions rich in the γ matrix. It is noteworthy that no chemical segregations were observed, indicating that the SA treatment effectively eliminated any chemical inhomogeneity present in the matrix resulting from the PBF-LB process. Additionally, Fig. 5a displays the previously mentioned Hfrich MC carbides (indicated by light-blue arrow).

Primary γ' precipitates are usually intergranular and only form when the superalloy is heat treated below the γ -solvus temperature. Annealing above the γ -solvus dissolves the primary γ' and subsequent solution cooling, at rates of $\lesssim 10^{\circ}$ C/min results in the formation of intragranular secondary γ' precipitates during the early stage of cooling. Further cooling results in the growth of secondary γ' precipitates until the low elemental diffusivities of the γ stabilizing elements make it difficult for these elements to reach the comparatively coarse secondary γ' . This leads to supersaturation of these elements within the γ matrix and consequently drives the nucleation of additional nanosized intragranular tertiary γ' [44,45].

However, they are not precisely located on the boundary itself but are several hundreds of nanometres away from it. Very thin (< 100 nm) tertiary γ ' precipitates can be observed within the inter-particles γ -channels (indicated by the black arrow). In Fig. 5b, an EDS map was performed at a very high magnification to highlight a grain boundary that separates two γ ' particles. Even in this case, the grain boundary exhibits a chemical composition similar to that of the γ phase. Consequently, even when two γ ' particles are divided by a grain boundary, dislocations can find a preferential path along which they can slide, partially bypassing the resistance imposed by the γ - γ ' interface.

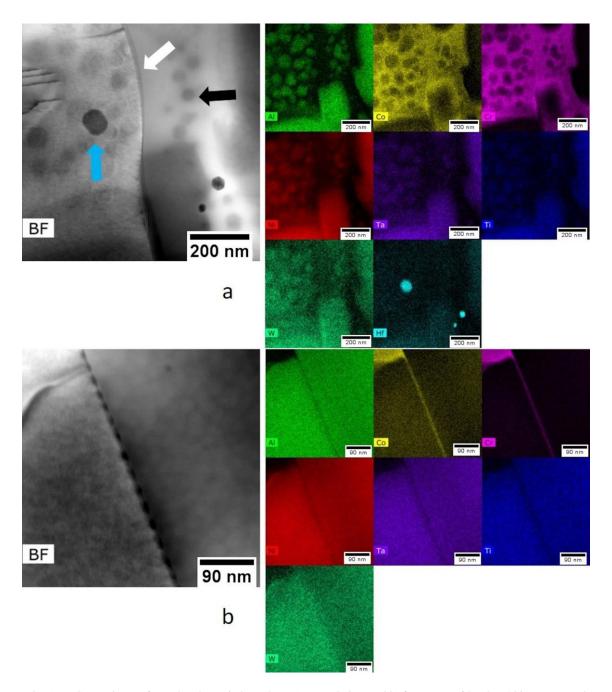


Fig. 5. EDS mapping performed at the grain boundary: a) general view and b) focus on γ - γ' border. Al is represented by light green, Co by yellow, Cr by light purple, Ni by red, Ta by dark purple, Ti by dark blue, W by dark green and Hf by light blue.

3.2. SEM assessment of samples tested at different temperatures and strain rates

In order to evaluate the influence of temperature and strain rate on GBS activity, the surface relief of the tested specimens was examined using FEG-SEM. To illustrate the qualitative evolution of surface morphology, a comprehensive synoptic table is presented in Fig. 6. It provides an overview of the observed changes in surface topography under different temperature and strain rate conditions, allowing for a qualitative assessment of the GBS behaviour.

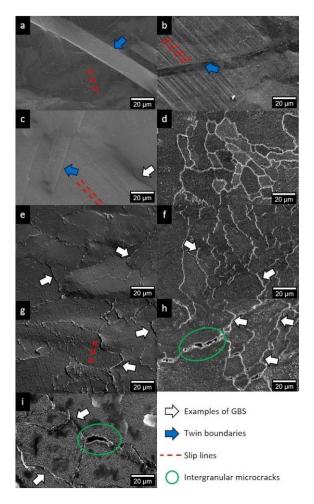


Fig. 6. Morphological characterization of samples' surfaces. At left, samples tested with a strain rate of 10⁻³ s⁻¹ at a) RT, c) 649°C, e) 760°C, g) 871°C and i) 1093°C. On the right, samples tested with a strain rate of 10⁻⁴ s⁻¹ at b) RT, d) 649°C, f) 760°C and h) 871°C.

Fig. 6a and Fig. 6b illustrate the surface relief after compression testing at RT with different strain rates. No GBS is observed at RT for both applied strain rates, but intense slip activity is detected (marked by red lines). Twin boundaries are also clearly visible on the surface (marked by blue arrows). Fig. 6c and Fig. 6d present SEM images of the surface relief after testing at 649°C with strain rates of 10^{-3} and 10^{-4} s⁻¹, respectively.

The sample tested at 649°C/10⁻³ s⁻¹ exhibits well-pronounced slip lines (marked by red lines), while only sporadic sliding events are recorded, as marked by the white arrow in Fig. 6c. In the latter specimen, straight highly deformed boundaries can be observed, which are identified as twins (marked by blue arrows). No significant GBS is noticeable after testing at 649°C/10⁻⁴ s⁻¹ (Fig. 6d). These observations confirm that GBS is practically inactive up to 649°C.

Fig. 6e and Fig. 6f show the surface relief of specimens tested at 760°C. GBS is now activated throughout the surface (indicated by white arrows). Fig. 6g and Fig. 6h illustrate the surfaces after testing at 871°C with strain rates of 10⁻³ and 10⁻⁴ s⁻¹, respectively. GBS events (marked by white arrows) are observed after the test performed at 871°C/10⁻³ s⁻¹ (Fig. 6g), as expected. Slip lines

(marked by dashed red lines) are still recognizable within some grains despite a slight surface oxidation. The sample subjected to the 871°C/10⁻⁴ s⁻¹ test exhibits more pronounced GBS activity (Fig. 6h). Additionally, some microcracks are formed along grain boundaries (marked by green circles) and at triple junctions. Finally, Fig. 6i shows the surface relief after the test conducted at 1093°C/10⁻³ s⁻¹. Intense GBS (marked by white arrows) and intergranular microcracking (indicated by green circles) can be observed.

3.3. Qualitative AFM analysis of surface relief

The outcomes of AFM analysis are consistent with the observations made using SEM (Section 3.2). The AFM 3D map of the deformed surface clearly indicates that the primary mechanism of plastic deformation at RT is dislocation glide (Fig. 7). Dislocations were observed to traverse twin boundaries, as evidenced by the AFM scan. The mechanism of dislocation transmission through twin boundaries in Ni-base superalloys has been previously proposed by different authors [46,47]. The gliding dislocations slip through a twin boundary when the slip plane with the highest Taylor factor intersects the twin boundary. Furthermore, recent studies [48] have documented strain localization at twin boundaries, which is believed to result in the appearance of twins on surfaces of specimens deformed at RT. The absence of GBS events makes this mechanism particularly visible.

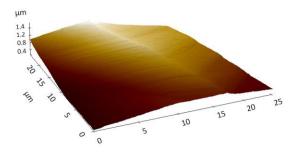


Fig. 7. Transmission of slip lines through twin boundaries under AFM observed in the sample RT/10⁻³ s⁻¹.

Sporadic GBS events were observed on the surface of specimens tested at 649°C. Fig. 8a presents an example of such an event, where two distinct grain boundaries underwent sliding during deformation with a strain rate of 10^{-3} s⁻¹ (marked by red circle). However, these events were confined to a limited spatial extent, spanning only a few microns. Additionally, under AFM, only rare occurrences similar to the one reported have been recorded for both the specimens tested at 649°C. In the case of the specimen tested at 649°C/ 10^{-4} s⁻¹, the presence of γ particles being cut by dislocation glide was evident (Fig. 8b). This behaviour aligns with the observations made by Stinville et al. [46].

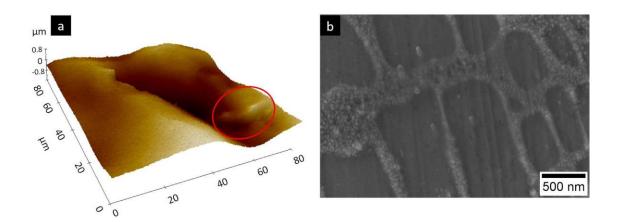


Fig. 8. a) Surface 3D reconstruction under AFM for the specimen tested at 649°C with strain rate of 10^{-3} s⁻¹ (red circle indicates GBS) and b) detail on the γ ' particles cut by dislocation activity

Fig. 9 illustrates a surface reconstruction obtained through AFM for the specimens tested at 760°C. In both cases, the occurrence of GBS (red circle in Fig. 9a) involves a greater number of grains compared to the specimens tested at 649°C. However, the intensity of these events remains relatively low and comparable to what was observed at 649°C. Fig. 9a and Fig. 9b indicate that the intensity of the GBS is consistent across different grains and is evenly distributed throughout the examined area. Nonetheless, it is important to note that GBS does not affect all grains, as some of them exhibit no evidence of sliding.

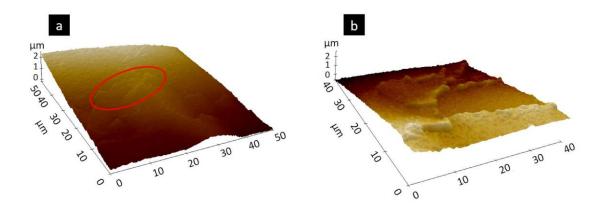


Fig. 9. Surface 3D reconstruction under AFM for the specimens tested at 760 °C with strain rate of a) 10^{-3} s⁻¹ and b) 10^{-4} s⁻¹

Surface reconstructions for specimens tested at 871°C with strain rates of 10⁻³ s⁻¹ and 10⁻⁴ s⁻¹ are presented in Fig. 10a and Fig. 10b, respectively. In these cases, the occurrence of GBS is more pronounced, and its intensity is significantly enhanced by the lower strain rate. To emphasize the intensity of GBS activity, a line scan is shown in Fig. 10c. In this plot, the Y-axis represents the relative position of the AFM tip with its 0-position set at the beginning of the analysis. The X-axis corresponds to the length of the line scan.

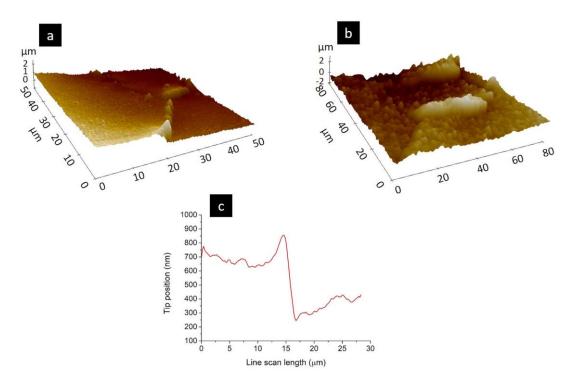


Fig. 10. Surface 3D reconstruction under AFM for the specimens tested at 871° C with strain rate of a) 10^{-3} s⁻¹ and b) 10^{-4} s⁻¹ and c) example of a line scan performed across a GBS event

Finally, Fig. 11 depicts the surface reconstruction obtained through AFM following the test conducted at 1093°C with a strain rate of 10⁻³ s⁻¹. The figure reveals the formation of prominent large steps, which are a consequence of intense sliding events occurring at this high temperature. The intergranular microcracking is considerably more pronounced at this high temperature. Additionally, Fig. 11b provides an example of a line scan that highlights the activity of GBS.

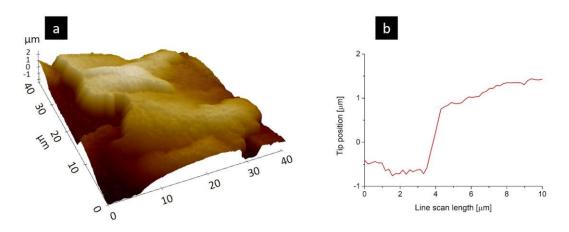


Fig. 11. Surface 3D reconstruction under AFM for the specimens tested at 1093° C with strain rate of 10^{-3} s⁻¹ and b) example of a line scan performed across a GBS event.

Based on the previous findings, it is indicated that once the GBS is activated, it triggers the reciprocal sliding of adjacent grains according to the mechanism illustrated in Fig. 12. The displacement resulting from this sliding can be described by the sliding vector \vec{S} , which is made

of the three components v, u and w.

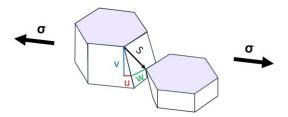


Fig. 12. Sliding mechanism of neighbouring grains

In the current literature it has been well established that the relative displacement of the grains results in strain incompatibilities at the grain boundaries, which lead to the stress concentration at the multiple junction points due to dislocations pile ups (as indicated by the white arrow in Fig. 13a) [13]. At very high testing temperatures, the stress concentration at these junctions accompanied by insufficient accommodation leads to the nucleation of voids (or cavities). These voids act as stress concentrators, promoting the formation and propagation of intergranular microcracks (as shown in Fig. 13b). The stress/strain ahead of the crack leads to further cavity nucleation and growth by interlinking with new cavities formed ahead of crack tip, as one can see on Fig. 6h [49]. The formation of cavities at the multiple junctions can be visually represented as illustrated in the schematic of Fig. 13c, where one of these cavities is formed due to the reciprocal displacement of the neighbouring grains. It should be noted that no particle free zones (PFZ) are observed in the vicinity of grain boundaries of all tested samples, thus pointing at dislocation glide as the main accommodation mechanism of the GBS process independently on the testing parameters [8]. Therefore, it can be concluded that the Rachinger sliding takes place during plastic deformation of the studied material.

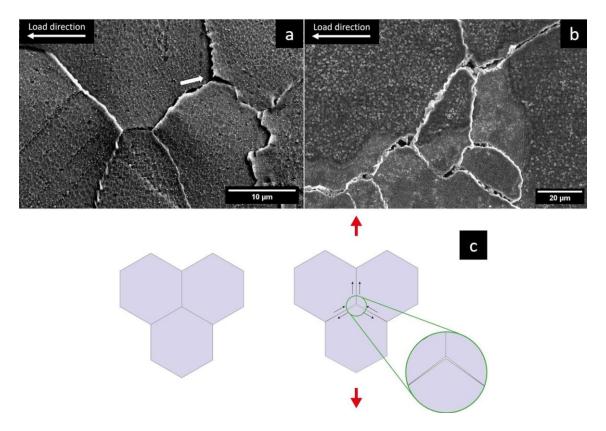


Fig. 13. Effect of the strain incompatibilities at the multiple junction points: a) concentration of GBS at the lower temperatures (760°C/10⁻³ s⁻¹), b) inter-granular microcracks formation at the higher temperatures (871°C/10⁻³ s⁻¹) and c) visual description of cavities formation

3.4. Grain boundary sliding quantification

After examining the morphological characteristics of the deformed specimens, the quantitative analysis of GBS contribution (ξ) to the overall plastic deformation of PBF-LB CM247 LC alloy was conducted. The measurement process involved determining the out-of-plane offsets (v_r) and mean distances between GBS events (l_r). The AFM 3D maps and images captured using FEG-SEM were utilized for this purpose. The measured values of these parameters for various test conditions are presented in Tab. 4.

Tab. 4. Out-of-plane offsets intensity (v_r) and mean distance (l_r) calculated for all the samples.

Temperatur	Strain rate	$\mathbf{v_r}$	$l_{\rm r}$	
e	[s ⁻¹]	[nm]	[µm]	
[°C]	[s]			
649	10-3	251±156	1000*	
760	10-3	313±127	70±22	
871	10^{-3}	552±225	46±7	

1093	10^{-3}	1459±697	57±10
649	10-4	257±56	1000*
760	10-4	597±93	77±10
871	10 ⁻⁴	1976±942	106±33

^{*}values approximated due to the very rare appearance of sliding events

The out-of-plane offsets remain constant at 649°C at both strain rates (10⁻³ s⁻¹ and 10⁻⁴ s⁻¹), indicating that the intensity of sliding events is not influenced by the strain rate at this temperature. Due to the rarity of GBS at 649°C, an actual distance between sliding events could not be determined. To account for this, the authors chose to set a value of 1 mm, which is a realistic approximation based on surface observations. At 760°C/10⁻³ s⁻¹, there is a slight increase in the out-of-plane offsets compared to the specimens tested at 649°C. In this case, the mean distance of GBS events was measurable and found to be 70 µm. The main difference between the tests performed at 760°C at 10⁻³ s⁻¹ and at 10⁻⁴ s⁻¹ is the sliding intensity. The average out-of-plane offsets are 313 nm and 597 nm, respectively. However, the distance between these events does not show significant variation. Despite undergoing plastic deformation for a longer duration, there is no notable increase in the number of activated grain boundaries. The concept of "activated" boundaries was discussed in [50]. This observation in the studied materials could be attributed to the morphology of grain boundaries. Experimental evidence suggests that serrated grain boundaries in Ni-based superalloys are effectively locked for sliding during plastic deformation, while non-serrated grain boundaries undergo sliding [51]. From the SEM images taken at higher magnifications (Fig. 13a, b), it is evident that the grain boundaries showing sliding are nonserrated. As the test temperature increases to 871°C at a strain rate of 10⁻³ s⁻¹, both the intensity and the mean distance of the sliding events increase, indicating that more grain boundaries are able to slide. The GBS intensity at $871^{\circ}\text{C}/10^{-4}\text{s}^{-1}$ shows a significant increase ($v_t > 1000 \text{nm}$), while the spacing between events tends to be higher indicating a lower frequency. This behaviour may be attributed to inter-granular microcracking, which could result in local stress relaxation. A similar trend can be observed at 1093°C/10⁻³ s⁻¹ in terms of intensity, although the mean distance of the sliding events is lower.

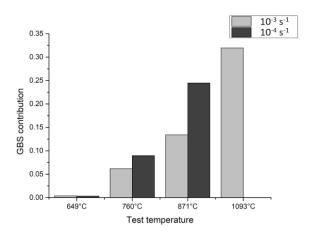


Fig. 14. GBS contribution (ξ) as a function of test temperature and strain rate

Finally, the value of ξ representing the GBS contribution into plastic deformation, was calculated using equations (3) and (4) and plotted in Fig. 14 as a function of temperature and strain rate. The calculated \xi values are normalized, ranging between 0 and 1, where 0 represents the absence of GBS and 1 represents the exclusive occurrence of GBS without any other plastic deformation, which is not realistic. At 649°C, the estimated GBS contribution is less than 1% based on the given reference value of 1 mm for the mean distance between GBS events. This indicates that the plastic deformation is provided predominantly by dislocation gliding, whereas contribution of GBS is negligibly small, and the contribution of diffusion is also very low due to low homologue temperature. For the 1093°C test, some steps were higher than the measurement range in AFM (±4 μm), suggesting that the actual GBS contribution would be higher. It can be concluded that the GBS contribution at 1093°C is at least 32% of the total plastic deformation. In the specimen tested at 760°C with a strain rate of 10⁻³ s⁻¹, the GBS contribution was calculated to be 0.062 (6.2%), and at a strain rate of 10⁻⁴ s⁻¹ it was 0.089 (8.9%). The GBS contribution increases further to 0.134 (13.4%) and 0.245 (24.5%) at 871°C. It is evident that ξ increases with the rise in testing temperature and the decrease in strain rate. This behaviour is consistent with the influence that these test parameters have on both atomic diffusion and dislocation mobility, affecting the occurrence and intensity of GBS.

4. Conclusions

Monotonic compression tests in a wide temperature range and two different strain rates were performed on CM247 LC samples processed with PBF-LB. SEM and AFM analyses of the surface relief of tested samples were performed, and the GBS activity was analysed as a function of temperature and strain rate. Contribution of GBS into total plastic deformation was estimated. The following conclusions can be drawn:

• However, the microstructures inherited from the Additive Manufacturing are somewhat

different than those obtained with traditional technologies, such as investment casting [30]. Alloys processed by Additive Manufacturing typically show finer grains elongated along the building direction and crystallographic texture, which often depend on the process parameters and the scanning strategy [31–33]. These microstructures can significantly impact the final properties of the material, leading to mechanical anisotropy [34,35].

- The contribution of GBS (ξ) to the total plastic deformation was estimated by measuring the out-of-plane offsets using AFM. It was found that ξ value increases with increasing temperature (T) and decreasing strain rate (ἐ). This observation is consistent with the enhancing effects of T and ἐ on grain boundary diffusional processes and dislocation mobility.
- Despite the elongated and non-equiaxed grain structure (aspect ratio AR=3.5±2.1), GBS significantly contributes to the plastic deformation at temperatures of 760°C and above.
- Microstructural observations revealed the absence of chemical segregations at grain boundaries which could promote GBS activity. Additionally, large matrix γ channels between precipitated γ' particles were observed, potentially facilitating diffusion processes and dislocation mobility involved in GBS. Absence of PFZ in the vicinity of grain boundaries of the tested samples indicates that the Rachinger sliding takes place and GBS process is accommodated mainly by dislocation glide at all studied temperatures and strain rates.
- Intergranular microcracking starting at triple junctions was observed at 871°C/10⁻⁴ s⁻¹ and 1093°C/10⁻³ s⁻¹, indicating the intense stress concentration caused by GBS at triple junctions.

Given the intense intergranular microcracking observed at 871°C and 1093°C, it is recommended to avoid or minimize the occurrence of GBS in CM247 LC alloy under in-service creep conditions in order to enhance its mechanical properties at high temperatures. To achieve this, future studies will focus on raising the SA temperature treatment. By increasing the SA temperature, complete γ ' solution and recovery of the PBF-LB structure can be achieved. This approach aims to modify the material's microstructure and eliminate or reduce the presence of grain boundaries that are susceptible to GBS, thereby improving the alloy's resistance to intergranular microcracking and enhancing its overall mechanical performance at elevated temperatures.

Authorship contribution statement

Pietro Antonio Martelli: conceptualization, writing, editing, samples production, experimental investigation and data analysis. **Ilchat Sabirov**: supervision, conceptualization, experimental investigation, data analysis and editing. **Miguel Alberto Monclus**: experimental investigation, data analysis. **Emilio Bassini**: heat treatments, conceptualization and editing. **Giulio Marchese**: experimental investigation. **Daniele Ugues**: supervision and conceptualization.

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Data Availability

All the experimental data to reproduce these findings are already contained in the paper and further material can be disclosed contacting the corresponding author.

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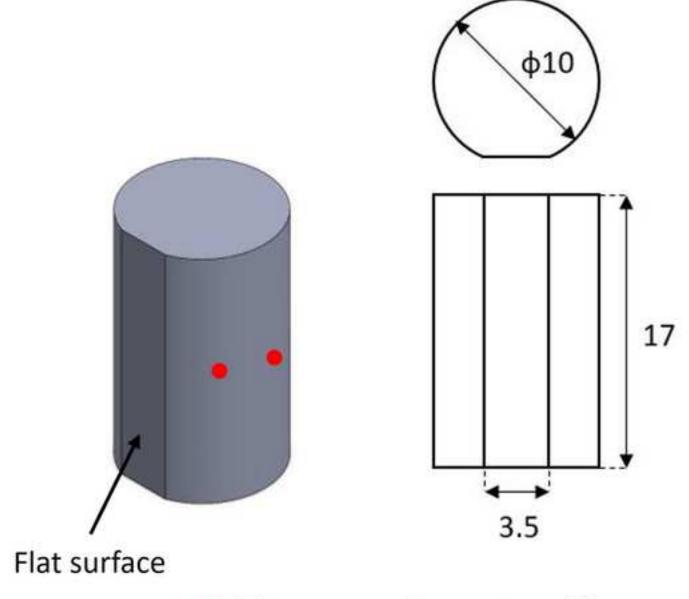
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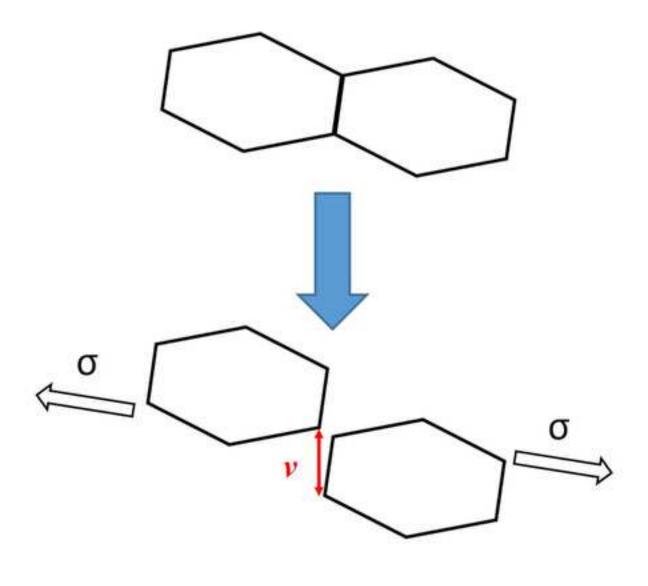
Declaration of Interest Statement

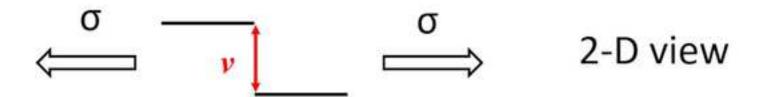
Declaration of interests

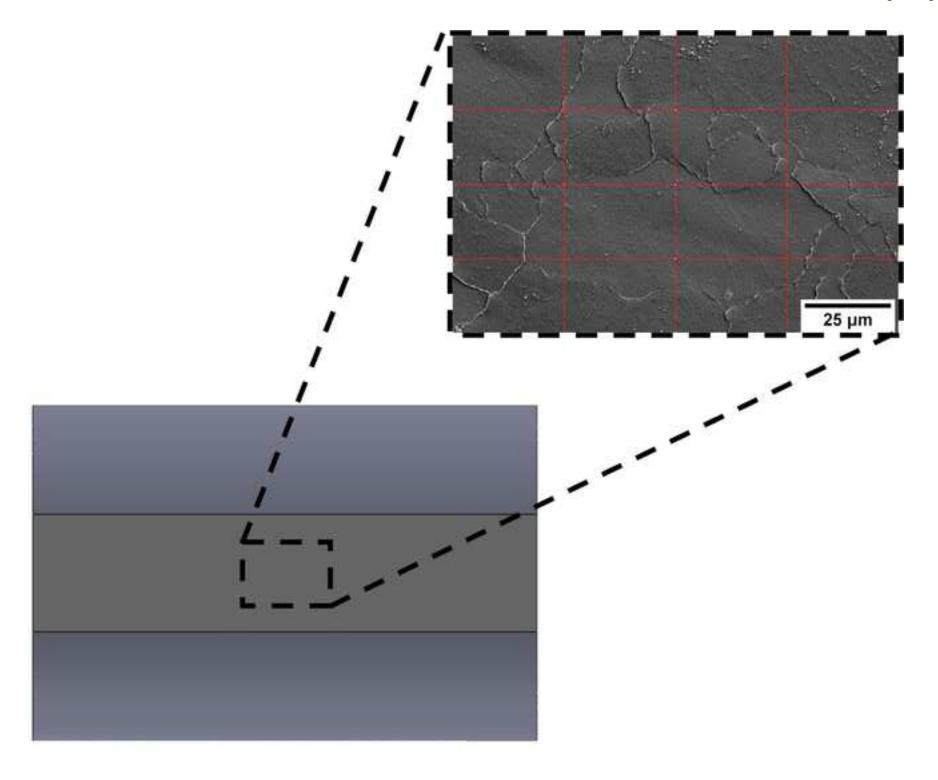
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☐The authors declare the following financial interests/personal relationships which may be considered as potential competing interests:

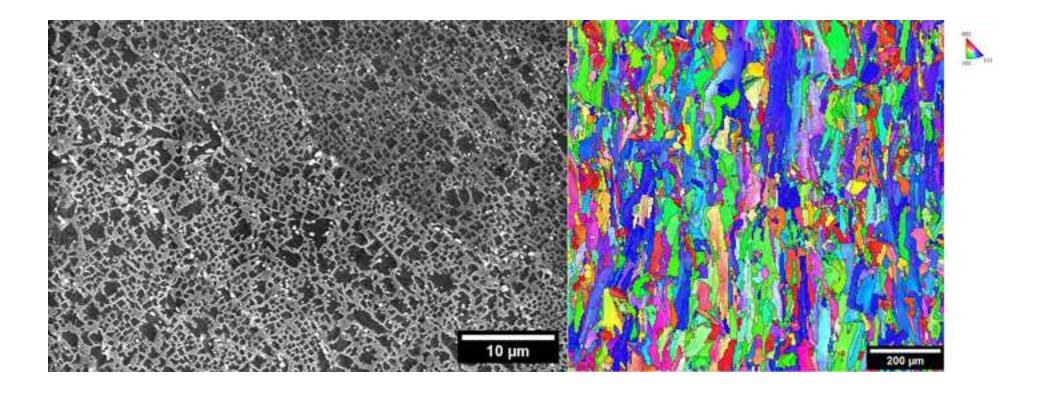


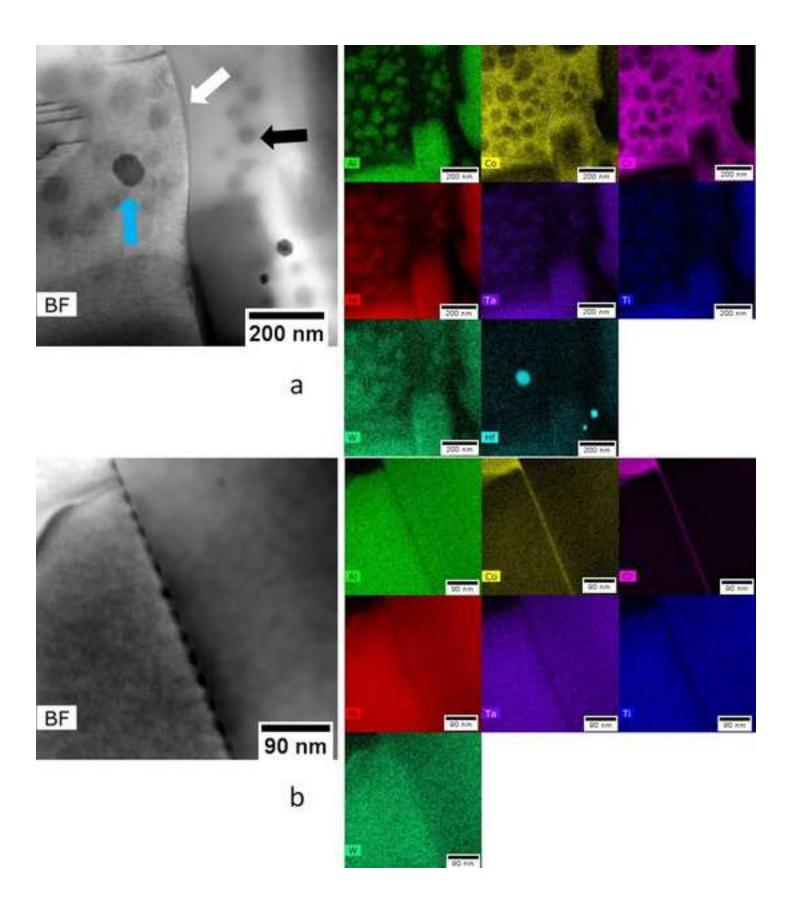
Thermocouples spot-welding

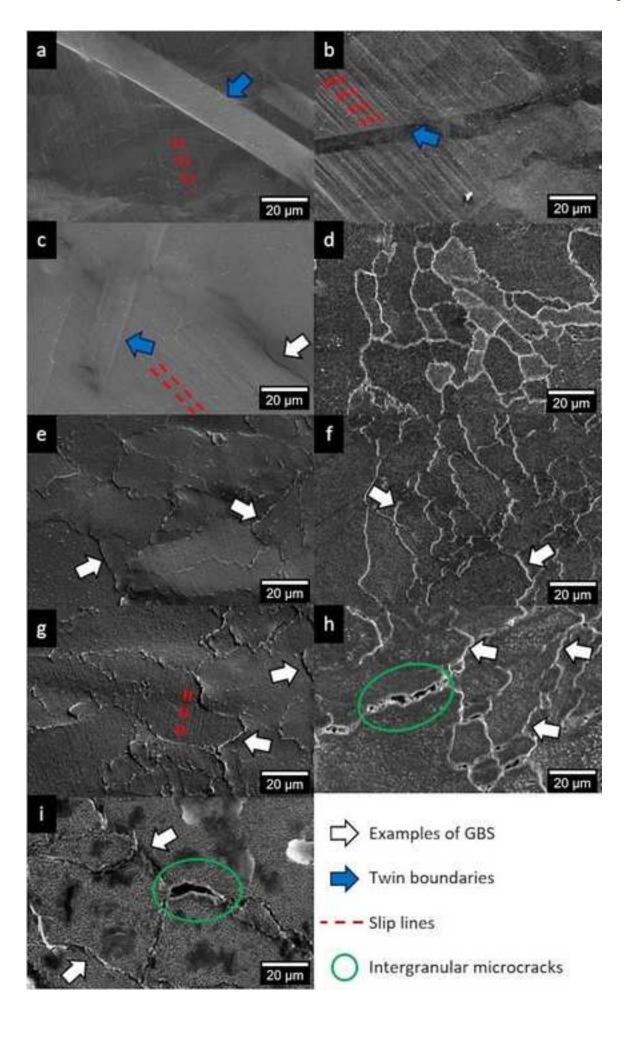


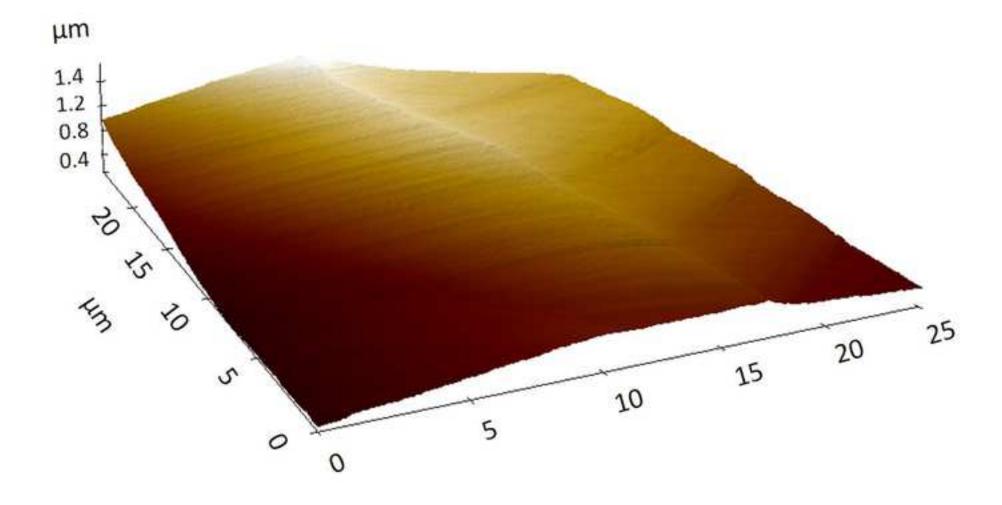


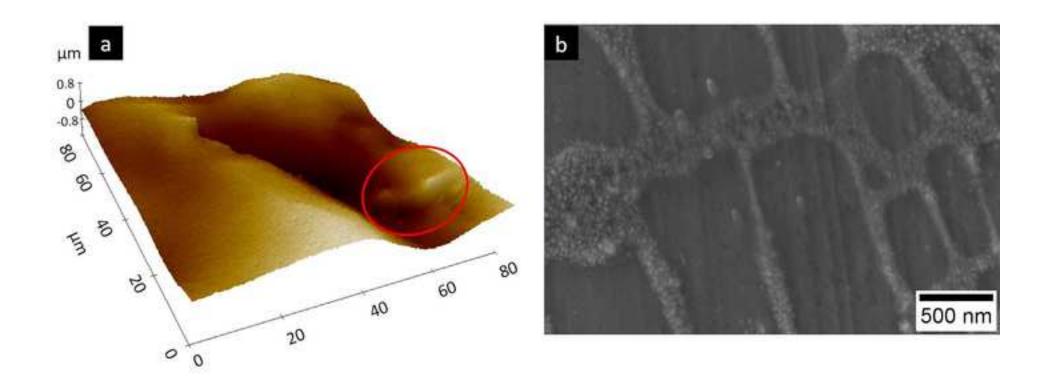


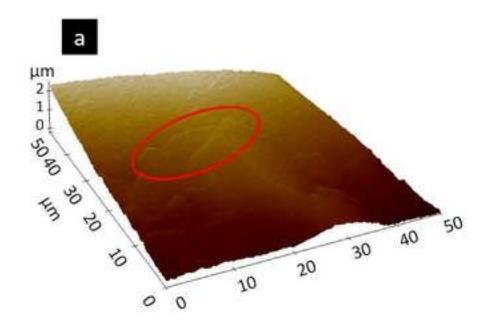


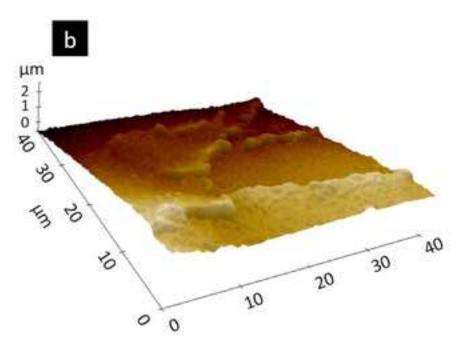


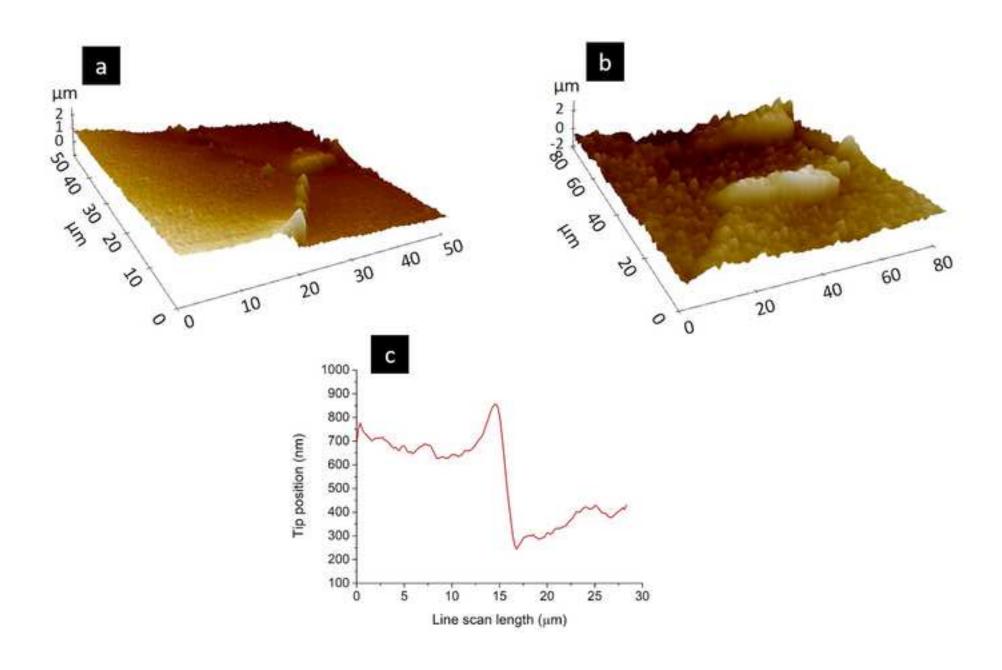


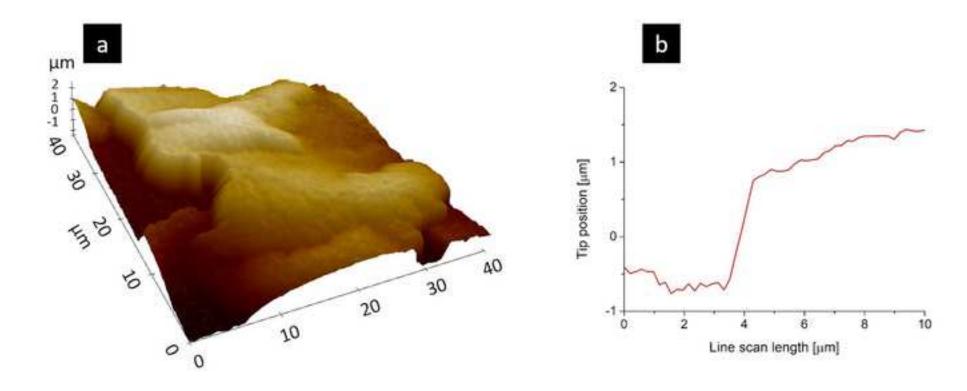


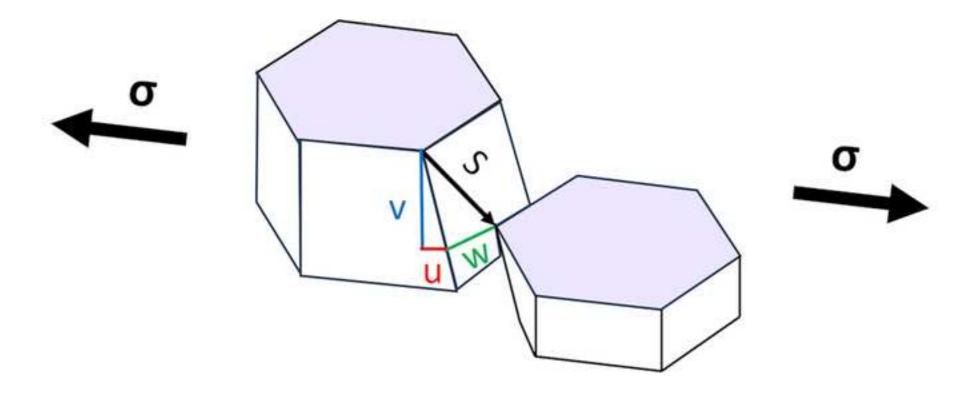


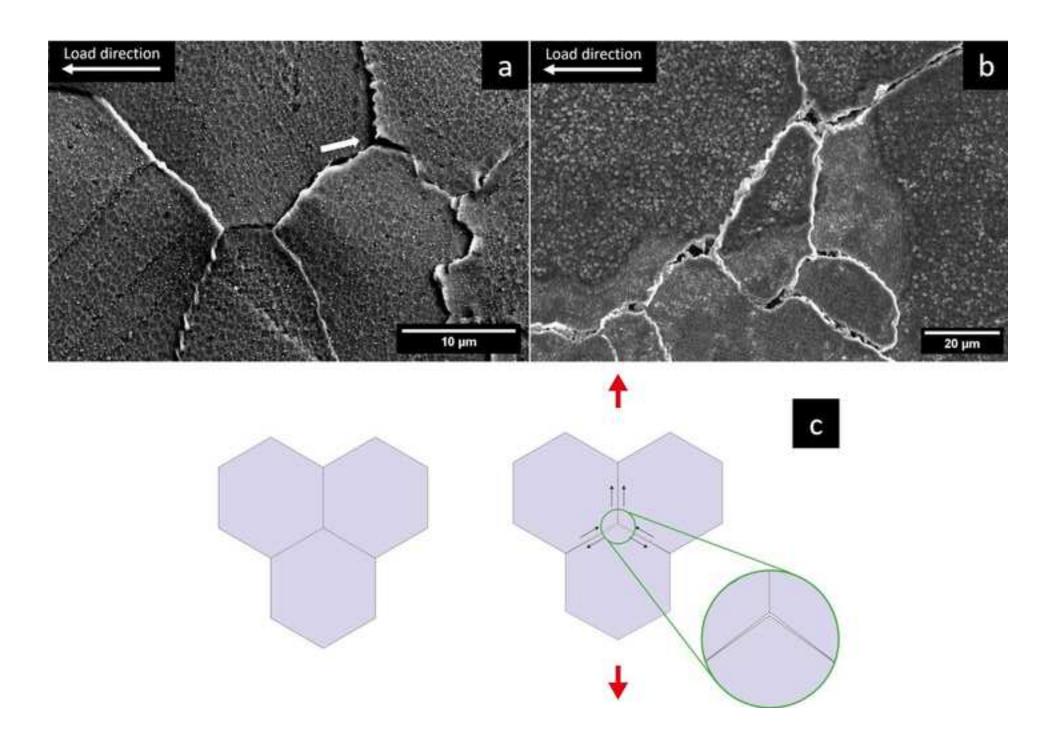


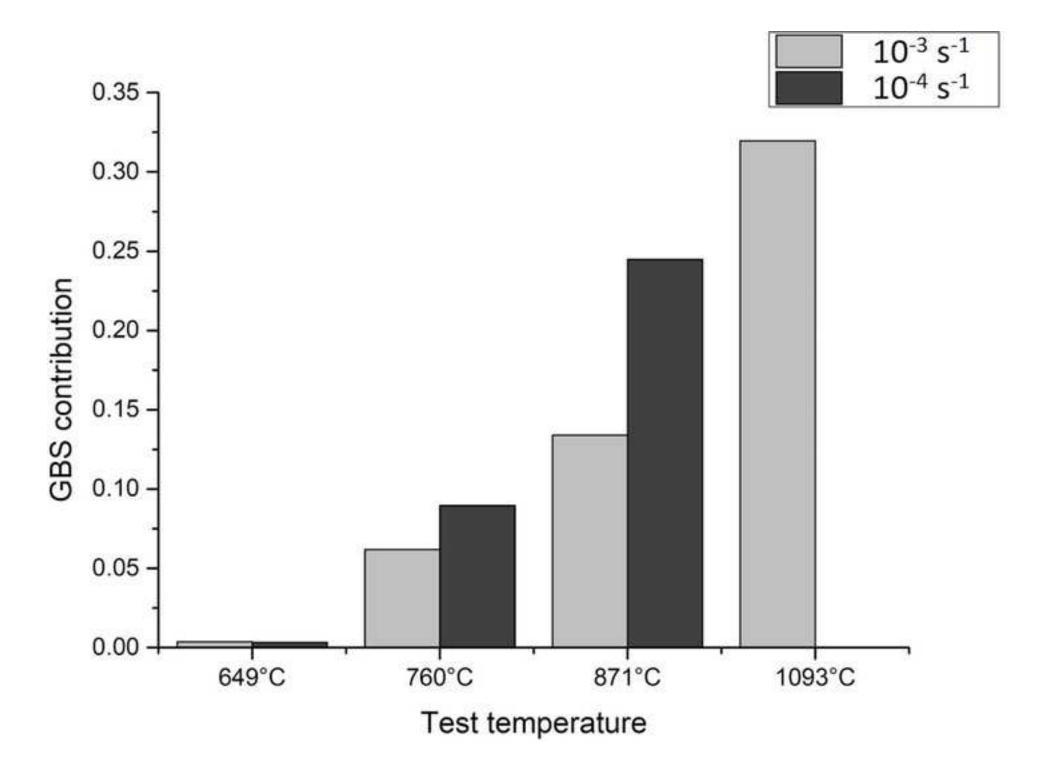












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